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Thesis submitted to the city University for the degree of Doctor of Philosophy, May. 1989

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# This thesis is dedicated to 

## CHITRA,

ANKUR, SANYAM,
and

AKANSHI

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## DECLARATION

## COPYRIGHT DECLARATION

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PRAMOD AGRAWAL

LONDON 1989

## SUMMARY

This thesis describes a new approach to UHS protection of $H V$ transmission line using high frequency components of fault generated noise. The approach is based on a scheme of the non-unit type which has features of unit type of protection. Advantages of unit-type protection are very well known for providing discrimination between internal and external faults. Such a scheme makes use of a stack-tuner circuit of the HV system which has a conventional coupling capacitor of the voltage transformer. The stack-tuner is tuned to a certain frequency bandwidth which does not interfere with the frequency of carrier communication if used, and the output is processed for comparison with a threshold voltage and the resulting signal being fed to a decision logic. A high speed directional detector has been used for discrimination against external faults. A signal from this directional detector can also be fed to the decision logic. The final operation of tripping depends on the output signal of the decision logic. For testing the scheme, a 100 km long 400 kV transmission line has been digitally simulated with sources of 5 GVA and 20 GVA for high frequency applications, typically 300 kHz . In order to cancel out the effect of noise due to other sources like communication etc., the tuner output of each phase has been summed in Aerial mode-2 form (1, -2, 1) and tests have been carried out for cases with and without bus bar capacitance.
Further effects of an electric arc on the newly developed scheme have been discussed as most faults involve arcing or some other kind of discharge. In the present scheme, a significant high frequency component of the arc voltage has been noticed which can affect the performance of the scheme. When carrying out simulation tests, a remarkable feature of the arc repeating every half cycle has been noticed. Another revolutionary feature of the arc model has been noticed; a significant response which repeats every half cycle, even for zero voltage fault inception. It has thereby been found that the scheme can operate at least in principle, for zero-voltage point-on-wave faults. This represents a significant improvement over schemes based on travelling wave phenomena which have been considered a failure for faults at zero degree fault inception angle. This has been a difficult task to simulate the system due to limited memory of the computer Gould 9095 available at the City University, London. For this purpose, program has been optimised for two power cycles of observation time for tuner frequency of 100 kHz .

## LIST OF SYMBOLS

| $\mathrm{C}_{\text {C }}$ | = coupling capacitor |
| :---: | :---: |
| cVt | = capacitor voltage transformer |
| NC | = short circuit or normally close mode |
| No | = open circuit or normally open mode |
| $\mathrm{s}_{\mathrm{S}}, \mathrm{S}_{\mathrm{R}}$ | = stack tuner as switch at sending end and receiving end |
| S | = a switch of the stack tuner operated by high speed direction detector |
| SE | = sending end |
| RE | = receiving end |
| DD or da | = high speed directional detector |
| $\mathrm{F}_{2}$ | = in-zone or internal fault |
| $\mathrm{F}_{1}$ | = out of zone or reverse fault |
| $\mathrm{V}_{\mathrm{F}}$ | = stack tuner output voltage signal after summation (if used) |
| $\mathrm{v}_{\text {A }}$ | = auto-correlator output voltage signal |
| $\mathrm{V}_{\mathrm{T}}$ | $=$ threshold voltage |
| $\mathrm{v}_{\mathrm{D}}$ | $=$ logic output 1 or 0 of the level detector |
| $\mathrm{T}_{\mathrm{F}}$ or TF | = forward direction logic signal of DD 0 or 1 |
| $\mathrm{T}_{\mathrm{R}}$ or TR | = reverse direction logic signal of DD 0 or 1 |
| $\mathrm{v}_{\text {o }}$ or $\mathrm{v}_{\text {OUT }}$ | = stack tuner output voltage |
| $\mathrm{v}_{1}$ or $\mathrm{v}_{\text {IN }}$ | = stack tuner input voltage or line voltage |


| $\mathrm{C}_{\mathrm{BB}}$ | = bus bar capacitance |
| :---: | :---: |
| $\mathrm{v}_{\mathrm{oa}}, \mathrm{v}_{\text {ob }}, \mathrm{v}_{\text {oc }}$ | = stack tuner output voltages for three phases a, b, c |
| $\mathrm{T}_{\mathrm{F}, \mathrm{R}}$ | = forward or reverse signal logic of DD 0 or 1 |
| ST | = stack tuner |
| TRIP or $\mathrm{T}_{\mathrm{S}}$ | $=$ tripping signal logic 0 or 1 |
| $\mathrm{L}, \mathrm{L}_{1}, \mathrm{~L}_{2}, \mathrm{~L}_{3}$ | = inductive components of power line carrier system used in Britain |
| $\mathrm{C}_{\mathrm{s}}, \mathrm{C}, \mathrm{C}_{1}$, | = capacitive components of power line carrier |
| $\mathrm{C}_{2}, \mathrm{C}_{3}, \mathrm{C}_{4}$ | system used in Britain |
| $\mathrm{R}_{0}$ | $=$ line characteristic impedance for plc system |
| $\mathrm{w}_{\mathrm{o}}=2 \pi \mathrm{f}_{\mathrm{o}}$ | $\begin{aligned} = & \text { angular power frequency (angular resonant } \\ & \text { or mid band frequency in plc applications) } \end{aligned}$ |
| $\mathrm{Z}_{\mathrm{T}}$ | $=$ line trap or bus bar impedance |
| $\mathrm{R}_{\mathrm{T}}, \mathrm{X}_{\mathrm{T}}$ | = resistance, reactance of line trap |
| E | $=$ transmitter voltage |
| $\mathrm{v}_{1}, \mathrm{v}_{2}$ | $=$ high frequency voltage on line without and with $R_{T}$ |
| f | = bandwidth in plc applications |
| $\mathrm{f}_{1}, \mathrm{f}_{2}$ | $=$ lower and upper cut-off frequency |
| CS, CL | $\begin{aligned} = & \text { coupling capacitance and series tuning coil } \\ & \text { inductance for designed stack tuner } \end{aligned}$ |
| $Q_{C}$ | $=$ quality factor of coil for designed stack tuner |
| RS | $=$ resistance of series tuning coil |
| RSP | = output impedance of stack tuner for short mode |
| CP, CLP | ```= capacitance, inductance of parallel tuned circuit``` |



| $v_{f s}(t)$ | = pre-fault voltage at fault point (time domain) |
| :---: | :---: |
| t | = instantaneous time |
| $\hat{\mathrm{v}}_{\mathrm{fS}}$ | $=$ peak value of voltage $\mathrm{v}_{\mathrm{fS}}(\mathrm{t})$ |
| $\Theta_{f s}$ | $=$ phase angle of voltage $\mathrm{v}_{\mathrm{fS}}(t)$ |
| $v_{f f}(t)$ | = superimposed voltage at fault point in time domain |
| $\bar{v}_{f f}(j w)$ | = transform of superimposed voltage $\left(v_{f f}(t)=0, t<0\right)$ |
| $\overline{\mathrm{v}}_{\mathrm{fs}}$ | = transform of prefault voltage at point of fault |
| $\bar{v}_{f f}, \bar{E}_{\text {ff }}$ | ```= transform of superimposed voltages at fault point``` |
| $\overline{\mathrm{v}}_{\text {Ss }}, \overline{\mathrm{I}}_{\text {SS }}$ | = transform of prefault voltages and currents at sending end of line |
| $\overline{\mathrm{V}}_{\text {RS }}, \overline{\mathrm{I}}_{\text {RS }}$ | $=$ transform of prefault voltages and currents at receiving end of line |
| $\overline{\mathrm{V}}_{\mathrm{S}}, \overline{\mathrm{v}}_{\mathrm{R}}$ | ```= sending and receiving end voltage transforms``` |
| $\bar{I}_{S}, \bar{I}_{R}$ | $=$ sending and receiving end current transforms |
| $h(t)$ | = unit step function |
| $\tilde{\mathrm{V}}_{S f}, \dot{I}_{S f}$ | = transform of superimposed voltage and current at sending end of line |
| $\overline{\bar{I}}_{f S f}, \overline{\bar{I}}_{f R f}$ | ```= transform of superimposed currents at fault point``` |
| $\overline{\mathrm{V}}_{\mathrm{Rf}}, \overline{\mathrm{I}}_{\text {Rf }}$ | = transform of superimposed voltage and current at receiving end of line |
| ${ }^{\mathrm{z}} \mathrm{SS}, \mathrm{z}_{\text {SR }}$ | $=$ sending and receiving end source impedance matrices |


| $\mathrm{R}_{\mathrm{f}}$ | = fault resistance matrix |
| :---: | :---: |
| $\dot{E}_{S a}, \bar{E}_{S b}, \bar{E}_{S c}$ | = phase a, b, c source voltage transform |
| $\overline{\mathrm{I}}_{S \mathrm{Sa}}, \overline{\mathrm{I}}_{S b}, \overline{\mathrm{I}}_{\mathrm{Sc}}$ | = phase $a, b, c$ source current transform |
| a,b, c | $=$ phases a,b,c |
| U | = unit matrix |
| $\mathrm{z}_{\mathrm{SO}}, \mathrm{z}_{\mathrm{S} 1}$ | = zero and positive phase-sequence source impedances |
| $\mathrm{Z}_{\mathrm{Sn}}$ | = neutral impedance of source |
| SCL | = short-circuit level |
| $\left\|\mathrm{V}_{\mathrm{L}}\right\|$ | $=$ magnitude of line voltage |
| $\mathrm{I}_{\text {Sc }}$ | $=$ short circuit current |
| $\mathrm{R}_{\mathrm{S} 1}, \mathrm{~L}_{\mathrm{S} 1}, \mathrm{X}_{\mathrm{S} 1}$ | = source resistance, inductance, reactance |
| $\mathrm{Y}_{\text {ST }}$ | $=$ admittance matrix of stack tuner |
| $\mathrm{Y}_{\mathrm{S} 1}, \mathrm{Y}_{\mathrm{Sn}}$ | = positive sequence, neutral admittance of source |
| $\mathrm{Y}_{\mathrm{Sn} 1}$ | = series combination of $\mathrm{Y}_{\mathrm{S} 1}, \mathrm{Y}_{\mathrm{Sn}}$ admittances |
| ${ }^{\text {V }}$ TT | = total source admittance |
| $\mathrm{y}_{\mathrm{CB}}$ | = bus bar capacitive admittance matrix |
| $\overline{\mathrm{V}}_{\text {fsnew }}$ | = transform of stack tuner output pre-fault voltages at fault point |
| $\overline{\mathrm{V}}_{\text {Sfnew }}$ | = transform of close mode stack tuner output superimposed voltages at sending end |
| $\overline{\mathrm{V}}_{\text {Rfnewc }}$ | = transform of close mode stack tuner output superimposed voltages at receiving end |
| $\overline{\mathrm{v}}_{\text {Sfnewo }}$ | = transform of open mode stack tuner output superimposed voltages at sending end |
| $\overline{\mathrm{V}}_{\text {Rfnewo }}$ | $=$ transform of open mode stack tuner output superimposed voltages at receiving end |
| $\overline{\mathrm{V}}_{\text {ff }}{ }_{\text {max }}(\mathrm{w})$ | = maximum superimposed voltage at fault point |


| $\overline{\mathrm{V}}_{\text {ffMIN }}(\mathrm{w})$ | = minimum superimposed voltage at fault point |
| :---: | :---: |
| $\mathrm{T}_{\mathrm{DD}}$ | = operating time delay of DD |
| $\mathrm{Z}_{\mathrm{L}}$ | = load impedance used for refraction coefficient |
| $\mathrm{R}_{\mathrm{r}}$ | = refraction coefficient |
| 'x',y | = signals of signal processing scheme |
| $P_{x y}(T)$ | = auto-correlated function |
| $\mathrm{x}(\mathrm{k} \delta \mathrm{t})$ | $=$ window of $N$ samples where $k=1 \cdots N$ and $\delta t$ is sampling time |
| $\mathrm{A}_{\mathrm{Y} 1^{1}}, \mathrm{~B}_{\mathrm{Y} 1}$ | = coefficients of two-port admittance matrix |
| $z, Y$ | = impedance, admittance per unit length of |
|  | line |
| V | $=$ Voltage vector of a semi-infinite line |
| $\mathrm{V}_{\mathrm{i}}$ | = incident voltage vector of semi-infinite |
|  | line |
| $\mathrm{v}, \mathrm{v}_{\mathrm{i}}$ | = variables related to modal analysis |
| Г1, 「3 | $=$ mode 1, mode 3 propagation constants |
| $\mathrm{a}_{\mathrm{S}}$ | = symmetrical component operator |
| $\mathrm{V}_{0}, \mathrm{v}_{1}, \mathrm{~V}_{2}$ | = modal voltages |
| $\mathrm{v}_{\mathrm{a}}, \mathrm{v}_{\mathrm{b}}, \mathrm{V}_{\mathrm{c}}$ | $=$ three phase voltages |
| $\mathrm{T}_{\mathrm{T}}$ | = operating time of the scheme/relay |
| $\mathrm{R}_{1}$ | $=$ reflection coefficient |
| $\mathrm{V}_{\mathrm{X}}, \mathrm{v}_{\mathrm{Y}}, \mathrm{V}_{\mathrm{Z}}$ | = signals of signal processing scheme |
| WS | = moving window for auto-correlation |
| WC | = fixed window for auto-correlation |
| $\mathrm{n}, \mathrm{N}_{\mathrm{S}}$ | $=$ number of samples of signal |
| $I_{p}$ | = peak value of steady state arc current |
| $\mathrm{V}_{\mathrm{p}}$ | $=$ maximum arc voltage per cm length of arc |




## CHAPTER 1

## INTRODUCTION

### 1.1 Literature survey

A literature survey of modern trends of high voltage transmission line protection and the effect of the electric arc in power system applications has been performed:

### 1.1.1Modern trends in HV transmission line protection

Conventional schemes [1,2] for EHV protection using pilot relays, distance and/or directional over current relays are very well known. Early forms of directional comparison relays and the later- developed phase comparison relays are commonly used. These schemes, together with distance relays employ non-unit measurements and they lack the advantages of inherent discrimination for external faults. Carrier aided protection schemes are commonly used as pilot schemes. These schemes are known as unit-type and provide the best discrimination against external faults. For these, the carrier communication link has to be highly reliable and secure. In order to attain reliability and security of this type of protection, a continuous monitoring of the communication channel is necessary which makes the scheme relatively expensive.

As these schemes operate on power frequency, these are liable to maloperate during initial disturbance of the system due to the transient effect if sufficient filtering or time delay is not provided. But this delay may cause damage and system instability, and therefore ultra-highspeed fault clearance has been warranted [3,5]. With this view many experts have been investigating new concepts of protection of EHV transmission lines and it is worth mentioning the idea of Johns [6] to develop a method which is capable of measuring accurately a signal from a window of information for a fraction of one period of power frequency without increasing significant delays due to extensive filtering. Johns [7] also claims that further reduction in overall fault clearance time has to be concentrated on fast speed protection schemes rather than operating times of circuit breakers. Later Johns [8] developed a new method of distance protection using Finite Transform Technique for UHS tripping. Other distance schemes using filtering and curve-fitting techniques $[9,10]$ have been developed which have half power cycle detection time.

In order to further improve detection times many authors [11-17] have used non-power frequency components in the fault signals. Chamia [11] uses the change in voltage and current signals to determine the direction of the fault. This relay is being commercially used. Takagi [12-15] computes travelling waves at both ends and uses their difference to detect an internal fault. Dommel [16] derives a travelling wave discriminant whose co-ordinates
at two ends of the line decides the operation of the relay. Johns [17] derives a sequence in which forward and reverse travelling waves exceed given thresholds to determine the direction of the fault. Vitins [18] describes a distance scheme based on the wave equation. Swift [19] gives a spectral analysis of post-fault signals which determine a dominant frequency component with a period of twice the propagation time from source to the fault. Crossley [20] describes a distance scheme using a correlation technique to measure the distance to the fault.

All unit-type schemes based on travelling waves developed so far need carrier communication link for discrimination against external faults. At present there is no such nonunit type of scheme which has features of unit type of protection. Mehdi [21] describes a non-unit scheme having features of unit-type of protection which uses processing of a fault generated current waveform. Such a scheme has limitations of response time of current transformers. For discrimination purposes, wave-traps on the primary side of the transmission line have been used. Normally, any change in, or application of equipment to the primary side is not appreciated. Later Johns [22] has postulated and described the principles of a new scheme in which the high frequency fault noise is detected in switched stack-tuning circuits. This work was the starting point for the work of this thesis and the author has performed extensive investigations into this techniquein order to lay the foundation for a practicable design of the hardware necessary for its impletation.

### 1.1.2Consideration of electric arcs in EHV Protection

It is a known fact that most faults on transmission lines involve arcing or some other discharge. The voltage drop across the arc is generally much smaller than the transmission line voltages and does not affect the operation of the relay which normally works on the steady state fault current. It is only the current which can affect the operation. Arcing faults in low voltage distribution systems have been studied by many authors [23

- 26]. Consideration of arc resistance in distance schemes [27-30] has been used for a long time and the arc has been taken into account as a linear resistance. Many authors [31-37] have studied static characteristics of electric arcs. Strom [31] has investigated the dynamic arc characteristics of a long arc at atmospheric pressure and has found that the characteristic follows different paths for increasing and decreasing current, calling it an arc-cylogram. It is also found that the arc characteristics do not repeat for every successive cycle. Brown [38] has also investigated the dynamic characteristics but for low current. Brown [38], Eaton [39], Slepian [40] have studied the arc extinction mechanism. Eaton [39] has found that the arc extinction may result from two causes: (1) following a current zero, the recovery voltage across the arc terminals fails to reach a value equal to the dielectric strength of the arc path; (2) the arc length is increased by the wind (or
otherwise) at such a rate that the voltage required to maintain the positive column of the arc is greater than that which can be supplied by the current. Slepian [40] has studied the factors for extinguishing the A.C. arc at a current zero: (1) the rate of recovery of the voltage across the arc, (2) the ability of the arcing medium to recover the dilectric strength. If the rate of recovery of the voltage across the arc exceeds the rate at which the arcing medium recovers dielectric strength, the arc will persist. Slepian also shows that the arc hystersis cylogram repeats only if the voltage across the arc is such that the time for which the arc current remains at zero following voltage reversal is sufficiently small to ensure that the arc is re-established without an appreciable restriking voltage i.e. negligible arc deionization occurs during zero current time. Where this is not a case, a significant reignition voltage has to be developed before the arc cylogram is attained. Recently, studies of arc characteristics in secondary arc application has gained importance and the work of Fukunishi [41]' for extinction of long secondary arcs is important. Johns [42 - 43] gives an expression for the reignition voltage and arc modelling with the reignition effect is possible. Cornick [44] has studied the power system transients caused by arcing faults considering weak and strong arc currents.

These transients may have high frequency components and their detection for protection application has not been previously considered. In the present study, methods have been developed for simulating the effect of electric arcs
on the high frequency components of the fault generated noise have been studied and their application for protection purposes is discussed.

### 1.2 Obiective of the thesis

Distance and/or directional over-current schemes are non unit-type schemes and lack the advantages of discrimination for external or reverse faults. Similarly carrier aided pilot schemes are unit-type schemes which have advantages of discrimination for external fault but depend upon the reliability of the communication systems. With this view, the objective of the present thesis is to investigate a new principle of a non-unit-type scheme having advantages of unit-type schemes. The principle is based on the travelling wave phenomena and has been postulated by Johns [22]. Basically, whenever a disturbance occurs on a transmission line, travelling waves travel in all directions and it is possible to detect these travelling wave components of voltages and currents at both ends of the transmission line without application of any communication link. To detect these travelling wave components generated due to fault, a signal of a particular band of frequencies has to be detected and to processed to establish the scheme.

Since most faults involve arcing or any other kind of discharge, studies of the effect of electric arcing on these fault-generated travelling wave components of a particular band of frequencies have to be performed. In
order to study the effect of higher frequency components of the arc e.g. 300 kHz for long observation time e.g. few cycles, it is also the objective of this thesis to optimise the computer program for system simulation due to limited capacity of the memory: 12 M byte of the computer Gould 9095, available at the City University, London.

It is well known that the arc voltage which is much smaller than the transmission line voltage, may repeat for a few power cycles following the instant of fault inception. This arc voltage may have high frequency components near the zero crossing of the arc voltage waveform. An extremly important aim of the thesis is also to detect such components repeating every half a cycle of the power frequency.

Further, schemes based on travelling wave phenomena have been considered failures for zero degree of fault inception angle. But if it is possible to detect the above travelling wave components of high frequency at zero degree of fault inception angle, perhaps it could be a success for the schemes based on travelling waves to operate on zero degree fault inception angle. With this view, it is also the aim of the scheme to detect high frequency arc voltage components at zero degree fault inception angle repeating every half a cycle of the power frequency.

These aims can be categarically stated as below:

1. To develop computational models of transmission line end source systems incorporating the equipment reported by Johns [22].
2. To investigate its performance and establish whether further work as its implementation is worthwhile.
3. To develop a means of incorporating arc models into the system simulation and prove their validity.
4. To investigate the extent to which the performance is affected by arcing fault conditions.
5. To refine the design of the equipment and, if necessary to propose means of improving the system developed for the stack tuner circuits and the signal processing means.

### 1.3 Structure of the thesis

This thesis has been structured on the basis of twelve chapters. The first chapter of the thesis has been devoted to the introduction of the thesis describing literature survey, objective of the thesis and structure of the thesis. Chapter 2 to Chapter 8 describe theoretical concepts and experimental results to establish the objectives of the thesis. Synopses of each of these chapters is described later in this section. Chapter 9 is devoted to the conclusion of the thesis. Chapter 10 is devoted to suggesstions for the future work. Chapter 11
lists out references for the thesis. Chapter 12 is an appendix in five sections: (A) power line carrier , (B) stack tuner frequency response data, (C) calulation of transmission line basic parameters, (D) program for band pass filter, (E) simulated test results of the arc cyclogram.

### 1.3.1 Chapter 2 synopsis

Chapter 2 describes theoretical concepts, operational principles, and design aspects of the basic scheme.

### 1.3.2 Chapter 3 synopsis

Chapter 3 is devoted to work regarding digital simulation of the stack tuner, transmission line, stack tuner with transmission line and for the output of the stack tuner.

### 1.3.3 Chapter 4 synopsis

Chapter 4 describes: (1) comparison of fault responses at tuner frequencies of $300,200,100,100-300,10-90 \mathrm{kHz}$, (2) establishment of the principle based on internal and reverse fault responses, (3) effect of different fault inception angles i.e. 90, 60, 30, 15, 5, 0 degrees, (4) effect of different source capacities i.e. 1 GVA at each end, 99 GVA at each end, 5 GVA at sending end and 20 GVA at receiving end.

## 1.3 .4 Chapter 5 synopsis

Chapter 5 is devoted to simulation techniques of the (1, 2, 1) modal component and describes the results on the basis of the (1, -2, 1) modal component responses for: (1) establishment of the principle, (2) detailed signal processing, (3) single phase to earth fault results at different points on the transmission line i.e. $1 \mathrm{~km}, 25 \mathrm{~km}$, 50 km , (3) other faults i.e. single phase to earth fault at phase A, phase B, phase C, three phase fault, two phase fault.

### 1.3.5 Chapter 6 synopsis

Chapter 6 is devoted to investigation of the scheme considering bus bar capacitance. Simulated test results for various tests have been included: (1) establishment of the principle, (2) effect of different fault inception angles i.e. $90,60,30,15,5,2,0$ degrees, (3) effect of different source capacities: 1 GVA, 50 GVA, 90 GVA at both ends.

### 1.3.6 Chapter 7 synopsis

Chapter 7 is devoted to studies relating to an electric arc. This chapter describes digital simulation techniques of the electric arc model considering the reignition effect. Later the chapter describes the digital simulation techniques of the superimposed arc voltages of the transmission line system.

### 1.3.7 Chapter 8 synopsis

Chapter 8 is devoted to the simulated test results considering the effect of an electric arc. Various test results are: (1) discrimination feature of the scheme, (2) advantages of electric arc at zero degree fault inception angle, (3) different arc lengths i.e. 2 meter, 20 meter, 40 meter, (4) arc effect repeating every half a cycle.

## CHAPTER 2

## THEORETICAL ASPECTS OF THE NEW SCHEME

This chapter describes the theoretical aspects of the proposed scheme:

### 2.1 Principle of the scheme

A fault on the power line produces wideband noise [45]. This noise has been considered a nuisance, e.g. it can interfere with carrier signalling equipment [46,47]. In the proposed principle this noise has been used for protection application. Various methods could be used to apply this principle, but in the scope of present work only the method of processing the noise through high frequency capacitance stack-tuning circuits at the ends of the protected circuit has been discussed. In most applications the coupling capacitor $C_{C}$ forms part of the capacitor voltage transformer (CVT) which is required to provide $50 / 60 \mathrm{~Hz}$ low voltage for metering and protection purposes. The stack tuner can be arranged to short circuit or open circuit any high frequency signals induced by a fault. The gain of the tuner can be arranged high in short circuit mode NC and low in open circuit mode NO. At the moment if these short and
open ciruit arrangements of the tuner circuit can be considered as a switch $S_{S}$ at the sending end $S E$ or $S_{R}$ at the receiving end $R E$ in short and open mode respectively, the principle can be explained with reference to Figure 2.1 as below:
$S_{S}$ and $S_{R}$ are normally open and are closed if a reverse fault is detected through a local directional detector DD inhibiting the tripping at local end. If this reverse fault occurs at SE , switch $\mathrm{S}_{\mathrm{S}}$ will be closed due to reverse fault and would restrict any in-band components from reaching RE thereby, even though the switch is in open mode at $R E$, inhibiting tripping at $R E$. In-band is referred to as the high frequency band of the tuner. If directions with reference to Figure 2.1 can be defined as forward at sending end $S E$ and receiving end $R E$ for in-zone fault $F_{2}$, then, for an out of zone fault at say $F_{1}$. SE would see $F_{1}$ as reverse and $R E$ would see $F_{1}$ as forward. Now to detect these directions any type of high speed directional detector $D D$ based on locally derived parameters e.g. distance protection / directional comparision relay or any other device can be used. For in-zone fault at $F_{2}$ the inband signals remain at both ends causing tripping at $S E$ and RE.

### 2.1.1 Consideration of the bus bar capacitance

In the HV transmission line system, many lines terminate at single bus-bar which responds for effective capacitance
between the bus-bar and the earth. This capacitance does not affect the power frequency signals due to very high impedance but terminate the line at any high frequency signals due to low impedance. The value of the capacitance may vary from line to line but a typical value of 0.1 micro-farad has been considerd for the present scope of the work. Under such circumtances, the only possible way of receiving the in band frequency signals is by providing the impedance of the comparable value of the stack tuner circuit. Therefore the stack tuner switch has been maintained in the normally closed, NC position for the internal faults as shown in Figure 2.2. For the reverse fault at $F_{1}$ the stack tuner switch has to be operated to open NO position by the contact of a DD switch inhibiting the tripping at the SE as shown in Figure 2.3. In the reverse fault condition, the high frequency signals are short circuited effectively through the low impedance of the bus bar capacitive circuit at $S E$ and can not reach the RE inhibiting the tripping at the RE.

### 2.2 Block-diagram representation

In general, the fault can be of any type and therefore it is necessary to combine the output of each stack tuner in such a way that a composite signal can be formed. A convenient way of combining the outputs involves the use of aerial mode components of propagation, and the $1,0,-1$ and 1, $-2,1$ mode distribution approximations are particularly convenient. Figure 2.2 shows more details of
the precise arrangement used for a 1, $-2,1$ combination.
The above arrangement ensures that any fault combination can be readily detected and it also ensures:
(1) any remote low level sources e.g., radio transmitters, do not cause interference with the operation of the scheme because any such remote source generally causes common-mode-signals to be induced in the power line and the 1, -2, 1 combination clearly causes such signal cancellation,
(2) to reduce the in-band signal detected by the healthy circuit following a fault on an adjacent circuit in double circuit or parallel feeder applications.

The output of the summation process of Figure 2.4, the tuned voltage $\left(V_{F}\right)$, is fed to an auto-correlator to strengthen the signal. The output of the auto-correlator $\mathrm{V}_{\mathrm{A}}$ is fed to a level detector which is set to pick up any signals present which lie above a threshold voltage $\mathrm{V}_{\mathrm{T}}$ i.e. , the output of the level detector is such that $V_{D}=1$, $\left|V_{A}\right|>V_{T}$ and $=0,\left|V_{A}\right|\left\langle V_{T}\right|$. The threshold voltage is set so that any noise in $V_{F}$ (generated by the tuner) and $V_{A}$ (generated by the auto-correlator) is effectively ignored. The threshold can also be used to control the sensitivity of the equipment to system faults. Estimating the threshold value after auto-correlating the signal improves discrimination against external faults due to strengthen of the signal.

The decision logic circuit of Figure 2.4 receives the
signal $\mathrm{V}_{\mathrm{D}}$ and the forward/reverse signals generated by the local directional detector $D D$. It is arranged to assert tripping of the local circuit breaker according to internal or reverse fault. As mentioned previosly, the local DD reverse signal is used to trigger the operation of switch $S$ to $N C$ mode in the stack tuning circuit as shown in Figure 2.1 or to $N$ mode as shown in Figure 2.3 for the case of the effective bus bar capacitance. This description of the signal processing is based on the theoretical concepts only. The signal processing used in this thesis has been described later in other chapters.

## 2.3 operational aspects

The operation of the scheme has been illustrated in Figure 2.5. Figure A is for an internal fault and Figure $B$ is for a reverse fault at $S E . V_{F}, V_{D}, T F, T R$, trip are the tuned voltage signal of the summation process, the level detector output signal, the forward direction signal of the directional detector, the reverse direction signal of the directional detector, the tripping signal based on the logic respectively. These are shown high or low at the SE or the $R E$ according to internal or reverse fault conditions. Now triping can take place only when $\mathrm{V}_{\mathrm{F}}$ and TF both are high (shown by 1) and triping has been shown by 1. For the reverse fault condition $T R$ is high shown by 1 and tripping cannot take place shown by low (0). For the internal fault condition, $V_{F}$ and $T F$ are high and tripping will take place at both ends SE and RE . For a reverse fault
condition at $S E$, the signal $T R$ is high and tripping cannot take place. At $R E, V_{F}$ is low(0), $T F$ is high and therefore tripping cannot take place at $R E$. Internal fault has been shown nearer to $R E$ so $D D$ at $R E$ is expected to operate earlier than $D D$ at $S E$ as shown in Figure 2.5. Similarly reverse fault is nearer to $S E$ and $D D$ at $S E$ is expected to operate earlier than $D D$ at $R E$ as shown in Figure 2.5. If relatively slow $D D$ is used, pulse stretching of the main signal may be required. A care is required to the extent, a pulse can be stretched as it may loose the security of the system for reverse faults.

When the bus bar capacitance is effective, the operational principle has been explained in Figure 2.6. Figure A shows the internal fault condition and Figure $B$ shows the reverse fault condition. Based on the decision logic input of $V_{D}$, $T_{F}, T_{R}$, the output trip signal can be seen 0 for the reverse fault condition and 1 for the internal fault condition. If $D D$ is relatively slow, the pulse stretching may be required.

### 2.4 Design consideration [46.47]

Figure 2.7 shows a typical circuit to tune with the stack tuning capacitance $C_{C}$ at 300 kHz in such a way that the impedance presented to the flow of in-band high frequency components is small with high gain (approximate to closure of $S_{S}$ and $S_{R}$, $N C$ in Figure 2.1 , or very large with relatively low gain (approximate to open of $S_{S}$ and $S_{R}, N O$ ). With switch $S$ closed operated through DD switch, the
circuit provides high gain output voltage $V_{o}$ in $N C$ condition which is proportional to the in-band system voltage. The values of the components in Figure 2.7 can also be used to adjust the band receptability relative to any carrier signalling channels (if these are used) to avoid any interference. The centre frequency can be adjusted to suit the application and, if no carrier signalling is used , there is virtually no restriction on the band of receptability. With switch $S$ open, the stack tuner provides low gain to distinguish with the condition of $S$ closed. If $V_{1}$ is the line voltage and $V_{0}$ is the stack tuner output voltage, the gain is $\mathrm{V}_{0} / \mathrm{V}_{1}$. This design is based on the theoretical concepts only. The design circuit used in this thesis has been described later in Chapter 3.


Figure 2.1 ARRANGMENT SHOWING STACK-TUNER AS A SWITCH AND
FAULT DIRECTIONS FOR REFERENCE


Figure 2.2 ARRANGMENT CONSIDERING BUS - BAR CAPACITANCE FOR INTERNAL FAULT


Eigure 2.3 ARRANGMENT CONSIDERING BUS BAR CAPACITANCE FOR REYERSE FAULTS


Figure 2.4 BLOCK DIAGRAM REPRESENTATION FOR SIGNAL PROCESSING


Figure 2.5 OPERATIONALPRINCIPLES OFTHE SCHEME


Figure 2.6 OPERATIONALPRINCIPLE OFTHE SCHEME CONDIDERING


Figure 2.7 STACK TUNER CIRCUIT

## CHAPTER 3

## DIGITAL SIMULATION OF TRANSMISSION IINE SYSTEM

This chapter describes the digital simulation of the transmission line and the stack tuner in the following sections:

### 3.1 Design digital simulation and frequency response of the stack tuner

Primary application of a coupling capacitor to the transmission line is very well known for measuring the line voltages for metering and protection applications. Another application of the capacitor is known for tuning with the line tuner for carrier communication purposes. Application of the coupling capacitor to match the line tuner for protection purposes is unknown so far and in the absence of such information, the circuits based on carrier communications have been studied and taken for reference [28,48] to develop a line tuner for protection applications.

### 3.1.1 Frequency range

Power line carrier generally operates over the band of frequencies between 30 and 300 kHz . These frequencies can be coupled to and transmitted over power lines with relatively small loss. Below 30 kHz , line coupling become impractical and above 300 kHz , the line losses are higher and radiation into space may interfere with licensed space. In U.K. the range is between 70 and 500 kHz .

### 3.1.2 Modes of operation

The power line carrier signal may use on-off, frequencyshift, or single-side band (ssb)operation. The on-off signal is used to block-tripping in an unfaulted line section; frequency-shift operation is used either for blocking or transfer-tripping. When functions are combined on an ssb channel, audio tones are used to modulate the carrier, thus providing the relaying intelligence.

## 3.1 .3 Line coupling

The power-line phase conductor is used to transmit the carrier signal between power stations. Many protective relaying carrier channels use single-phase-to-ground coupling, which requires only one set of coupling equipment at a line terminal. Phase to phase coupling may be used to improve dependability. It offers a better chance of getting a signal through an internal fault when the channel
is used for a transfer-trip relaying system. In Britain, it is usual to use phase-to-phase transmission [48].

A conventional stack tuner circuit used in Britain is shown in Figure 3.1. $C_{1}$ and $C_{4}$ are parts of a coupling capacitor to provide tapping for voltage measurement. $L_{1}$ is a series-tuning coil. $L_{2}$ and $C_{2}$ form a parallel-tuned circuit and connected to the primary of the isolating transformer. This isolating transformer is used for impedance matching. The parallel tuned circuit provides a high impedance path to high frequency signals and low impedance path to 50 Hz voltage of the line. The carrier equipment is connected across this parallel tuned circuit. $L_{3}$ and $C_{3}$ is also a parallel tuned circuit known as a line trap which provides low impedance at 50 Hz and high impedance at carrier frequencies to avoid short circuiting of the carrier signals through low impedance of the bus bar capacitance. The complete circuit of a phase-to-phase coupling is shown in Figure 3.2. Figure 'a' shows layout of the coupling equipment. Figure 'b' shows the actual circuit. Figure 'c' shows the derived equivalent circuit assuming that the bus bar capacitance is effectively an high frequency short-circuit.

It can be seen that this circuit has the form of a $\pi$ section band-pass filter, the band is fairly wide as shown in Figure 3.3. The frequency band is selected by varying line-trap capacitance $C_{3}$, series inductance $L_{1}$ and $L_{2}, C_{2}$ of a parallel tuned circuit. Using values of line trap inductance of $200 \mu \mathrm{H}$ and an $\mathrm{h} . \mathrm{v}$. capacitance of 2000 pF ,
the bands used on 132 kV and 275 kV are shown below:

| Band | Mid-band frequency kHz | Band Width kHz |
| :--- | :---: | :---: |
| A | 95 | $82-110$ |
| B | 140 | $110-180$ |
| C | 200 | $145-255$ |
| D | 340 | $194-600$ |
| E | 180 | $90-132$ |
| F | 152 | $116-196$ |

In the case of a 400 kV system, using line-trap inductance of 100 uH and coupling capacitor of 2000 pF , the bands are shown as below:

| Band | Mid-band frequency kHz | Band width kHz |
| :--- | :---: | :---: |
| 1 | 75.5 | $70-81$ |
| 2 | 87.5 | $80-95$ |
| 3 | 99.5 | $90-110$ |
| 4 | 112 | $100-125$ |
| 5 | 125 | $110-140$ |
| 6 | 138 | $120-158$ |
| 7 | 151 | $130-175$ |
| 8 | 179 | $150-214$ |
| 9 | 225 | $180-280$ |
| 10 | 353 | $250-500$ |

### 3.1.4 Design principle of the coupling equipment

3.1.4.1 132 kV and 275 kV systems: Upto 275 kV systems, bus bar capacitance can be assumed as a short circuit at high frequencies and can be based upon the filter theory. The equivalent circuit is as shown in Figure 3.4. $R_{o}$ is the line characteristic-impedance. If the line characteristicimpedance is matched with the carrier transmitter impedance, then the transmitter impedance is also $\mathrm{R}_{\mathrm{o}}$. Two shunt tuning circuits and one series tuning circuit have the same resonant frequency, $f_{o}$ (angular frequency, $w_{o}$ ) which is also a mid-band frequency. $C_{S}$ and $L$ are fixed values of the coupling capacitor and the line trap inductance. $C_{S}$ is half the value of the coupling capacitor and $L$ is twice the value of the line trap inductane due to phase-to-phase coupling. The mid-band frequency is varied by the shunt capacitors $C$ and the series tuning coil $L_{1}$. $L_{1}$ is twice of the value of the inductance of the series tuning coil. For correct matching,
$\mathrm{R}_{\mathrm{O}}=\left(\mathrm{L}_{1}\right)^{1 / 2} / 2 \mathrm{C}$
but $\quad w_{0}{ }^{2}=1 / L, C_{S}=1 / L C$
hence $R_{o}=\left(1 / \mathrm{w}_{\mathrm{O}}{ }^{2} \mathrm{C}_{\mathrm{S}}\right)^{1 / 2} \mathrm{x} \mathrm{w}_{\mathrm{O}}^{2} \mathrm{~L} / 2=(\mathrm{L})^{1 / 2} / 2 \mathrm{C}_{\mathrm{S}}$

For 132 kV and 275 kV systems, $\mathrm{R}_{\mathrm{o}}$ is equal to 450 Ohm if line trap inductance is $200 \mu \mathrm{H}$ and coupling capacitor is 2000 pF.

Now it can be shown that

$$
\begin{align*}
\text { Bandwidth } & \mathrm{f}=1 / \pi\left(2 \mathrm{~L}_{1} \mathrm{C}\right)^{1 / 2}  \tag{3.4}\\
\text { or } & \mathrm{f} \tag{3.5}
\end{align*}=\pi \mathrm{f}_{\mathrm{o}}^{2} \int 2 \mathrm{LC} \mathrm{~S}_{\mathrm{S}}
$$

It is also known that $f=f_{2}-f_{1}$
and $\mathrm{f}_{\mathrm{o}}=\int \mathrm{f}_{1} \mathrm{f}_{2}$

The frequency bands obtained have been listed earlier.
3.1.4.2 400 kV equipment $:$ In this system, bus bar capacitance is taken into account. The circuit arrangement and the equivalent circuit are shown in figures 'a' and 'b' of Figure 3.5. Now the line-trap and the bus bar capacitance (total impedance $\mathrm{Z}_{\mathrm{T}}$ ) are considered in terms of controlled loss on the coupling equipment.

It can be seen that the coupling filter is an $L$ section band pass, and $Z_{T}$ varies according to frequency and the bus bar capacitance.

In order to study the effect of the bus bar capacitance which may cause high attenuation peaks in the the pass band, a simple arrangement has been shown in Figure 3.6, ignoring attenuation of the coupling filter in the pass band.

Frequency response of a parallel tuned circuit for resistance $R_{T}$ and reactance $X_{T}$ has been shown in Figure 3.7.

In the pass band of the equipment, the reactive component of the line trap may resonant with the capacitance of the bus bars which varies over a wide range depending on particular switching conditions. Such resonance is a series resonance and if the circuit is a low loss circuit, the resulting value of the resistance $\mathrm{R}_{\mathrm{T}}$ across the line impedance $R_{o}$ is low and loss of power can be found considerable.

To avoid such a situation, a damping is provided in the line trap delibrately so that the resistive component is always relatively high in the pass band. If series resonance takes place, the resulting resistance across the line is sufficiently high to limit the loss of power to an acceptable level e.g. fifty percent reduction in power input to the line. This is a reason for considering bus bar capacitance and line trap as a controlled insertion loss rather than an integral element of the filter.

If a 50 percent power loss is assumed, insertion resistance of the line trap can be expressed in terms of the line impedance as below using a simple circuit shown in Figure 3.8:
h.f.voltage on the line without considering $R_{T}$ is
$V_{1}=E / 2$

The voltage on the line with $R_{T}$ is
$V_{2}=[E / 2] R_{T} /\left(R_{T}+R_{o} / 2\right)$

Ratio of Powers $=1 / 2=\left(V_{2} / V_{1}\right)^{2}$
$\mathrm{V}_{2} / \mathrm{V}_{1}=1 / / 2$
$\left[E / 2 \cdot R_{T} /\left(R_{T}+1 / 2 R_{o}\right)\right] /(E / 2)=1 / / 2=R_{T} /\left(R_{T}+\left(R_{0} / 2\right)\right)$ (3.11)

So $R_{T}=1.24 R_{0}$

This represents a lower limit of $\mathrm{R}_{\mathrm{T}}$. The higher value can be decided on the basis of the bandwidth. It can be shown that that the maximum bandwidth is available when $R_{T}=2$ times the lower limit i.e. $2.48 \mathrm{R}_{\mathrm{O}}$. If $\mathrm{R}_{\mathrm{T}}$ is more or less than the optimum value, bandwith may be lost as shown in Figure 3.9 .

Using the optimum value of this $R_{T}$ the bandwidth available for phase to phase coupling can be expressed as

$$
\begin{equation*}
\mathrm{f}=2 \pi \mathrm{f}_{\mathrm{O}} \mathrm{~L} / \mathrm{R}_{\mathrm{T}} \tag{3.13}
\end{equation*}
$$

where $L$ is the inductance of the line trap

$$
\begin{equation*}
\mathrm{R}_{\mathrm{T}}=1.24 \mathrm{R}_{\mathrm{o}} \tag{3.14}
\end{equation*}
$$

$R_{0}=$ Characteristic impedance of the line

$$
\begin{equation*}
\text { also } f=f_{2}-f_{1} \tag{3.15}
\end{equation*}
$$

and $\mathrm{f}_{\mathrm{o}}=\int \mathrm{f}_{1} \mathrm{f}_{2}$

With a view to increase the bandwidth, additional damping can be provided by adding another series resonant circuit tuned to $f_{0}$ and the damping resistance as shown in Figure 3.10 .

With this arrangement, the bandwith can be increased to twice the earlier value with simple damping line trap as shown in Figure 3.11.

The bandwidth so obtained is normally smaller than the bandwidth obtained with the filter theory applied for 132 kV and 275 kV systems.

## Manufactured Data

Manufactured data of the Power Carrier System of the Westinghouse Electric Corporation, USA are described in Appendix A for reference purposes. These include data of the line traps, the coupling capacitor, the line tuner, the drain coil, the coaxial cable and the power line characteristics.

### 3.1.5 Design of the stack tuner for protection applications

As shown in Figure 3.12, the basic circuit arrangement consists of three parts: (1) series resonance low impedance path as a short circuit, (2) directional detector (dd) switch ( as described in Chapter 2.4) in close mode as a short circuit to provide low impedance path to the primary of the output transformer, (3) a parallel resonance high impedance path as an open mode when dd switch is open. With reference to Figure 3.13 , the coupling capacitor CS is
assumed to be fixed. For series resonance of a selected mid band frequency $f_{0}$, the parameters: series inductance CL for case (1), quality factor $Q_{C}$ of the coil, equivalent resistance $R S$ can be evaluated in a sequence. Another resistance $\operatorname{RSP}$ is evaluated based on the stack tuner close mode gain. The typical value of the close mode gain is assumed to be 90 percent. The 'gain' has been defined earlier in Figure 2.7 as the tuner output voltage divided by the line to earth voltage. In this case, RS represents the component loss of the main coil CL and RSP represents the primary impedance of the transformer for the output voltage in the short mode. For case (2), the dd switch provides low impedance to the primary of the output transformer when closed. For case (3), a parallel tuned circuit tuned to mid band frequency $f_{o}$ provides high impedance to the primary of the output transformer when dd switch is opened. The capacitor $C P$ is assumed equal to the capacitor CS. Similarly, the inductance CLP is assumed equal to the inductance CL. Now quality factor of the coil, $Q_{C}$ for a parallel tuned circuit can be evaluated. Knowing $Q_{C}$, a parallel equivalent resistance, $R P$, is evaluated. With a view to match the output impedance with the input impedance in the open mode, it is necessary to provide an equivalent resistance $R S 1=R P$ as shown in Figure 3.13. Finally, in order to provide minimum circuit resistance in the short circuit mode, the switch $S$ (operated by a directional detector dd as referred earlier) is arranged to short circuit the parallel resonance circuit and the resistance RS1.

### 3.1.6 Digital simulation of the stack tuner

The equivalent circuit of the stack tuner in close mode is shown in Figure 3.14. This shows impedance of the circuit, YTSM (with reference to line). Total input voltage $V_{\text {IN }}$ across YTSM and impedance YSM proportional to output voltage $\mathrm{V}_{\text {OUT }}$ are also shown. Now gain can be simulated as below:

$$
\begin{equation*}
\text { GAIN }=V_{\text {OUT }} / V_{\text {IN }}=Y S M / Y T S M \tag{3.17}
\end{equation*}
$$

(absolute value for frequency response) GAIN in close mode is referred as YNC
so, YNC = (YSM/YTSM)
(absolute value for frequency response) Similarly, equivalent circuit in open mode is shown in Figure 3.15. This shows total impedance $Y T M$, voltage $V_{\text {IN }}$. Output voltage $\mathrm{V}_{\text {OUT }}$ is also shown. Now gain can be evaluated as below:

GAIN $=\mathrm{V}_{\text {OUT }} / \mathrm{V}_{\text {IN }}=(\mathrm{YOM} / \mathrm{YTM})$
(absolute value for frequency response)
Gain in open mode is referred as YNO
so, YNO = (YOM/YTM)
(absolute value for frequency response)
YNC = close mode gain
YNO = open mode gain
YTSM $=$ close mode impedance
YTM $=$ open mode impedance

### 3.1.7 Frequency response of the stack tuner

Based on the design method, the stack tuner has been designed for different mid band frequencies: $300 \pm 2.5 \mathrm{kHz}$, $200 \pm 2.5 \mathrm{kHz}, 100 \pm 2.5 \mathrm{kHz}, 200 \pm 100 \mathrm{kHz}, 50 \pm 40 \mathrm{kHz}$. For each mid band frequency, the stack tuner has been digitally simulated and the frequency responses for (1) open mode gain YNO, (2) open mode impedance YTM, (3) close mode gain YNC, (4) close mode impedance YTSM are plotted in the frequency range from 1 to 600 kHz . Figure 3.16 shows 'voltage gain' versus frequency response in the open mode and the close mode of the stack tuner (reference Chapter 2). Figure 3.17 shows the open mode impedance versus frequency response of the stack tuner for mid frequencies of $300 \pm 2.5 \mathrm{kHz}, 200 \pm 2.5 \mathrm{kHz}, 100 \pm 2.5 \mathrm{kHz}$. Figure 3.18 shows the close mode impedance versus frequency response of the stack tuner for mid frequencies of $300 \pm 2.5 \mathrm{kHz}, 200 \pm 2.5$ $\mathrm{kHz}, 100 \pm 2.5 \mathrm{kHz}$. Figure 3.19 shows impedance versus frequency response in the open mode and the close mode of the stack tuner for mid frequencies of $200 \pm 100 \mathrm{kHz}$ and $50 \pm 40 \mathrm{kHz}$. Frequency response characteristics have been found to be excellent as desired for the satisfactory performance of the whole scheme. The close mode gain has
been found to be 0.9. The open mode gain has been found to be 0.5. These values of the voltage gain can be varied subject to selection of the components and the design requirements. Simulated values for a case of mid frequency of $300 \pm 2.5 \mathrm{kHz}$ tuner are shown in Appendix B.

### 3.2 Digital simulation of the transmission line

### 3.2.1. Basic theoretical concept

A system voltage steady state equation can be expressed as

$$
\begin{equation*}
\frac{\mathrm{d}^{2} \overline{\mathrm{v}}}{\mathrm{dx}}=\mathrm{Z} \mathrm{Y}^{2} \overline{\mathrm{~V}} \tag{3.21}
\end{equation*}
$$

Based on the theory of natural mode developed by Wedepohl [49] , a solution to Equation 3.21 can be transformed into a number of independent differential eqations of the form:
$\overline{\mathrm{V}}=\exp (-\Phi \mathrm{x}) \overline{\mathrm{V}}_{\mathrm{i}}+\exp (\Phi \mathrm{X}) \overline{\mathrm{V}}_{\mathrm{r}}$
where $\Phi=Q \Gamma Q^{-1}$

Using a method of matrix transformation leading to diagonalisation, it is easy to calculate the surge impedance and hyperbolic functions,e.g.:

$$
\begin{align*}
& Z_{O}=Q \Gamma^{-1} Q^{-1} Z  \tag{3.23}\\
& \text { and } \cosh (\Phi x)=Q \cosh (\Gamma x) Q^{-1} \tag{3.24}
\end{align*}
$$

### 3.2.2 Selection for a two port transfer matrix representation or a two-port nodal admittance matrix:

A faulted transmission system up to the fault point(distance $x$ from the fault point) can be represented by the two port matrix formation as shown in figure 3.20 either impedance matrix $\left(A_{1}, B_{1}, C_{1}, D_{1}\right)$ as in Equation 3.25 or the admittance matrix $\left(A_{y 1},-B_{y 1}, B_{y 1},-A_{y 1}\right)$ as in Equation 3.26 .

$$
\left[\begin{array}{l}
\bar{v}_{S}  \tag{3.25}\\
\bar{I}_{S}
\end{array}\right]=\left[\begin{array}{ll}
A_{1} & B_{1} \\
C_{1} & D_{1}
\end{array}\right] \quad\left[\begin{array}{l}
\bar{E}_{\mathrm{fS}} \\
\overline{\mathrm{I}}_{\mathrm{fS}}
\end{array}\right]
$$

$$
\begin{align*}
& \text { where } A_{1}=\cosh (\Phi x) \quad B_{1}=\sinh (\Phi x) Z_{0} \\
& C_{1}=Y_{0} \sinh (\Phi x) \\
& D_{1}=Y_{0} \cosh (\Phi x) Z_{o} \text {, and } \\
& \Phi=Q \Gamma Q^{-1} \\
& {\left[\begin{array}{l}
\bar{I}_{S} \\
\bar{I}_{f S}
\end{array}\right]=\left[\begin{array}{ll}
A_{y 1} & -B_{y 1} \\
B_{y 1} & -A_{y 1}
\end{array}\right]\left[\begin{array}{l}
\bar{v}_{S} \\
\bar{E}_{f S}
\end{array}\right]} \tag{3.26}
\end{align*}
$$

$$
\text { where } \begin{aligned}
A_{Y 1} & =Y_{O} \operatorname{coth}(\Phi x) \\
B_{Y 1} & =Y_{O} \operatorname{cosech}(\Phi x) \\
Y_{0} & =[Z]^{-1} Q \Gamma Q^{-1}
\end{aligned}
$$

One main advantage of Equation 3.25 is that if the transmission system has different sections, it is easy to calculate voltage or currents at either end of the line simply by process of multiplication of the matrices e.g.:

$$
\left[\begin{array}{l}
\overline{\mathrm{V}}_{S}  \tag{3.27}\\
\overline{\mathrm{I}}_{S}
\end{array}\right]=\left[\begin{array}{cc}
\mathrm{A}_{1} & \mathrm{~B}_{1} \\
\mathrm{C}_{1} & \mathrm{D}_{1}
\end{array}\right] \quad\left[\begin{array}{ll}
\mathrm{A}_{\mathrm{f}} & \mathrm{~B}_{\mathrm{f}} \\
\mathrm{C}_{\mathrm{f}} & \mathrm{D}_{\mathrm{f}}
\end{array}\right] \quad\left[\begin{array}{ll}
\mathrm{A}_{2} & \mathrm{~B}_{2} \\
\mathrm{C}_{2} & D_{2}
\end{array}\right] \quad\left[\begin{array}{c}
\overline{\mathrm{V}}_{R} \\
\overline{\mathrm{I}}_{\mathrm{R}}
\end{array}\right]
$$

But, when dealing with the network impedances or admittances matrix, this form may not be very useful. The other difficulty can arise because of overflow during the evoluation of $\left(e^{\Phi x}\right)$ in $\cosh \Phi x$ and $\sinh \Phi x$ at higher frequencies. Since these difficulties are avoided in nodal admittance matrix, this particular method has been chosen for this thesis.

### 3.2.3 Fault transient modelling

Consider a single phase transmission line as shown in Figure 3.21 having a fault at point (F) and two end sources $P$ and $Q$.

By using the theory of superimposition, the fault can be considered as a voltage generator equal and opposite to the steady-state voltage at the instant of the fault inception,t $=0$. It can be represented as:

$$
\begin{equation*}
v_{f}(t)=v_{f s}(t)+v_{f f}(t) \tag{3.28}
\end{equation*}
$$

where $v_{f}(t)$ is the total voltage at point $F . v_{f s}(t)$ is prefault steady state voltage and $v_{f f}(t)$ is the superimposed voltage in the time domain.

$$
\begin{align*}
v_{f f}(t) & =0, t<0 \\
& =-v_{f S}(t), \quad t \geq 0 \tag{3.29}
\end{align*}
$$

In general, the prefault steady state voltage at the fault point can be expressed as

$$
\begin{equation*}
v_{f s}(t)=\hat{v}_{f S} \cdot \sin \left(w_{o} t+\Theta_{f S}\right) \tag{3.30}
\end{equation*}
$$

where the phase angle $\Theta_{f s}$ determine the point-on-the prefault waveform of the steady state voltage at which the fault occurs. The voltage $v_{f f}(t)$ can be assumed to exist in an unenergised system. By using the Heavyside Theorem, it can be expressed as:
$v_{f f}(t)=-\bar{V}_{f s} \sin \left(w_{o} t+\Theta_{f s}\right) h(t)$
or $v_{f f}(t)=-\hat{V}_{f S}\left(\sin w_{o} t \cos \Theta_{f S}+\cos w_{o} t \sin \Theta_{f S}\right) h(t)$

Now, by applying the Laplace Transform, Equation 3.32 becomes:
$v_{f f}(s)=-\hat{V}_{f s} \frac{w_{o} \cos \Theta_{f s}}{s^{2}+w_{o}{ }^{2}}+\frac{s \sin \Theta_{f s}}{s^{2}+w_{o}{ }^{2}}$
This is the basic form of equation which is responsible for the fault transient modelling and has been used in this work.

### 3.2.4 Frequency spectrum

Now, reconsider Equation 3.33 as $v_{f f}(t)$ is finite for the condition $t \geq 0$, and can be written as:
$\bar{V}_{f f}(j w)=\bar{V}_{f f}(s)$

$$
\begin{equation*}
\text { if } s=j w \tag{3.34}
\end{equation*}
$$

or $\bar{V}_{f f}(j w)=\hat{V}_{f s} \frac{w_{o} \cos \Theta_{f S}}{w^{2}-w_{o}^{2}}+\frac{j w \sin \Theta_{f S}}{w^{2}-w_{o}{ }^{2}}$
This represents a continuous function of frequency. The full spectrum is carried out through a Fast Fourier Transform Routine.

### 3.2.5 Determination of the time varying components

With a view to determine the time varying voltage, the Inverse Fourier Transform may be used. The total superimposed voltage can be expressed as;
$v_{f f}(t)=1 / 2 \int_{-\infty}^{\infty} \bar{v}_{f f}(j w) e^{j w t} d w$

Now, substituting value of $\tilde{v}_{f f}(j w)$ from Equation 3.35 to Equation 3.36 and can be written as:
$v_{f f}(t)=1 / 2 \pi \int_{-\infty}^{\infty} \frac{w_{o} \cos \Theta_{f s}+j w \sin \Theta_{f s} e^{j w t} d w}{\left(w^{2}-w_{o}^{2}\right)}$

Earlier it has been seen [49] that these time varying components are not readily available as divergence occurs and leads to distortion. However, it has been found that a satisfactory result can be obtained using a Modified Fourier Transform Technique and can be expressed as:
$f(t)=\operatorname{REAL} \frac{e^{\alpha t}}{\pi} \int_{0}^{\Omega} f(w-j \alpha) e^{j w t} \sigma d w$

The finite range of integration gives rise to Gibbs Oscillation, and this can be overcome by introducing the factor

$$
\begin{equation*}
\sigma=\sin (\pi \mathrm{w} / \Omega) /(\pi \mathrm{w} / \Omega) \tag{3.39}
\end{equation*}
$$

In addition, a frequency shift constant ' $\alpha$ ' is introduced to ensure numerical stability when the integral is evaluated digitally. $\Omega$ is truncation frequency.

### 3.2.6. Steady state calculation

Pre-fault voltages and currents can be related by

$$
\left[\begin{array}{l}
\overline{\mathrm{V}}_{\mathrm{SS}}  \tag{3.40}\\
\overline{\bar{I}}_{\mathrm{SS}}
\end{array}\right]=\left[\begin{array}{ll}
\mathrm{A}_{1} & \mathrm{~B}_{1} \\
\mathrm{C}_{1} & \mathrm{D}_{1}
\end{array}\right]\left[\begin{array}{ll}
\mathrm{A}_{2} & \mathrm{~B}_{2} \\
\mathrm{C}_{2} & \mathrm{D}_{2}
\end{array}\right]\left[\begin{array}{l}
\overline{\mathrm{V}}_{\mathrm{RS}} \\
\overline{\mathrm{I}}_{\mathrm{RS}}
\end{array}\right]=\left[\begin{array}{ll}
\mathrm{A} & \mathrm{~B} \\
\mathrm{C} & \mathrm{D}
\end{array}\right]\left[\begin{array}{l}
\overline{\mathrm{V}}_{\mathrm{RS}} \\
\overline{\mathrm{I}}_{\mathrm{RS}}
\end{array}\right]
$$

The sending end current before the fault is given by
$\overline{\mathrm{I}}_{S S}=\left[\mathrm{C}-\mathrm{D} \mathrm{B}^{-1} \mathrm{~A}\right]\left[\overline{\mathrm{V}}_{\mathrm{RS}}\right]+\mathrm{DB} \overline{\mathrm{V}}_{S S}$
and, the steady state value $\overline{\mathrm{v}}_{\mathrm{f}}$ is given by
$\bar{V}_{f s}=\left[\begin{array}{lll}A_{2} & \left.B_{2} B^{-1} A\right] \bar{V}_{R S}+B_{2} B^{-1} \bar{V}_{S S},\end{array}\right.$

Solution for the above equations is obtained at power frequency. The voltages and currents are calculated in the phaser form and converted directly in the time domain. It does not involve inverse transform.

### 3.2.7. Analysis of superimposed voltages for internal faults

Considering the homogeneous sections of the transmission line in Figure 3.22, the following equations can be written as:

$$
\left[\begin{array}{l}
\bar{I}_{S f}  \tag{3.43}\\
\bar{I}_{f S f}
\end{array}\right]=\left[\begin{array}{cc}
A_{y 1} & -B_{y 1} \\
B_{y 1} & -A_{y 1}
\end{array}\right]\left[\begin{array}{c}
\bar{v}_{S f} \\
\bar{E}_{f f}
\end{array}\right]
$$

and

$$
\left[\begin{array}{l}
\overline{\bar{I}}_{f R f}  \tag{3.44}\\
\overline{\mathrm{I}}_{\mathrm{Rf}}
\end{array}\right]=\left[\begin{array}{ll}
\mathrm{A}_{\mathrm{y}^{2}} & -\mathrm{B}_{\mathrm{Y} 2} \\
\mathrm{~B}_{\mathrm{y}^{2}} & -\mathrm{A}_{\mathrm{Y} 2}
\end{array}\right] \quad\left[\begin{array}{c}
\bar{E}_{\mathrm{ff}} \\
\overline{\mathrm{v}}_{\mathrm{Rf}}
\end{array}\right]
$$

Considering Equation 3.43 :

$$
\begin{equation*}
\overline{\mathrm{I}}_{S f}=\mathrm{A}_{\mathrm{Y} 1} \overline{\mathrm{~V}}_{\mathrm{Sf}}-\mathrm{BY} 1 \bar{E}_{\mathrm{ff}} \tag{3.45}
\end{equation*}
$$

and,

$$
\begin{align*}
& \bar{I}_{f S f}=B_{y 1} \bar{v}_{S f}-A_{y 1} \bar{E}_{f f}  \tag{3.46}\\
& \bar{v}_{S f}=-z_{S S} \bar{I}_{S f} \tag{3.47}
\end{align*}
$$

using Equation 3.47 and Equation 3.45

$$
\begin{equation*}
\left[z_{S S}^{-1}+A_{y 1}\right] \bar{v}_{S f}=B_{y 1} \bar{E}_{f f} \tag{3.48}
\end{equation*}
$$

using Equation 3.48 and Equation 3.46

$$
\bar{E}_{f f}=\left[B_{y 1}\left\{\left(Z_{S S}^{-1}\right)+A_{y 1}\right\}^{-1} B_{y 1}{ }^{-A_{y 1}}\right]^{-1} \cdot \bar{I}_{f S f}(3.49)
$$

now, considering the section from the fault location to the receiving end, a relationship between $\overline{\mathrm{I}}_{\mathrm{fRf}}$ and $\overline{\mathrm{E}}_{\mathrm{ff}}$ can be obtained:
using Equation 3.44

$$
\begin{equation*}
\bar{I}_{f R f}=A_{y 2} \bar{E}_{f f}-B_{y 2} \bar{V}_{R f} \tag{3.50}
\end{equation*}
$$

$\overline{\mathrm{I}}_{\mathrm{Rf}}=\mathrm{B}_{\mathrm{y} 2} \overline{\mathrm{E}}_{\mathrm{ff}}-\mathrm{A}_{\mathrm{y} 2} \overline{\mathrm{~V}}_{\mathrm{Rf}}$
but,
$\bar{v}_{R f}=Z_{S R} \cdot \bar{I}_{R f}$
using Equation 3.52 and Equation 3.51
$\left[\left(Z_{S R}\right)^{-1}+A_{y 2}\right] \dot{V}_{R f}=B_{y 2} \cdot \bar{E}_{f f}$
now, using Equation 3.53 and Equation 3.50
$\bar{E}_{f f}=\left[A_{y 2}-B_{y 2}\left\{\left(Z_{S R}\right)^{-1}+A_{y 2}\right\}^{-1} B_{y 2}\right]^{-1} \cdot \bar{I}_{f R f}$

Considering any general earth fault at the fault location, the following relationship can be deduced:
$\bar{E}_{f f}=\bar{v}_{f f}+\operatorname{Rf}\left[\overline{\bar{I}}_{f S f}-\overline{\bar{I}}_{f R f}\right]$
where $R f$ is a diagonal matrix representing the fault resistance in each phase.
now, using Equation 3.54 and Equation 3.49

$$
\begin{align*}
{\left[\overline{\bar{I}}_{f S f}-\overline{\mathrm{I}}_{f R f}\right]=} & {\left[\left\{B_{y 1}\left(Z_{S S}{ }^{-1}+A_{y 1}\right)^{-1} \cdot B_{y 1}-A_{y 1}\right\}-\right.} \\
& \left.\left\{A_{Y 2}-B_{y 2}\left(\left(Z_{S R}\right)^{-1}+A_{y 2}\right)^{-1} B_{y 2}\right\}\right] \cdot \dot{E}_{f f} \tag{3.56}
\end{align*}
$$

Finally, substituting Equation 3.56 into Equation 3.55 :

$$
\begin{align*}
\bar{V}_{f f}= & {\left[\left\{\left(B_{y 1}\left(Z_{S S}^{-1}+A_{y 1}\right)^{-1} B_{y 1}-A_{y 1}\right)-\right.\right.} \\
& \left.\left.\left(A_{Y 2}-B_{Y_{2}}\left(\left(Z_{S R}\right)^{-1}+A_{Y 2}\right)^{-1} B_{y 2}\right)\right\}-R f\right] . \\
& {\left[\bar{I}_{f S f}-\bar{I}_{f R f}\right] } \tag{3.57}
\end{align*}
$$

### 3.2.8. Evaluation of superimposed voltages currents at the relaving points

When the relay is located at the sending end, voltage $\overline{\mathrm{V}}_{\mathrm{Sf}}$ and current $\overline{\mathrm{I}}_{\text {Sf }}$ can be evaluated as follows : using

Equation 3.48
$\bar{V}_{S f}=\left[\left(Z_{S S}{ }^{-1}\right)+A_{y 1}\right]^{-1} B_{y 1} \dot{E}_{f f}$
substituing this equation in Equation 3.45
$\bar{I}_{S f}=\left[A_{y 1}\left\{\left(Z_{S S}{ }^{-1}\right)+A_{Y 1}\right\}^{-1} B_{y 1}-B_{y 1}\right] \cdot \bar{E}_{f f}$

Similarly, when the relay is located at the receiving end, voltage $\overline{\mathrm{V}}_{\mathrm{Rf}}$ and current $\overline{\mathrm{I}}_{\mathrm{Rf}}$ can be evaluated as follows : using Equation 3.53

$$
\begin{equation*}
\overline{\mathrm{V}}_{\mathrm{Rf}}=\left[\left(\mathrm{Z}_{\mathrm{SR}}^{-1}\right)+A_{\mathrm{Y} 2}\right]^{-1} \cdot \mathrm{~B}_{\mathrm{Y} 2} \bar{E}_{\mathrm{ff}} \tag{3.60}
\end{equation*}
$$

substituing this equation in Equation 3.51
$\bar{I}_{R f}=\left[B_{y 2}-A_{y 2}\left\{\left(Z_{S R}^{-1}\right)+A_{y 2}\right\}^{-1} \cdot B_{y 2}\right] \cdot \bar{E}_{f f}$

### 3.2.9 Analysis of main source network models:

The digital simulation for the main source is based upon selecting the short-circuit level arbitrarily at the terminating bus bars. Neglecting any capacitance effect, a simplified model (see Figure 3.23) has been used as follows:

$$
\begin{equation*}
\dot{E}_{S a}=\overline{\mathrm{I}}_{S a} \cdot \mathrm{Z}_{S 1}+\left(\overline{\mathrm{I}}_{S a}+\overline{\mathrm{I}}_{S b}+\overline{\mathrm{I}}_{S C}\right) \cdot \mathrm{Z}_{\mathrm{Sn}}+\mathrm{V}_{\mathrm{Sa}} \tag{3.62}
\end{equation*}
$$

or $\overline{\mathrm{E}}_{\mathrm{Sa}}=\overline{\mathrm{I}}_{\mathrm{Sa}} \cdot\left(\mathrm{Z}_{\mathrm{S} 1}+\mathrm{Z}_{\mathrm{Sn}}\right)+\overline{\mathrm{I}}_{\mathrm{Sb}} \cdot \mathrm{Z}_{\mathrm{Sn}}+\overline{\mathrm{I}}_{\mathrm{Sc}} \cdot \mathrm{Z}_{\mathrm{Sn}}+\mathrm{V}_{\mathrm{Sa}}$
similarly,

$$
\begin{align*}
& \bar{E}_{S b}=\overline{\bar{I}}_{S a} \cdot Z_{S n}+\bar{I}_{S b}\left(Z_{S 1}+Z_{S n}\right)+\bar{I}_{S C} \cdot Z_{S n}+\bar{V}_{S b}  \tag{3.64}\\
& \bar{E}_{S C}=\overline{\bar{I}}_{S a} \cdot Z_{S n}+\bar{I}_{S b} \cdot Z_{S n}+\bar{I}_{S C} \cdot\left(Z_{S 1}+Z_{S n}\right)+\bar{V}_{S C} \tag{3.65}
\end{align*}
$$

In matrix form,

$$
\left[\begin{array}{l}
\dot{\mathrm{I}}_{\mathrm{Sa}}  \tag{3.66}\\
\overline{\mathrm{I}}_{\mathrm{Sb}} \\
\dot{\mathrm{I}}_{\mathrm{Sc}}
\end{array}\right]=\left[\begin{array}{lll}
\left(z_{\mathrm{S} 1}+\mathrm{z}_{\mathrm{n}}\right) & z_{\mathrm{Sn}} & z_{\mathrm{Sn}} \\
z_{\mathrm{Sn}} & \left(z_{\mathrm{S} 1}+z_{\mathrm{Sn}}\right) & z_{\mathrm{Sn}} \\
z_{\mathrm{Sn}} & \mathrm{z}_{\mathrm{Sn}} & \left(z_{\mathrm{S} 1}+z_{\mathrm{Sn}}\right)
\end{array}\right] \quad\left[\begin{array}{l}
\dot{\mathrm{v}}_{\mathrm{Sa}} \\
\dot{\mathrm{v}}_{\mathrm{Sb}} \\
\dot{\mathrm{v}}_{\mathrm{Sc}}
\end{array}\right]
$$

In a general form, it can be stated as:

$$
\left[\begin{array}{l}
\overline{\mathrm{E}}_{S}  \tag{3.67}\\
\overline{\bar{I}}_{S}
\end{array}\right]=\left[\begin{array}{lc}
\mathrm{U} & \mathrm{Z}_{\mathrm{SS}} \\
0 & \\
&
\end{array}\right]\left[\begin{array}{l}
\overline{\mathrm{V}}_{\mathrm{S}} \\
\overline{\mathrm{I}}_{\mathrm{S}}
\end{array}\right]
$$

Where $\left[Z_{S S}\right]$ is the source impedance matrix of diagonal elements $\left(Z_{S 1}+Z_{S n}\right)$ and off-diagonal elements of $\left(Z_{S n}\right)$.These elements are calculated as:

$$
\begin{align*}
& \mathrm{SCL}=3 \cdot\left|\mathrm{~V}_{\mathrm{L}}\right| \cdot \mathrm{I}_{\mathrm{SC}}=\frac{\sqrt{3}\left|\mathrm{~V}_{\mathrm{L}}\right|^{2}}{\sqrt{3}\left|\mathrm{Z}_{\mathrm{S} \uparrow}\right|} \\
& \text { or } \mathrm{Z}_{\mathrm{Si} 1}=\left|\mathrm{V}_{\mathrm{L}}\right|^{2} / \mathrm{SCL}
\end{align*}
$$

But,
$Z_{S 1}=R_{S 1}+j w L_{S 1}=R_{S 1}+j X_{S 1}$

By knowing the $X / R$ ratio of the sending and receiving end sources at power frequency, elements of Equation 3.68 can be found out. This ratio is of the order of 30-300.

The neutral source impedance ( $Z_{\mathrm{Sn}}$ )is calculated using both positive phase sequence( $\left.\mathrm{Z}_{\mathrm{S} 1}\right)$ and zero phase sequence $\left(\mathrm{Z}_{\mathrm{S} 0}\right)$ impedances for the circuit of Figure 3.23, where
$z_{S 0}=z_{S 1}+3 z_{S n}$

Hence, $z_{S n}=z_{S 1}\left\{\left(z_{S 0} / z_{S 1}\right)-1\right\} / 3$

Various parameters like short circuit level(SCL), (X/R) ratio at power frequency, $\left(\mathrm{Z}_{\mathrm{S} 0} / \mathrm{Z}_{\mathrm{S} 1}\right)$ ratio, system line voltage $\mathrm{V}_{\mathrm{L}}$ are input data for the computer program and used for this simulation.

### 3.2.10. Fault-simulation studies

For simulating and studying the high frequency transients, a $400 \mathrm{kV}, 3$-phase (a, b, c) 100 km long transmission line with sources of 5 GVA at the sending end (SE) and 20 GVA at the receiving end (RE) has been considered. A phase to earth fault has been considered in the phase-a at a distance of 50 km from the SE and the fault induced transients have been studied in the faulted phase-a and the healthy phases $b$ and $c$. The data for the vertical configuration of the conductors (shown in Figure 3.24) are:

1) phase conductors $4 \times 54 / 7 / 0.33 \mathrm{~cm}$ with 0.305 m bundle spacing, 2) earth wire $54 / 7 / 0.33 \mathrm{~cm}, 3)$ earth resistivity 100 Ohm meter. The phase angle which determine the fault at the pre-fault waveform is 90 degree. The frequency shift constant ' $\alpha$ '(reference Equation 3.38) has been critically adjusted to a value of 1000 to get satisfactory waveforms. In the present study, the sampling frequency (SF) of 819.2 kHz has been used. The fault observation time is 5 msec with the number of samples of 4096. The fault observation time of 5 msec is appropriate for the present study as most transients die in this period. Figure 3.25 and Figure 3.26 show the fault responses in the faulted phase a and the healthy phases $b$ and $c$ with the sampling frequency of 6.4 kHz at the sending end (SE) and the receiving end (RE). Figure 3.27 and Figure 3.28 show the responses with 819.2 kHz of the sampling frequency. The pre-fault time is only 10 micro-seconds which is not visible in the fault response of 5 msec . A delay of about 0.18 msec is clearly shown in Figure 3.25 to Figure 3.28 . This delay can be confirmed due to the travelling time of the fault induced travelling waves. Assuming the speed of the travelling wave is nearly same as the speed of the light i.e. 300 km per second, then the time taken by the travelling wave to travel a distance of 50 km is about 0.17 msec. Figure 3.27 and Figure 3.28 indicate clearly the high frequency components which are not visible in Figure 3.25 and Figure 3.26. The fault induced transients can be seen in all the phases. The peak value of the transients in the faulted phase has been found to be about twice the
peak value of the transients in the healthy phases.

### 3.3 Digital simulation of stack tuner with transmission line

When sources are earthed at the two ends of the transmission line, the source impedance can be considered in parallel with the stack-tuner impedance. The arrangement of the transmission line, the end sources, and the stack-tuner has been shown in Figure 3.29. A technique for digital simulation of the end sources is explained in Section 3.2.

Admittance matrix for stack tuner circuit $Y_{S T}$ can be written as:

$$
\left[\begin{array}{lll}
{ }^{Y_{S T}} & 0 & 0  \tag{3.71}\\
0 & Y_{S T} & 0 \\
0 & 0 & Y_{S T}
\end{array}\right]
$$

$Y_{S T}=1 / Y T S M$ for close mode (reference Chapter 3.1.7)
$Y_{S T}=1 / Y T M$ for open mode (reference Chapter 3.1.7) where YTSM and YTM are short mode and open mode impedances of the stack tuner.

If $Y_{S 1}=1 / Z_{S 1}$ and $Y_{S n}=1 / Z_{S n}$ in Equation 3.66,
the admittance matrix for the source can be written as:

$$
\left[\begin{array}{lll}
\left(\mathrm{Y}_{\mathrm{Sn} 1}\right) & \mathrm{Y}_{\mathrm{Sn}} & \mathrm{Y}_{\mathrm{Sn}}  \tag{3.72}\\
\mathrm{Y}_{\mathrm{Sn}} & \left(\mathrm{Y}_{\mathrm{Sn} 1}\right) & \mathrm{Y}_{\mathrm{Sn}} \\
\mathrm{Y}_{\mathrm{Sn}} & \mathrm{Y}_{\mathrm{Sn}} & \left(\mathrm{Y}_{\mathrm{Sn} 1}\right)
\end{array}\right]
$$

where $Y_{S n 1}=\frac{Y_{S 1} \cdot Y_{S n}}{Y_{S 1}+Y_{S n}}$

Since source and stack tuner have been assumed in parallel for the digital simulation purposes, elements of both matrices can be added for the combined admittance purpose shown as below:

$$
\left[\begin{array}{lll}
\left(\mathrm{Y}_{\mathrm{Sn} 1}+\mathrm{Y}_{\mathrm{ST}}\right) & \mathrm{y}_{\mathrm{Sn}} & \mathrm{Y}_{\mathrm{Sn}}  \tag{3.73}\\
\mathrm{Y}_{\mathrm{Sn}} & \left(\mathrm{Y}_{\mathrm{Sn} 1}+\mathrm{Y}_{\mathrm{ST}}\right) & \mathrm{Y}_{\mathrm{Sn}} \\
\mathrm{Y}_{\mathrm{Sn}} & \mathrm{Y}_{\mathrm{Sn}} & \left(\mathrm{Y}_{\mathrm{Sn} 1}+\mathrm{Y}_{\mathrm{ST}}\right)
\end{array}\right]
$$

If $\quad\left(\mathrm{Y}_{\mathrm{Sn} 1}+\mathrm{Y}_{\mathrm{ST}}\right)=\mathrm{Y}_{\mathrm{TT}}$ as total admittance, the above matrix can be written as below:

$$
\left[\begin{array}{ccc}
Y_{T T} & Y_{S n} & Y_{S n}  \tag{3.74}\\
Y_{S n} & Y_{T T} & Y_{S n} \\
Y_{S n} & Y_{S n} & Y_{T T}
\end{array}\right]
$$

Now inverting all the elements of this matrix, impedance matrix can be re-written as:

$$
\left[\begin{array}{lll}
\mathrm{z}_{\mathrm{TT}} & { }^{\mathrm{z}_{\mathrm{Sn}}} & { }^{z_{S n}}  \tag{3.75}\\
\mathrm{z}_{\mathrm{Sn}} & \mathrm{z}_{\mathrm{TT}} & { }^{z_{S n}} \\
\mathrm{z}_{\mathrm{Sn}} & \mathrm{z}_{\mathrm{Sn}} & { }^{z_{T T}}
\end{array}\right]
$$

Equation 3.66 can be re-written as:
$\left[\begin{array}{l}\overline{\mathrm{I}}_{\mathrm{Sa}} \\ \overline{\mathrm{I}}_{\mathrm{Sb}} \\ \overline{\mathrm{I}}_{\mathrm{Sc}}\end{array}\right] \quad\left[\begin{array}{lll}\mathrm{z}_{\mathrm{TT}} & \mathrm{z}_{\mathrm{Sn}} & \mathrm{z}_{\mathrm{Sn}} \\ \mathrm{z}_{\mathrm{Sn}} & \mathrm{z}_{\mathrm{TT}} & \mathrm{z}_{\mathrm{Sn}} \\ \mathrm{Z}_{\mathrm{Sn}} & \mathrm{z}_{\mathrm{Sn}} & \mathrm{z}_{\mathrm{TT}}\end{array}\right] \quad\left[\begin{array}{l}\overline{\mathrm{V}}_{\mathrm{Sa}} \\ \overline{\mathrm{V}}_{\mathrm{Sb}} \\ \overline{\mathrm{V}}_{\mathrm{SC}}\end{array}\right]$

Now this source impedance matrix which includes the impedance of the stack tuner in close or open mode as the case may be, can be used as explained in Chapter 3.2.

### 3.3.1 Fault simulation studies

Figure 3.30 and Figure 3.31 are the transient responses with the tuned circuit at SE and RE respectively. When these responses are compared with the responses of Figure 3.27 and Figure 3.28 , it can be seen that there is no difference in the waveforms. This study concludes that the stack tuner does not affect the fault induced transient responses and can be used successfully for detecting the high frequency components of the fault induced transients in a particular frequency bandwidth.

### 3.4 Digital simulation of the bus bar capacitance and stack tuner with the transmission line

When the bus bar to earth capacitance and the tuned circuit both are considered, the simplified arrangement can be shown as in Figure 3.32. When sources are earthed at the two ends of the transmission line, all the three circuits: the source, the tuned circuit and the bus bar to earth impedances can be considered in parallel. Alternatively, the admittances of all the three circuits can be added directly. Admittance matrix for the bus bar to earth admittance can be written as:

$$
\left[\begin{array}{lcl}
\mathrm{Y}_{\mathrm{CB}} & 0 & 0  \tag{3.77}\\
0 & \mathrm{Y}_{\mathrm{CB}} & 0 \\
0 & 0 & \mathrm{Y}_{\mathrm{CB}}
\end{array}\right]
$$

where $Y_{C B}$ is bus bar to earth admittance.
The source admittance matrix considering the tuned circuit admittance and the bus bar to earth admittance can be written as below after combining all the three admittances i.e. matrices (3.73) and (3.77):

$$
\left[\begin{array}{lccc}
\left(Y_{S n 1}+Y_{S T}+Y_{C B}\right) & Y_{S n} & Y_{S n}  \tag{3.78}\\
Y_{S n} & \left(Y_{S n 1}+\right. & \left.Y_{S T}+Y_{C B}\right) & Y_{S n} \\
Y_{S n} & & Y_{S n} & \left(Y_{S n 1}+\right. \\
\left.Y_{S T}+Y_{C B}\right)
\end{array}\right]
$$

If $\left(Y_{S n 1^{+}} Y_{S T}+Y_{C B}\right)=Y_{T T}$ as total admittance, the above matrix is of the form of Matrix (3.74). Now inverting all the elements of this matrix, the impedance matrix is of the form of Matrix (3.75), and Equation (3.76) can be established.
3.41 Simulated test results

Figure 3.33 and Figure 3.34 show the transient responses with the tuner and the bus bar capacitance. A negligible amount of the phase shift has been noticed. The magnitude has been also affected but the effect is negligible. The transmission system and the fault location are same as in Section 3.2.10.

### 3.5 Digital simulation of output of the stack tuner

From Chapter 3.1.7, gain in closed mode, YNC is

YNC $=\mathrm{V}_{\text {OUT }} / \mathrm{V}_{\text {IN }}=\mathrm{YSM} / \mathrm{YTSM}$
and gain in open mode, YNO is

YNO $=\mathrm{V}_{\text {OUT }} / \mathrm{V}_{\text {IN }}=\mathrm{YOM} / \mathrm{YTM}$

YNC and YNO are in the 'phasor form'.
Voltages at the relaying points can be called as sending end (SE) and receiving end (RE) voltages. These are in two
components: (1) steady state, (2) superimposed. For pre fault voltages, steady state voltage as simulated in Equation (3.42) is multiplied by YNC for voltages in close mode and by YNO for voltages in open mode. These can be written as below:

For close mode,
$\overline{\mathrm{V}}_{\mathrm{fS}}{ }^{\text {new }}=\left(\overline{\mathrm{V}}_{\mathrm{fS}}\right) \quad \mathrm{YNC}$

For open mode,
$\overline{\mathrm{V}}_{\mathrm{fS}}$ new $=\left(\overline{\mathrm{V}}_{\mathrm{fS}}\right)$ YNO

For post fault calculations, only superimposed components have been taken into account due to interest of high frequency components. These are calculated using Equation (3.52) at SE and Equation (3.62) at RE as below: For close mode,
$\bar{v}_{S f}$ newc $=\left(\overline{\mathrm{V}}_{S f}\right)$ YNC at SE
$\overline{\mathrm{V}}_{\mathrm{Rf}}$ newc $=\left(\overline{\mathrm{V}}_{\mathrm{Rf}}\right)$ YNC at RE

For open mode,
$\overline{\mathrm{v}}_{\mathrm{Sf}}$ newo $=\left(\overline{\mathrm{v}}_{\mathrm{Sf}}\right)$ YNO at SE
$\overline{\mathrm{V}}_{\mathrm{Rf}}$ newo $=\left(\overline{\mathrm{V}}_{\mathrm{Rf}}\right)$ YNO at RE

These evaluations are in frequency domain. The superimposed voltages in time domain are evaluated using Modified Fourier Transform Technique as given by Equation (3.38).


Figure 3.1 A CONVENTIONAL STACK TUNER CIRCUIT


Figure 3.2 PHASE TO PHASE COUPLING EQUIPMENT REGARDED AS BAND PASS EILTER


Figure 3.3 WIDE BAND ARRANGEMENT IN A BAND PASS FILTER


Figure 3.4 EQUIVALENT CIRCUIT OF THE LINE UPTO 275 kV SYSTEM

'b' The equivalent circuit

Figure 3.5 PHASE TO PHASE COUPLING FOR A 400 kV SYSTEM


Figure 3.6 A SIMPLE ARRANGEMENT OF FIGURE 3.5


Figure 3.7 FREQUENCY RESPONSE OF A PARALLEL TUNED CIRCUIT


Figure 3.8 SIMPLIFIED CIRCUIT FOR CALCULATION OF RT


Figure 3.9 EFEECT OF $\mathrm{R}_{\mathrm{T}}$ ON BANDWIDTH


Figure 3.10 ADDITIONAL DAMP ING ARRANGEMENT


Figure 3.11 EFFECT ON BANDWIDTH DUE TO ADDITIONAL DAMPING


Figure 3.12 BASIC CIRCUIT ARRANGEMENT


Figure 3.13 CIRCUIT DIAGRAM OF THE TUNER


Figure 3.14 EQUIVALENT CIRCUIT IN CLOSE MODE


Figure 3.15 EQUIVALENT CIRCUIT IN OPEN MODE


1S: SHORT MODE, MID-FREQ $300 \pm 2.5 \mathrm{kHz}$
10: OPEN MODE, MID-EREQ $300 \pm 2.5 \mathrm{kHz}$
2S: SHORT MODE, MID-EREQ $200 \pm 2.5 \mathrm{kHz}$
20: OPEN MODE, MID-FREQ $200 \pm 2.5 \mathrm{kHz}$
3S: SHORT MODE, MID-FREQ $100 \pm 2.5 \mathrm{kHz}$
30: OPEN MODE, MID-FREQ $100 \pm 2.5 \mathrm{kHz}$
4S: SHORT MODE, MID-FREQ $200 \pm 100 \mathrm{kHz}$
40: OPEN MODE, MID-FREQ $200 \pm 100 \mathrm{kHz}$
5S: SHORT MODE, MID-FREQ $50 \pm 40 \mathrm{kHz}$
50: OPEN MODE, MID-FREQ $50 \pm 40 \mathrm{kHz}$
Figure 3.16 GAIN Vs FREQUENCY CHARACTERISTICS OF THE STACK TUNER


Figure 3.17 OPEN MODE IMPEDANCE CHARACIERISTICS OF THE S'LACK TUNER AT 300, 200, 100 kHz


Figure 3.18 SHORI MODE TMPEDANCE CHARACTERISTICS OE THE STACK IUNER AL $300,200,100 \mathrm{kHz}$


Eigure 3.19 OPEN / SHORT MODE TMPEDANCE CHARACIERJSTICS OE THE STACK 'IUNER EOR 100-300. $10-90 \mathrm{kNz}$


Figure 3.20 TRANSMISSION LINE RERESENTATION BY THE 2-PORT
MATRIX FORMATION


Figure 3.21 REPRESENTATION OF THE FAULTED TRANSMISSION LINE


Figure 3.22 REPRESENTATION OF THE TRANSMISSION LINE WITH
CONSIDERATION OF AN INTERNAL FAULT AT
DISTANCE X FROM SENDING END


Figure 3.23 MAIN SOURCE NETWORK MODEL


Eigure 3.24 LINE CONFIGURATION




Figure 3.27 SE Transient Response with SF 819.2 kHz



Figure $3.29 \frac{\text { REPRESENTATION OF STACK TUNER WITH }}{\text { TRANSMISSION LINE }}$


Figure 3.30 SE Transient Response with SF 819.2 kHz


Figure 3.31 RE Transient response with SF 819.2 kHz


Eigure 3.32 REPRESENTATION OF STACK TUNER WITH TRANSMISSION LINE CONSIDERING BUS BAR CAPACITANCE



Figure 3.34 RE Transient Response with $\mathbf{S F} \mathbf{8 1 9 . 2 k H z}$

## CHAPTER 4

## INVESTIGATIONS BASED ON STACK-TUNER-OUTPUT-RESPONSES OF THE FAULTED PHASE ONLY

This chapter investigates the stack tuner output (Figure 3.14 and Figure 3.15) responses of the faulted phase only at both ends of the transmission line. Firstly, simulated tests have been performed for different mid-frequencies of the stack tuner and then a particular mid-frequency has been selected based on the response and the design criterion. The establishment and the evaluation of the basic principle of the scheme have been described. The signal processing scheme used for investigations in this chapter is explained. Application of the auto-correlation technique in the signal processing has been highlighted. In the scheme (reference Chapter 2) which is to be investigated, a directional detector, DD has been used for the purpose of the discrimination of the reverse fault simply by detecting the direction of the fault. This directioal detector will have certain operating time delay. In the present scheme of the signal processing, this time delay has been taken in to consideration.

As it is a well known fact that the travelling wave based schemes are failures for the faults at lower inception angle, the fault responses at different fault inception angles of the voltage waveform have been investigated. The effect of diffent source capacities has been also investigated.

For all the simulated tests, the stack tuner has been considered in the open mode. In reverse fault condition only, the stack tuner has been simulated in normally close mode at the sending end, SE, only. For all the cases, a single phase to earth fault has been considered at 1 km from the $S E$ of a 100 km long 400 kV transmission line with end sources of 5 GVA at SE and 20 GVA at RE. Time of faultobservation is 600 micro-seconds. The sampling frequency is 853.33 kHz . The fault inception angle of the voltage is 90 degrees. The performance of the simulated tests have been described on the basis of the positive peak value of the response during the post-fault observation time. Later few results are based on the fault-observation time of 5 msec .

### 4.1 Fault response at mid-frquency of $300 \pm 2.5 \mathrm{kHz}, \quad 200 \pm 2.5$ $\mathrm{kHz}, 100 \pm 2,5 \mathrm{kHz}, 200 \pm 100 \mathrm{kHz}, 50 \pm 40 \mathrm{kHz}$ of the stack tuner and selection of an appropriate tuner mid-frequency

Figure 4.1 shows the fault responses at various midfrequencies of the stack tuner: $300 \pm 2.5 \mathrm{kHz}, 200 \pm 2.5 \mathrm{kHz}$, $100 \pm 2.5 \mathrm{kHz}, 200 \pm 100 \mathrm{kHz}, 50 \pm 40 \mathrm{kHz}$ at SE and RE. It can be seen that the peak value of the response at mid-frequency
of $100 \pm 2.5 \mathrm{kHz}$ of the stack tuner is about twice of the peak value of the response at the mid-frequency of $200 \pm 2.5$ kHz of the stack tuner. Similarly, the peak response at the mid-frequency of $200 \pm 2.5 \mathrm{kHz}$ has been found to be about twice of the peak response at the mid-frequency of $300 \pm 2.5$ kHz at the sending end, SE. This fact can be established theoretically as below:

A technique of the digital simulation of the superimposed component of the voltage at the fault-point has been explained in Chapter 3.2.3 and Chapter 3.2.4. Equation (3.35) for the superimposed component of the voltage in the frequency domain is -

$$
\begin{equation*}
\bar{V}_{f f}(j w)=\hat{V}_{f s} \frac{{ }_{o} \cos \Theta_{f s}}{w^{2}-w_{o}^{2}}+\frac{j w \sin \Theta_{\hat{i}}}{w^{2}-w_{o}^{2}} \tag{4.1}
\end{equation*}
$$

In this expression, $\Theta_{f_{S}}$ is the phase angle which determine the point-on-the pre-fault waveform of the steady state voltage at which the fault occurs.

For the case of the fault inception angle $\Theta_{f s}=90$ degree, there is an extreme limit of the voltage, called as voltage maximum fault, $\bar{V}_{\text {ffMAX }}$ which is given by

$$
\begin{equation*}
\bar{V}_{\mathrm{ffMAX}}(w)=\hat{V}_{f s} \frac{w}{w^{2}-w_{o}} 2 \tag{4.2}
\end{equation*}
$$

where 'wo' is power frequency in radians and ' $w$ ' is frequency in radians. When $w=w_{o}, \quad \bar{V}_{f f M A X}$ is infinite.

In the present case, mid frequencies of the stack tuner are being considered in the range of 100 to 300 kHz which is of course very much greater than the power frequency and the above Equation (4.2) can be simplified as

$$
\begin{equation*}
\overline{\mathrm{V}}_{\mathrm{ffMAX}}(\mathrm{w})=\hat{\mathrm{V}}_{\mathrm{fs}} \frac{1}{\mathrm{w}} \tag{4.3}
\end{equation*}
$$

for different mid-frequemcies: (1) $100 \mathrm{kHz}, \mathrm{w}=2 \pi 10^{5}$, (2) $200 \mathrm{kHz}, \mathrm{w}=2 \pi \times 2 \times 10^{5}$, (3) $300 \mathrm{kHz}, \mathrm{w}=2 \pi \times 3 \times 10^{5}$, and above relation (4.3) can be written as for (1) of 100 kHz as

$$
\begin{equation*}
\overline{\mathrm{V}}_{\mathrm{ffMAX}}(\mathrm{w})=\mathrm{C}_{\mathrm{V}} / 1 \tag{4.4}
\end{equation*}
$$

for (2) of 200 kHz as

$$
\begin{equation*}
\overline{\mathrm{V}}_{\mathrm{ffMAX}}(\mathrm{w})=\mathrm{C}_{\mathrm{V}} / 2 \tag{4.5}
\end{equation*}
$$

for (3) of 300 kHz as

$$
\begin{equation*}
\bar{v}_{\mathrm{ffMAX}}(w)=C_{V} / 3 \tag{4.6}
\end{equation*}
$$

where $C_{V}$ is a voltage constant $=\hat{\mathrm{V}}_{\mathrm{fS}} / 2 \pi \times 10^{5}$
equations (4.4) and (4.5) establish that the $\overline{\mathrm{V}}_{\mathrm{ffMAX}}(w)$ for the mid-frequency of 100 kHz is twice of the $\overline{\mathrm{V}}_{\mathrm{ffMAX}}(\mathrm{w})$ for the mid-frequency of 200 kHz . Similarly equations (4.5) and (4.6) establish that the $\bar{v}_{\text {ffMAX }}(w)$ for the midfrequency of 200 kHz is 1.5 times the $\overline{\mathrm{V}}_{\text {fimAX }}(\mathrm{w})$ for the
mid-frequency of 300 kHz .
The peak values of the response for other mid-frequencies of $200 \pm 100 \mathrm{kHz}$ and $50 \pm 40 \mathrm{kHz}$ have been found very small in comparison to the responses for other mid-frequencies. Therefore, these two mid-frequencies $(200 \pm 100 \mathrm{kHz}$ and $50 \pm 40$ kHz ) have not been considered for the simulated tests. Based on the peak values of the fault response, the midfrequency of $100 \pm 2.5 \mathrm{kHz}$ can be selected for the stack tuner. But, the value of the capacitance of the coupling capacitor (reference Chapter 2.1) has been assumed constant and a larger value of the inductance of the stack tuning coil is required for resonance at the mid-frequency of $100 \pm 2.5 \mathrm{kHz}$ in comparison to the mid-frequency of $300 \pm 2.5$ kHz . This will result in increase of the size and the cost of the equipment. Therefore, the mid-frequency of $300 \pm 2.5$ kHz of the stack tuner has been selected for all the simulated tests in this thesis.

### 4.2 Establishment of the new principle

Figure 4.2 a shows the voltage waveforms at the stack tuner output in NO, NO mode at both ends SE and RE for an internal fault at peak voltage for phase 'a' to earth fault. Figure 4.2b shows the voltage waveforms at the stack tuner output in NC, NO mode at the ends $S E$ and RE for reverse fault (external fault). A positive peak voltage of 300 Volts at RE in Figure 4.2 b has been found in comparison to 800 Volts at RE in Figure 4.2a. At SE, the positive peak voltages are 7000 Volts and 3500 Volts for
internal and reverse faults. In such a case, if a threshold voltage of the level detector (reference Chapter 2.2: assuming the faulted phase signal from the output of the stack tuner goes direct to the level detector) corresponding to a positive peak of about 400 Volts can be arranged, then discrimination at the RE for reverse fault is possible. At the $S E$, the discrimination for reverse fault is taken care by the directional detector, DD (reference Chapter 2.2). In such a case, the decision logic (Figure 2.2) inputs are $\mathrm{V}_{\mathrm{D}}=1, \mathrm{~T}_{\mathrm{F}}=1, \mathrm{~T}_{\mathrm{R}}=0$ at SE and $R E$ for internal faults which results in a $\operatorname{TRIP}=1$, signal at the output of the decision logic. Similarly for the reverse fault, the decision logic inputs are $V_{D}=1$, $T_{F}=0, T_{R}=1$ at $S E$ which results in a signal, $\operatorname{TRIP}=0$ and prevents the tripping at the SE. Also at RE for the reverse fault, the decision logic inputs are $\mathrm{V}_{\mathrm{D}}=0, \mathrm{~T}_{\mathrm{F}}=$ 1, $\mathrm{T}_{\mathrm{R}}=0$ which results in a $\operatorname{TRIP}=0$, signal and prevents the tripping at RE. With this criterion, the scheme has been established at least in principle. Detailed investigation on the basis of the signal processing (reference Chapter 2.2) is carried out later on. These results are based on the post-fault observation time of 600 micro-seconds. If the operating time of the directional detector, dd, typically say 1 msec is considered, it is necessary to study the fault responses at least for a quarter cycle to establish the scheme satisfactorily. With this view, Figure 4.3 shows the fault responses for a postfault time of 5 msec . It can be seen from the responses in Figure 4.3 that the fault response repeats after a
particular time and reduces in magnitude each time when it repeats. This particular time of repeat corresponds to the time taken by the travelling-wave to travel a distance between the fault point and the end of the transmission line i.e. SE or RE. Using the same criterion as in Figure 4.2 for the internal and the reverse faults based on the positive peak responses, a discrimination has been found to be possible and, the scheme at least in principle has been established satisfactorily.

### 4.3 Signal processing and investigations of the signals

Figure 4.4 shows the block representation of the signal processing which has been used for investigation purposes in this chapter. The output of the stack tuner, $V_{F}$ is sent to an auto-correlator block. The output signal of the auto-correlator, $V_{A}$ is sent to another block 'DD operating delay'. Now the signal $V_{A}$ is delayed by the operating time of the $D D$ i.e. $T_{D D}$. In the present investigation, this delay $T_{D D}$ has been taken as 1 ms which is a typical value. This delayed signal is $X$. This signal $X$ is squared and integrated with time in another block for the purpose of deciding the level of the signal. This signal is y. This $y$ signal is sent to another block, called as the level detector. In this block, the signal $y$ is compared with a threshold value and the output $V_{D}$ is sent to another block 'Decision logic' which also receives a signal $T_{F, R}$ from the directional detector. If both the signals are 1, the trip signal is sent for the tripping of the local breaker.

The principle of the auto-correlation has been described as below:

The discrete cross-correlation function $P_{X Y}(T)$ measures the correlation between sections of sampled signal $x$ and the delayed signal $y$ as a function of the delay $T$
$P_{x y}(T)=1 / N \sum_{k=1}^{N} x(k \delta t+T) \cdot y(k \delta t)$
where $k=1,2, \ldots-N$
$\delta t=$ sampling time
if $T=m \delta t$ where $m$ can vary from 0 to $N-1$, then $N$ values of $P_{x y}$ can be found out.

The correlation function $P_{x y}(T)$ shows the similarity between two waveshapes as a function of the time shift between them. Maximum output of the correlation function indicates the best match between the waveshapes.

Now if signal $y$ is the signal $x$ i.e. the signal ' $x$ ' is being correlated against itself then the technique is called an auto-correlation technique and the auto-co rrelation function can be written as:

$$
\begin{aligned}
P_{\mathrm{xx}}(\mathrm{~T}) & =1 / \mathrm{N} \sum_{\mathrm{k}=1}^{\mathrm{N}} \mathrm{x}(\mathrm{k} \delta \mathrm{t}+\mathrm{T}) \cdot \mathrm{x}(\mathrm{k} \delta \mathrm{t}) \\
\text { where } \mathrm{k} & =1,2, \cdots-\mathrm{N} \\
\delta \mathrm{t} & =\text { sampling time }
\end{aligned}
$$

if $T=m \delta t$ where $m$ can vary from 0 to $N-1$, then $N$ values of $P_{x x}$ can be found out. $x(k \delta t)$ is known as the window of $N$ samples of the signal $x$.

The correlation function $P_{x x}(T)$ shows the similarity between two waveshapes as a function of the time shift between them. Maximum output of the correlation function indicates the best match between the waveshapes.

Depending upon the value of the 'm' in the expression
$T=m \delta t$, the signal ' $x$ ' can be started from any sampling point for the purpose of the auto-correlated response. If there are 100 sampling points in a signal 'x'and 'm'is 51, then the signal 'x' can be auto-correlated with the signal ' $x$ ' after a delay of half the signal time. If 'm' $=0$, the signal 'x' can be auto co-rrelated with the same signal 'x' without any delay provided window $x(k \delta t)$ is available. This means that first half of the signal 'x' can be autocorrelated with the second half of the same signal 'x' if m $=$ 51. The choice for selection of 'm' depends upon the duration and waveform of the signal.

In the present case, the application of the autocorrelation technique is being investigated for the purpose of the increase in the discrimination factor (explained later in this section) and the effect of the DD operating time has not been considered on the fault responses. With this view, Figure 4.5 shows the auto-correlated responses for all the signals shown in Figure 4.3. For these responses 'm' is 0 as explained above and the response $V_{A}$
is only for a period of 2.4 msec which is about half the signal time, and has been considered sufficient for the purpose of investigation. The delay $T_{D D}$ due to the operating time of the directional detector, $D D$, is clearly shown in Figure 4.5. The response after a delay, $T_{D D}$ is 'x' as shown in Figure 4.5. These signals 'x' are squared and integrated with respect to time and the responses are called 'y' which are shown in Figure 4.6. It can be seen that the factor of discrimination has been improved to about 30. Now these signals are processed to a level detector whose threshold value $\mathrm{V}_{\mathrm{T}}$ (reference Chapter 2) can be set to a value larger than the value corresponding to the response ' $y$ ' at the receiving end in the reverse fault condition. As explained in Chapter 2, the output of the level detector, $V_{D}$ is processed to a 'logic gate'which also receives a signal from the directional detector, dd. The output of the 'logic gate' gives a tripping signal for the internal faults only.
4.4 Effect of different fault inception angles of the
voltage

A technique of the digital simulation of the superimposed component of the voltage at the fault-point has been explained in Chapter 3.2.3 and Chapter 3.2.4. Equation (3.35) for the superimposed component of the voltage in the frequency domain is being re-written as below:

$$
\begin{equation*}
\bar{v}_{f f}(j w)=\hat{V}_{f s} \frac{w_{o} \cos \Theta_{\hat{I} S}}{w^{2}-w_{o}^{2}}+\frac{j w \sin \Theta_{f s}}{w^{2}-w_{o}^{2}} \tag{4.9}
\end{equation*}
$$

In this expression, $\Theta_{f S}$ is the phase angle which determine the point-on-the pre-fault waveform of the steady state voltage at which the fault occurs. In Equation (4.9), there are two extreme conditions for Vff(jw): (1) a minimum, $\dot{\mathrm{V}}_{\mathrm{ffMIN}}(\mathrm{w})$ corresponding to the value of $\Theta_{\mathrm{fS}}=0$, and (2) a maximum, $\overline{\mathrm{V}}_{\mathrm{ffMAX}}(w)$ corresponding to the value of $\Theta_{f s}=90$ degree which can be written as

$$
\begin{equation*}
\bar{v}_{f f M I N}(w)=\bar{v}_{f S} \frac{w_{o}}{w^{2}-w_{o}^{2}} \tag{4.10}
\end{equation*}
$$

and

$$
\begin{equation*}
\bar{v}_{f f M A X}(w)=\hat{V}_{f s} \frac{w}{w^{2}-w_{o}^{2}} \tag{4.11}
\end{equation*}
$$

where 'wo' is the power frequency in radians and 'w' is the frequency in radians. Equation (4.11) is same as Equation (4.2) explained earlier. When $w=w_{0} \bar{v}_{\text {ff MAX }}(w)$ and $\overline{\mathrm{V}}_{\mathrm{ffMIN}}(\mathrm{w})$ are infinite. But, when $w \gg_{o}$, above equations (4.9), (4.10), (4.11) can be simplified as below:

$$
\begin{aligned}
& \overline{\mathrm{V}}_{\mathrm{ff}}(\mathrm{w})=\frac{\hat{\mathrm{V}}_{\mathrm{fs}}}{\mathrm{w}} \sin \Theta_{\mathrm{fS}} \\
& \overline{\mathrm{~V}}_{\mathrm{ffMIN}}(\mathrm{w})=\hat{\mathrm{V}}_{\mathrm{fS}} \frac{1}{\mathrm{w}^{2}} \\
& \text { and }
\end{aligned}
$$

$$
\begin{equation*}
\bar{v}_{f f M A X}(w)=\hat{v}_{f S} \frac{1}{w} \tag{4.14}
\end{equation*}
$$

It can be seen from the equations (4.13) and (4.14) that $\bar{v}_{f f M A X}(w)$ is 'w' times the $\bar{v}_{f f M I N}(w)$. This means that the response at zero voltage point-on-wave faults can be improved by selecting a lower value of the mid-frequency of the stack tuner. Equation (4.12) gives the fault responses for various values of the $\Theta_{f s}$, and is proportional to the value of the $\sin \Theta_{f s}$ for a particular voltage of the transmission system and a particular mid-frequency of the stack tuner as shown below:

| $\Theta_{f S}$ | $\sin \Theta_{f s}$ |
| :--- | :--- |
| 90 | 1 |
| 80 | 0.984 |
| 70 | 0.939 |
| 60 | 0.866 |
| 50 | 0.766 |
| 45 | 0.707 |
| 40 | 0.642 |
| 30 | 0.5 |
| 20 | 0.342 |
| 15 | 0.258 |
| 10 | 0.173 |
| 5 | 0.087 |
| 1 | 0.017 |

This shows that the response at 1 degree voltage point-onwave faults is reduced to 1.7 percent of the response at 90 degree voltage point-on-wave faults at higher midfrequencies of the stack tuner.

Figures 4.7a, b, c, d, e, f show the stack tuner output responses for different point-on-wave faults of the voltage: (1) 90 degree, (2) 60 degree, (3) 30 degree, (4) 15 degree, (5) 5 degree, and (6) 0.

A table as below shows (1) the positive peak responses at sending end, $S E,(2)$ its percentage of the maximum voltage point-on-wave fault, (3) the positive peak response at the receiving end, RE, (4) its percentage of the maximum voltage point-on-wave faults, (5) calculated value of the $\sin \Theta_{f s}$ from the above table, and (6) its percentage of the maximum voltage point-on-wave fault for different point-onwave faults: $90,60,30,15,5$, and 0 .

TABLE

| $\Theta_{f s}$ | $\overline{\mathrm{~V}}_{\mathrm{ff}}(\mathrm{SE})$ | $\%$ | $\overline{\mathrm{~V}}_{\mathrm{ff}}(\mathrm{RE})$ | $\%$ | $\sin \Theta_{\mathrm{fS}}$ | $\%$ |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
|  | (Volts) |  | (Volts) |  |  | 100 |
| 90 | 7000 | 100 | 800 | 100 | 1 | 86.6 |
| 60 | 6000 | 85.7 | 700 | 87.5 | 0.866 | 86 |
| 30 | 3500 | 50 | 400 | 50 | 0.5 | 50 |
| 15 | 1800 | 25.7 | 200 | 25 | 0.258 | 25.8 |
| 5 | 600 | 8.5 | 70 | 8.7 | 0.087 | 8.7 |

This table shows a good agreement between the calculated values and the simulated test results. Now Equation (4.12) can be used as a useful tool for designing the scheme and to find out the point-on-wave of the voltage for which the scheme can operate satisfactorily from the discrimination point of view for reverse fault conditions. For example, the positive peak voltages at $R E$ in Figure 4.2 a is 800 Volts and in Figure 4.2 b is 300 Volts which is 37.5 percent of 800 Volts. If a margin of safety of 12.5 percent is considered then for a total value of 50 percent, point-on-wave of the voltage of 30 degree can be considered safely for the operation of the scheme as the value of sin $\Theta_{f s}=\sin 30=0.5$ i.e. 50 percent. This concludes that the scheme can work satisfactorily up to 30 degree of the fault inception angle of the voltage. The performance of the scheme can be improved for operation at lower value of the $\Theta_{f s}$ either by reducing the output voltage gain of the stack tuner in the open mode, NO, or by increasing the gain in the short mode, NC (reference Chapter 3.1.5 and Chapter 3.1.6) depending on the design of the stack tuner circuit. Regarding the performance of the scheme at zero voltage point-on-wave faults, the simulated test results have been shown in the Figure 4.7 f and it can be seen that the value of the response is so low that the scheme can not discriminate under reverse fault conditions. This fact has been established earlier from the equations (4.12) and (4.13) theoretically as the the $\overline{\mathrm{V}}_{\mathrm{ffMAX}}(w)$ is 'w' times the $\overline{\mathrm{V}}_{\text {ffMIN }}(\mathrm{w})$. According to Mehdi [21], zero voltage point-onwave faults are relevent to theory, but in practice modern

EHV systems have very high impulse insulation levels which causes almost all flashovers to occur near voltage maximum. However another method of improving the performance of the scheme at a lower value of the $\Theta_{f s}$ or zero voltage point-on-wave faults will be discussed later considering arcing faults.

### 4.5 Effect of different source capacities

The newly developed scheme is based on the detection of the high frequency signals ( 300 kHz in the present case) from the fault generated noise at the ends of the transmission line. Whenever a fault or disturbance occurs on the transmission line, the voltage and current waveforms travel from the point of the disturbance to the ends of the transmission line due to a well known travelling wave phenomena. As the scheme is based on the detection of the voltage signals, only voltage waveforms have been considered in this thesis. Depending on the distance of fault from the end, the voltage signal is attenuated due to losses in the line. When the voltage signal reaches to the end of the line, part of the signal is reflected towards the fault and a part of the signal is refracted towards the source which is again reflected towards the fault, and the process cotinues until the voltage signal fully dies out. The refracted component of the signal depends upon the well known refracted coefficient of the transmission line which is given by:

$$
\begin{equation*}
R_{r}=2 Z_{o} /\left(Z_{O}+Z_{L}\right) \tag{4.15}
\end{equation*}
$$

where $Z_{o}$ is the surge impedance of the line and $z_{L}$ is the source impedance. This source impedance varies depending on the capacities of the sources connected at the ends of the transmission line. Therefore the high frequency signals at the ends depend on the source capacities. With this aim, the newly developed scheme has been digitally simulated and tested for satisfactory operation with different source capacities.

The surge impedance, $Z_{o}$ of the transmission line under the scope of this thesis has been calculated and found to be of the order of 370 Ohm . Based on the conventional values of the source capacities: 5 GVA to 30 GVA, the source impedance, $\mathrm{Z}_{\mathrm{L}}$ may vary from 32 Ohm to 5 Ohm for a 400 kV line, and, the corrosponding refracted coefficient, $R_{r}$ varies from 1.86 to 1.98. With a view to study the performance of the simulated tests under extreme conditions, the simulated tests have been carried out with source capacities (1) 1 GVA at both ends, (2) 99 GVA at both ends, (3) 5 GVA at the sending end, SE, and 20 GVA at the receiving end, RE. The results of these simulated tests are shown in Figure 4.8. It can be seen in Figure 4.8 that there is no appreciable difference in the responses at both the ends.

Therefore, it can be concluded that the newly developed scheme can be used satisfactorily for different combinations of source capacities for the particular system considered in this thesis.


Edgure 4.1 EAUL'T RESDONSES AT DIEEEREN'I EREQUENCIES
cont.

$\mathrm{d}: 200 \pm 100 \mathrm{kHz}$



$\mathrm{e}: 50 \pm 40 \mathrm{kHz}$

$$
\begin{aligned}
& \mathrm{X} \text { axis: } \frac{\text { TIME }- \text { Seconds }}{\mathrm{Y} \text { axis: } \frac{\text { VOLTAGES }- \text { Volts }}{\text { (TUNER OUTPUT) }}} .
\end{aligned}
$$

Eigure 4.1 EAUL'T RESPONSES AT DIFFERENT

a: Internal fault

```
VD=1,TF=1
    VD=1,TE=1
TR=0,TRIP=1
b: Reverse fault
TR=0,TRIP=1
```

$\mathrm{VD}=1, \mathrm{TE}=0$
$T R=1, \operatorname{TRIP}=0$
$\mathrm{VD}=1, \mathrm{TE}=0$
$T R=1, \operatorname{TRIP}=0$
$\mathrm{VD}=0, \mathrm{TE}=1$
$\mathrm{VD}=1, \mathrm{TE}=1$
$T R=0, T R I P=1$
$\mathrm{VD}=0, \mathrm{TF}=1$
$T R=0, T R I P=0$

> X axis: TIME - Seconds
> Y axis: VOLTAGES - Volts (TUNER OUTPUT)

Eigure 4.2 ESTABLISHMENT OF THE PRINCIPLE (STACK-TUNER MID-EREQUENCY $300 \pm 2.5 \mathrm{kHz}$ )



b: Reverse Fault
X-Axis: TIME-Seconds
Y-Axis: TUNER OUTPUT VOLTAGE- Volts

Figure 4.3 DISCRIMINATION IES'I BASED ON RESPONSES FOR 5 MILLI-SECONDS (STACK-TUNER MID-FREQUENCY $300 \pm 2.5 \mathrm{kHz})$


Figure 4.4 BLOCK DIAGRAM REPRESENTATION FOR SIGNAL PROCESSING

a: Internal Fault

(DELAY DUE TO $x(k S t)$ NOT CONSIDERED - REF. CHAPTER 4.3)

Figure 4.5 AUTO-CORRELATED RESPONSES OF FIGURE 4.3


b: Reverse Fault
$X-A x i s:$ TIME-Seconds
Y-Axis: RESPONSE ' $y$ ' VOLTAGE- Units

Eigure 4.6 RESPONSES ' $y$ ' EOR DISCRIMINATION TEST




b: 60 Degrees


C: 30 Degrees

$$
\begin{aligned}
& \text { X axis: TIME - Seconds } \\
& \text { Y axis: VOLTAGES - Volts } \\
& \text { (TUNER OUTPUT) }
\end{aligned}
$$

Figure 4.7 EFEECT OF DIFFERENT FAULT INCEP'TION ANGLES

Cont.



e: 5 Degrees


f: 0 Deqrees

$$
\begin{aligned}
& \text { X axis: TIME - Seconds } \\
& \text { Y axis: VOLTAGES - Volts } \\
& \text { (TUNER OUTPUT) }
\end{aligned}
$$

Figure $4.7 \begin{gathered}\text { EFFECT OF DIEFEREN' } \\ \text { INCEDTION ANGLES }\end{gathered}$


Figure 4.8 EFEECT OF DIFFERENT SOURCE CAPACITIES

## CHAPTER 5

## NATURAL MODES OF PROPAGATION, DIGITAL SIMULATION AND <br> SIMULATED TEST RESULTS

This chapter describes the natural modes of propagation and its application to get a composite signal to establish the basic principle of the scheme and to detect any kind of fault. A technique for the digital simulation of the natural modes has been explained. Further, various simulated test results of the new scheme (reference Chapter 2) based on the composite signal using natural modes of propagation have been discussed.

### 5.1 NATURAL MODES OF PROPAGATION

This section describes: (1) a theoretical concept of the natural modes of propagation, (2) application of the natural modes of propagation in the Power Line Carrier System, (3) application of the natural modes of propagation for Power System Protection purposes.

### 5.1.1 THEORETICAL CONCERT OF THE NATURAL MODES OF <br> PROPAGATION

It is a well known fact that a multi-conductor line can be represented in terms of the phase variables: voltages and currents as in Equation (3.26) of Chapter 3 using Admittance Matrix Method. Equation (3.26) is being rewritten for a transmission line of length 'l' with sending end and receiving end voltages and currents as below:

$$
\left[\begin{array}{l}
\overline{\mathrm{I}}_{S}  \tag{5.1}\\
\overline{\mathrm{I}}_{R}
\end{array}\right]=\left[\begin{array}{cc}
A_{Y 1} & -B_{y 1} \\
B_{Y 1} & -A_{Y 1}
\end{array}\right] \quad\left[\begin{array}{c}
\bar{v}_{S} \\
\overline{\mathrm{v}}_{R}
\end{array}\right]
$$

$$
\text { where } \begin{aligned}
A_{y i} & =Y_{o} \operatorname{coth}(\Phi l) \\
B_{y i} & =Y_{0} \operatorname{cosech}(\Phi 1) \\
Y_{0} & =[Z]^{-1} Q \Gamma Q^{-1}
\end{aligned}
$$

$\overline{\mathrm{V}}_{\mathrm{S}}, \quad \overline{\mathrm{I}}_{\mathrm{S}}$ are the sending end voltages and currents.
$\overline{\mathrm{V}}_{\mathrm{R}}$ and $\overline{\mathrm{I}}_{\mathrm{R}}$ are the receiving end voltages and currents.
$Y_{o}$ is the surge admittance.
$\Gamma$ is $\Gamma Z Y$ where $Z$ is the impedance matrix and $Y$ is the admittance matrix of the line.

The solution of Equation (5.1) needs evaluation of $\Phi$. In the evaluation of $\Phi$, a numerical difficulty has been
noticed in general. For this purpose, diagonilisation of matrix is suggested so that the function can be obtained directly from the eigen value matrix ' $\Gamma$ '. If $\Gamma^{2}$ is the diagonal eigen value matrix then it follows from
$Z Y=Q \Gamma^{2} Q^{-1}$
that the relevant matrices used in the two port equations for example, can be evaluated by the theory of matrix functions leading to

$$
\begin{equation*}
\Phi=Q \Gamma Q^{-1} \tag{5.3}
\end{equation*}
$$

Q being a matrix of eigenvectors in which each column corresponds to an eigenvalue of $Z Y$. The diagonalisation leads to the concept of natural modes of propagation.

### 5.1.1.1 Modal voltaqes [52]

Consider the semi-infinite line of length $x$, propagation takes place according to the equation
$v=e^{-\Gamma x} \cdot V_{i}$
where $V_{i}$ is an incident voltage vector. Furthermore finding the eigenvalues and eigenvectors of

$$
\begin{equation*}
v=Q e^{-\Gamma x} Q^{-1} v_{i} \tag{5.5}
\end{equation*}
$$

and introducing the variables $v$ and $v_{i}$ such that
$v=Q^{-1} V$
and $v_{i}=Q^{-1} V_{i}$,
it follows that
$v=e^{-\Gamma x} v_{i}$

But $e^{-\Gamma x}$ is diagonal indicating that each element of $v$ can be considered to the mutual exclusion of all other elements. For any such element, it is possible to think in terms of a single phase line characterised by the corresponding element of ' $\Gamma$ ', there being as many such lines as there are distinct values of ' $\Gamma$ '.

A solution in terms of each component voltage is exactly
analogous to the solution of a single line problem. Each component voltage is associated with its own propagation constant $\Gamma_{k}$ where for

$$
\Gamma_{k}=\alpha_{k}+j \beta_{k^{\prime}}
$$

$\alpha_{k}$ and $\beta_{k}$ define attenuation and phase shift constants per unit length respectively. The latter also defines phase velocity such that $u_{k}=w / \beta_{k}$ where $w$ is frequency in radians.

The concept of component voltages leads to the simple interpretation that at any point $x$ the actual voltage vector may be considered as being composed of a number of components, the relative distribution of voltage corresponding to a given component being defined by a particular column of $Q$, known as the modal matrix. These columns in turn define natural modes of propagation. Thus if one of the natural modes is generated at one end of a multiconductor line it will propagate without changing its voltage distribution but it will of course be attenuated and shifted in phase. If the initial voltage vector does not correspond to a particular mode, the vector $V$ at any point $x$ will not have the same relative distribution as $V_{i}$. In general there are as many modes as there are effective conductors. As an example, consider a three phase horizontal line for which the eigen vector or mode distribution matrix $Q$ is given by

$$
\begin{align*}
Q & =\left[\begin{array}{ccc}
1 & 1 & 1 \\
1 & 0 & -2 \\
1 & -1 & 1
\end{array}\right]  \tag{5.9}\\
\text { Then from } Q^{-1} & =\left[\begin{array}{ccc}
1 & 1 & 1 \\
\frac{3}{3} & \frac{-}{3} & 3 \\
\frac{1}{2} & 0 & -\frac{1}{2} \\
1 & \frac{1}{6} & \frac{1}{3} \\
\frac{1}{6}
\end{array}\right]
\end{align*}
$$

(5.10)
it follows that when the centre conductor is energised from a unit voltage source with the remaining conductors short circuited

$$
v_{i}=Q^{-1} V_{i}=Q^{-1}\left[\begin{array}{l}
0  \tag{5.11}\\
1 \\
0
\end{array}\right]=\left[\begin{array}{c}
1 \\
3 \\
0 \\
-\frac{1}{3}
\end{array}\right]
$$

so that it may be seen that the initial voltage vector is composed of $1 / 3$ of mode 1 , zero mode 2 and $-1 / 3$ mode 3 . i.e.

and at any point $x$

$$
\begin{align*}
& V=e^{-\Gamma 1 x}\left[\begin{array}{c}
1 \\
- \\
3 \\
1 \\
- \\
3 \\
1 \\
3
\end{array}\right]-e^{-\Gamma 3 x}\left[\begin{array}{c}
1 \\
- \\
3 \\
2 \\
- \\
3 \\
1 \\
\frac{1}{3}
\end{array}\right]  \tag{5.13}\\
& \text { Mode } 1 \\
& \text { Mode } 3
\end{align*}
$$

where Г1, Г3 are propagation constants for Mode 1, Mode 3. Moreover, from the attenuation constants associated with the three modes, typically $10 \mathrm{~dB} / \mathrm{km}, 0.1 \mathrm{~dB} / \mathrm{km}$ and 0.01 $\mathrm{dB} / \mathrm{km}$ respectively at power line carrier frequencies, it is seen that the attenuation of modes 1 and 3 are in the ratio 1000/1. In consequence, for matched conditions say at a point $\mathrm{x}=87 \mathrm{~km}$ the arrival voltage has the distribution of mode 3 modified in magnitude by

$$
\begin{equation*}
\frac{1}{3} e^{(-0.01 \times(87 / 8.7))}=\frac{1}{3} e^{-0.1} \tag{5.14}
\end{equation*}
$$

(since $1 \mathrm{~Np}=8.7 \mathrm{~dB}$ ) assuming that mode 1 voltage received can be neglected because of its high attenuation. As a result an appreciable part of the arrival power is contained on the two conductors which in the first instance had no signal coupled to them.

### 5.1.1.2 Perfectly Transposed Three Phase Line[52]

For a perfectly transposed, and therefore balanced line, the modes correspond to the positive, negative and zero sequence components. If $a_{s}$ is the operator for the symmetrical components then
$Q=\left[\begin{array}{lll}1 & 1 & 1 \\ 1 & a_{S}{ }^{2} & a_{s} \\ 1 & a_{S} & a_{s}{ }^{2}\end{array}\right]$
and

$$
\begin{equation*}
K=Q^{-1} \tag{5.16}
\end{equation*}
$$

$$
=\frac{1}{3}\left[\begin{array}{ccc}
1 & 1 & 1 \\
1 & a_{s} & a_{s}{ }^{2} \\
1 & a_{s}{ }^{2} & a_{s}
\end{array}\right]
$$

It is known in this case that the propagation constant of the zero sequence mode is greater than that of the balanced sequences. This is due to the presence of the earth, i.e. a perfectly transposed line having a voltage distribution
corresponding to the zero sequence component must be coupled electromagnetically to the earth return path exclusively. On the other hand the balanced or aerial modes are characterised by their independence of earth effects. This results in the zero phase sequence impedance being greater than the positive and negative, these latter two being equal (as are the propagation constants). In practice a line is not transposed perfectly. For a single circuit horizontal line the eigenvector matrix can have the form:

$$
\left.\begin{array}{r} 
\\
\\
\text { Phases }  \tag{5.17}\\
Q \\
=\begin{array}{l}
\text { a } \\
\mathrm{b} \\
\mathrm{c}
\end{array} \\
\end{array}\right]
$$

and in this case the propagation constants (velocity and attenuation , are different as shown in Table 5.1 and to some degree depend upon earth effects. Mode 1 being highly dependent on earth effects, has a zero sequence characteristic and for this reason is often referred as the Earth Mode. Mode 2 even though it appears balanced, depends upon conductor height above ground and the separation between outer conductors but is less affected by the presence of the earth. In the extreme case of an infinite separation between outer conductors, these conductors can be considered as independent systems with separate earth return paths, whereas for a separation very much less than conductor height above ground, little electromagnetic coupling between the conductors and earth
exists. similar reasoning applies to Mode 3 but because of its more balanced nature, its propagation constant is generally less than that of Mode 2. This is particularly so at high frequencies. For Mode 3, smaller separation between adjacent conductors is more important than the separation between the outers.

It may be noted that modes having the lower velocities of propagation are also those with the highest attenuation. This is shown in Table 5.1 which gives typical values of velocity and attenuation for single and double circuit horizontal and vertical lines.

### 5.1.2 APPIICATION OF THE NATURAL MODES OF PROPAGATION IN POWER LINE CARRIER COMMUNICATION

For many years, the application of power-line-carrier was empirical. Lines were relatively short, and attenuation was usually well within the $d B$ rating of the transmitter and receiver. Line attenuation was measured after the equipment was installed, if at all. Ocasionally, inexplicable anomalies were noted. Often utilities discovered that a carrier channel was coupled to two different phases at opposite ends of the line, and that this combination gave the best received signal level. It was also noted that the rate of attenuation (dB/mi) on a short line was apparently higher than on a long line. At that time, these observations were simply accepted and used.

With the introduction of long EHV lines, however, it became necessary to predict carrier performance accurately, even before the line was built. Modal analysis provides such an approach. It is a mathematical tool similar to the symmetrical components technique used for analyzing unbalanced faults on three-phase power system at 50 or 60 Hz. Like its power counterpart, modal analysis is a practical system whose components or 'modes' can be electrically generated and measured separately.

If a given carrier-frequency power were applied to a single conductor line, the power level would decrease exponentially with distance. At a distance of about 10 miles, approximately 1 percent of the input power would be present. Such a system would obviously be impractical for a power-line carrier channel. If the same input power were fed into one phase conductor of a three-phase horizontal line, however, the 1 -percent signal can be obtained from any of the phase conductor at a distance of about 250 mi . Thus, it is apparent that, even though only one conductor is energized, the other two phase wires play an important part in the propagation of carrier energy. Indeed, modal theory is based on the premise that there are as many independent 'modes' of propagation on a multiconductor line as there are conductors involved in the propagation of energy.

There are five characteristics of natural modes:
a. Any set of phase-conductor currents or voltages existing at any point on a lossy, reflection-free three-phase line can be resolved into three sets of natural-mode components.
b. At any point on a line, the mode components must add up to the actual phase quantities. Also, the total power derived from phase currents and voltages must be equal to the sum of mode powers.
c. The ratio of mode voltage to mode current (the 'mode characteristic impedance' ) is constant on each phase conductor.
d. Each mode propagates with a specific attenuation per unit length and a specific velocity of propagation.
e. A set of phase components corresponding to one mode only cannot be resolved into other modes. The modes are independent, and there is no inter-mode coupling on a uniform line.

There are three modes of propagation which are being described as per the convention of the Westinghouse Electric Corporation, USA (Power Carrier Equipment manufacturer). These conventions are also as per IEEE publications. Mode 3 is a high attenuation mode which is propagated on all three phases with ground return. Because of its high attenuation, mode 3 can be neglected beyond about 10 miles from the transmitter.

Mode 2 is a medium-attenuated mode that is propagated on one outside phase and returns on the other outside phase. There is no mode 2 current in the center phase. Mode 2 losses are higher and more frequency-dependent than mode 1
losses.
Mode 1 is the least attenuated of the three modes, and makes carrier channels possible on long EHV lines. The energy is propagated on the two outer phases and returns on the center phase. Mode 1 attenuation is the lowest, and the least frequency-dependent.

In Section 5.1.1, Mode 3 and Mode 1 have been described as Mode 1 and Mode 3 respectively.

### 5.1.2.1 Modal quantities on an EHV Three-phase Line

Modal analysis has been most generally applied to a horizontally spaced, single-circuit, three-phase EHV line with two overhead static wires 'g' (Figure 5.1). Grounded at each tower, the static wires do not generate any transmission modes. The three modes present on such a system are shown in Figure 5.2. Coefficients p and q are the center-phase mode coefficients for modes 1 and 3 , respectively, and depend on the line under study. The value of 'q' will vary from about 1.1 to 1.3 ; p can range from about -1.6 to -1.9. Modal calculations and field tests on a $40-\mathrm{mile}, 500-\mathrm{kV}$ line, provided with reflectionfree termination gave the 'p' and 'q' values and mode impedance values listed in Table 5.2.

Values of attenuation coefficients and relative phase velocity for the three modes are shown in Table 5.3. Data shown are for lines from 345 to 765 kV .

### 5.1.3 APPLICATION OF THE MODAL CONCEPT FOR PROTECTION PURPOSES

Application of the Modal Concept for calculation of superimposed components of the voltages and currents in a transmission line system has been described in Section 5.1.1. Application of the Modal Concept in carrier applications has been described in Section 5.1.2. In this section, application of the Modal Concept has been described for protection purposes. When a disturbance or a fault occurs on a transmission line, voltages and currents (which include superimposed components) in different phases can be found at both ends of a transmission line. These voltages and currents are used for relaying applications. Normally three protection eqipments will be required for a three phase system which is generally an un-economic proposal. In case of radio-interference, signals can be coupled to all the phases, and the protection equipment may mal-operate. With these considerations, it is proposed to combine the three phase output signals in a suitable 'mode', and the advantages can be described as below:

1. economical,
2. provides a composite signal to detect any kind of fault (phase to earth, phase to phase, etc.) using a single protection equipment,
3. cancels the signals due to other sources of noise, e.g. radio-interference.

Since the earth-mode does not cancel signals due to other sources of noise e.g. radio interference, application of
the earth mode has not been considered. Only (1,0,-1) Mode and (1,-2,1) Mode have been considered.

### 5.2 DIGITAL SIMULATION TECHNIOUE OF $(1,0,-1)$ MODE AND (1,-2,1) MODE FOR_PROTECTION APRLICATION

Theoretically, a relationship between the phase voltages and the modal voltages can be expressed as
$v=Q^{-1} V$
where $v$ is the modal voltages in the matrix form (1,3), $V$ is the phase voltages in the matrix form $(1,3)$ and $Q$ is the eigen vector in the matrix form $(3,3)$ for a three- phase single circuit line. This can be expressed as

$$
\left[\begin{array}{r}
\mathrm{v}_{0}  \tag{5.19}\\
\mathrm{v}_{1} \\
\mathrm{v}_{2}
\end{array}\right]=\left[\begin{array}{rrr}
\mathrm{a} & \mathrm{a} & \mathrm{a} \\
\mathrm{a} & 0 & -\mathrm{a} \\
\mathrm{a} & -2 \mathrm{a} & \mathrm{a}
\end{array}\right] \quad\left[\begin{array}{c}
\mathrm{v}_{\mathrm{a}} \\
\mathrm{v}_{\mathrm{b}} \\
\mathrm{v}_{\mathrm{c}}
\end{array}\right]
$$

where $\mathrm{V}_{0}, \mathrm{~V}_{1}, \mathrm{~V}_{2}$ are the Earth Mode, Aerial Mode 1 , and Aerial Mode 2 respectively. 'a' is a coefficient of inverse of the eigen vector matrix $Q . V_{a}, V_{b}, V_{c}$ are phase voltages for the Phase 'a', Phase 'b', Phase 'c' respectively.

## Steady State Modal Voltages

For the purpose of the digital simulation of the pre-fault steady state voltages, Equation (5.19) can be expressed as below at the sending end, say $P$ (reference Chapter 2)

$$
\left[\begin{array}{c}
\operatorname{V0PSC}  \tag{5.20}\\
\operatorname{V1PSC} \\
\operatorname{V2PSC}
\end{array}\right]=\left[\begin{array}{lll}
\operatorname{SI}(1,1) & \operatorname{SI}(1,2) & \operatorname{SI}(1,3) \\
\operatorname{SI}(2,1) & \operatorname{SI}(2,2) & \operatorname{SI}(2,3) \\
\operatorname{SI}(3,1) & \operatorname{SI}(3,2) & \operatorname{SI}(3,3)
\end{array}\right]\left[\begin{array}{l}
\operatorname{VPS}(1,1) \\
\operatorname{VPS}(2,1) \\
\operatorname{VPS}(3,1)
\end{array}\right]
$$

## and at the receiving end Q (reference Chapter 2)

$$
\left[\begin{array}{l}
\operatorname{VOQSC}  \tag{5.21}\\
\operatorname{V1QSC} \\
\operatorname{V2QSC}
\end{array}\right]=\left[\begin{array}{lll}
\operatorname{SI}(1,1) & \operatorname{SI}(1,2) & \operatorname{SI}(1,3) \\
\operatorname{SI}(2,1) & \operatorname{SI}(2,2) & \operatorname{SI}(2,3) \\
\operatorname{SI}(3,1) & \operatorname{SI}(3,2) & \operatorname{SI}(3,3)
\end{array}\right]\left[\begin{array}{l}
\operatorname{VQS}(1,1) \\
\operatorname{VQS}(2,1) \\
\operatorname{VQS}(3,1)
\end{array}\right]
$$

Alternatively, euuations (5.20) and (5.21) can be written as

$$
\begin{align*}
& {[\operatorname{VpPSC}]=[S I(p, q)][\operatorname{VPS}(p, q)]}  \tag{5.22}\\
& {[\operatorname{VpQSC}]=[S I(p, q)][\operatorname{VQS}(p, q)]} \tag{5.23}
\end{align*}
$$

where $S I(p, q)$ denotes $Q^{-1}$ matrix if $p=1,2,3$ and $q=$ 1,2,3. VpPSC and VpQSC denote Sending End 'P'and Receiving End ' $Q$ ' Modal Voltages: Earth Mode (1, 1, 1), Mode $1(1,0,-1)$ and Mode $2(1,-2,1)$ if $p=0,1,2 . \operatorname{VPS}(p, q)$ and $\operatorname{VQS}(p, q)$ are Sending End 'P' and Receiving End 'Q' phase voltages for Phase 'a', Phase 'b' and Phase 'c' if $p=1,2,3$ and $q=$ 1.

From the equations (5.20) and (5.21), Mode (1.0,-1) voltages at Sending End 'P' and Receiving End 'Q' can be written as below:

```
V1PSC = VPS(1,1)*SI(2,1) + VPS(2,1)*SI(2,2) +
    VPS (3,1)*SI (2,3)
```

$$
\begin{align*}
\operatorname{V1QSC}= & \operatorname{VQS}(1,1) \star \operatorname{SI}(2,1)+\operatorname{VQS}(2,1) * \operatorname{SI}(2,2)+ \\
& \operatorname{VQS}(3,1) * \operatorname{SI}(2,3) \tag{5.25}
\end{align*}
$$

From the equations (5.20) and (5.21), Mode (1,-2, 1) voltages at Sending End 'P' and Receiving End 'Q' can be written as below:

$$
\begin{align*}
\operatorname{V2PSC}= & \operatorname{VPS}(1,1) * \operatorname{SI}(3,1)+\operatorname{VPS}(2,1) * \operatorname{SI}(3,2)+ \\
& \operatorname{VPS}(3,1) * \operatorname{SI}(3,3) \tag{5.26}
\end{align*}
$$

```
V2QSC = VQS (1, 1)*SI (3,1) + VQS (2, 1)*SI (3, 2) +
    VQS (3, 1)*SI (3, 3)
```

Pre-fault steady state modal voltages (1,0,-1) and (1,-2,1) can be found corresponding to each sampling point directly in the time domain at both ends of the transmission line using equations (5.24), (5.25), (5.26), (5.27).

## SUPERIMPOSED MODAL VOLTAGES

For the purpose of the digital simulation of the post-fault superimposed voltages, Equation (5.19) can be expressed in the frequency domain as below at the sending end $P$ (reference Chapter 2)
$\left[\begin{array}{l}\operatorname{VOPFC} \\ \operatorname{V1PFC} \\ \operatorname{V2PFC}\end{array}\right]=\left[\begin{array}{lll}\operatorname{SI}(1,1) & \operatorname{SI}(1,2) & \operatorname{SI}(1,3) \\ \operatorname{SI}(2,1) & \operatorname{SI}(2,2) & \operatorname{SI}(2,3) \\ \operatorname{SI}(3,1) & \operatorname{SI}(3,2) & \operatorname{SI}(3,3)\end{array}\right]\left[\begin{array}{l}\operatorname{VPF}(1,1) \\ \operatorname{VPF}(2,1) \\ \operatorname{VPF}(3,1)\end{array}\right]$
and at the receiving end $Q$ (reference Chapter 2)

$$
\left[\begin{array}{l}
\operatorname{VOQFC}  \tag{5.29}\\
\operatorname{V1QFC} \\
\operatorname{V2QFC}
\end{array}\right]=\left[\begin{array}{lll}
\operatorname{SI}(1,1) & \operatorname{SI}(1,2) & \operatorname{SI}(1,3) \\
\operatorname{SI}(2,1) & \operatorname{SI}(2,2) & \operatorname{SI}(2,3) \\
\operatorname{SI}(3,1) & \operatorname{SI}(3,2) & \operatorname{SI}(3,3)
\end{array}\right]\left[\begin{array}{l}
\operatorname{VQF}(1,1) \\
\operatorname{VQF}(2,1) \\
\operatorname{VQF}(3,1)
\end{array}\right]
$$

Alternatively, equations (5.28) and (5.29) can be written as

$$
\begin{align*}
& {[\operatorname{VpPFC}]=[S I(p, q)] \quad[\operatorname{VPF}(p, q)]}  \tag{5.30}\\
& {[\operatorname{VPQFC}]=[S I(p, q)] \quad[\operatorname{VQF}(p, q)]} \tag{5.31}
\end{align*}
$$

where $S I(p, q)$ denotes $Q^{-1}$ matrix if $p=1,2,3$ and $q=$ 1,2,3. VpPFC and VpQFC denote Sending End 'P'and Receiving End 'Q' Modal Voltages: Earth Mode (1, 1, 1), Mode 1 (1,0, 1 ) and Mode $2(1,-2,1)$ if $p=0,1,2$ in the frequency domain. $\operatorname{VPF}(p, q)$ and $\operatorname{VQF}(p, q)$ are Sending End ' $P$ ' and Receiving End 'Q' phase voltages for Phase 'a', Phase 'b' and Phase 'c' if $p=1,2,3$ and $q=1$ in the frequency domain.

From the equations (5.28) and (5.29), Mode (1,0,-1) voltages at Sending End 'P' and Receiving End 'Q' can be written as below:

$$
\begin{align*}
\operatorname{V1PFC}= & \operatorname{VPF}(1,1) * \operatorname{SI}(2,1)+\operatorname{VPF}(2,1) * \operatorname{SI}(2,2)+ \\
& \operatorname{VPF}(3,1) * \operatorname{SI}(2,3) \tag{5.32}
\end{align*}
$$

$$
\begin{align*}
\operatorname{V1QFC}= & \operatorname{VQF}(1,1) \star \operatorname{SI}(2,1)+\operatorname{VQF}(2,1) * \operatorname{SI}(2,2)+ \\
& \operatorname{VQF}(3,1) \star \operatorname{SI}(2,3) \tag{5.33}
\end{align*}
$$

From the equations (5.28) and (5.29), Mode (1, 2, 1) voltages at Sending End 'P' and Receiving End 'Q' can be written as below:

```
V2PFC = VPF(1,1)*SI(3,1) + VPF(2,1)*SI (3,2) +
    VPF (3,1)*SI (3,3)
```

```
V2QFC = VQF (1,1)*SI(3,1) + VQF(2,1)*SI (3,2) +
    VQF (3,1)*SI (3,3)
```

The above Modal voltages in the frequency domain can be written as a function $f(w)$ where 'w' is frequency in radians. Now to calculate these modal voltages in the time domain, the Modified Fourier Transform Technique has been applied as below:
$f(t)=\operatorname{REAL} \frac{e^{\alpha t}}{\pi} \int_{0}^{\Omega} f(w-j \alpha) e^{j w t} \sigma d w$
where $f(t)$ is a function of time representing Modal voltages in the time domain. $\alpha$ is a factor to avoid Gibbs oscillations. This factor is expressed as:
$\sigma=\sin (\pi \omega / \Omega) /(w \pi / \Omega)$

In addition, a constant ' $\alpha$ ' is introduced to ensure numerical stability. This constant is known as the frequency shift constant. $\Omega$ is the truncation frequency. In time domain, equations (5.32), (5.33), (5.34), (5.35) can be written as below:

V1PTC $=\operatorname{REAL} \frac{e^{\alpha t}}{\pi} \int_{0}^{\Omega} f(w-j \alpha) e^{j w t} \sigma d w$
where $f(w-j \alpha)$ is $\operatorname{V1PFC}(w-j \alpha)$

V1QTC $=\operatorname{REAL} \frac{e^{\alpha t}}{\pi} \int_{0}^{\Omega} f(w-j \alpha) e^{j w t} \sigma d w$
where $f(w-j \alpha)$ is V1QFC (w - ja)

V2PTC $=\operatorname{REAL} \frac{e^{\alpha t}}{\pi} \int_{0}^{\Omega} f(w-j \alpha) e^{j w t} \sigma d w$
where $f(w-j \alpha)$ is $\operatorname{V2FFC}(w-j \alpha)$
$V 2 Q T C=\operatorname{REAL} \frac{e^{\alpha t}}{\pi} \int_{0}^{\Omega} f(w-j \alpha) e^{j w t_{\sigma}} d w$
where $f(w-j \alpha)$ is V2QFC (w - j $\alpha$ )

Equations (5.38), (5.39), (5.40), (5.41) give Modal voltages $(1,0,-1)$ and $(1,-2,1)$ at both the ends of the transmission line in the time domain.

### 5.3 SIMULATED TEST RESULTS

The simulated test results are based on a single phase to earth fault at a distance of 1 km from the sending end (SE) of a 100 km long 400 kV 3 -phase transmission line with the end sources of 5 GVA at the $S E$ and 20 GVA at the receiving end (RE). The post-fault observation times are 600 microseconds and 5 milli-seconds for different simulated tests. The corresponding sampling frequencies are 853.33 kHz and 819.2 kHz respectively. The fault has been considered at the peak voltage point-on-wave i.e. 90 degree. The internal fault and the reverse or external fault have been considered with reference to Figure 2.1. The fault generated noise signals have been received through the stack tuner circuit as shown in Figure 2.1. The stack tuner circuit has been tuned to the mid-frequency of $300 \pm$ 2.5 kHz and all the simulated test results are based on this mid-frequency of $300 \pm 2.5 \mathrm{kHz}$. The reference for the direction of the fault i.e. forward or reverse is also shown in Figure 2.1. The forward or the reverse fault can be detected by any high speed directional detector device (DD). The operational aspects of the scheme under internal or reverse fault conditions are explained in Figure 2.5. Figure 'A' shows the operatioal aspects at the $S E$ and the RE under the internal fault condition. Figure 'B' shows the operational aspects at the $S E$ and the $R E$ under the reverse fault condition. The stack tuner circuit is arranged either in the normally open mode 'NO' i.e high impedance mode, or the closed mode 'NC' i.e short circuit
mode through a switch 'S' which is operated by the $D D$ as shown in Figure 3.13. Under the internal fault condition, the $D D$ detects the fault as 'forward' at the $S E$ and the RE i.e. the signal $T_{F}$ as '1'and the signal $T_{R}$ as '0'. For the internal faults, the switch 'S' is in the NO mode and is not affected by the $D D$. The signals $V_{D}$ corresponding to the fault signal at the $S E$ and the RE are also '1'. '0' or '1' is with reference to a decision logic circuit which gives a tripping signal '1' only when the signals $V_{D}$ and $T_{F}$ are '1'. If the fault is at 1 km from the $S E$ of a 100 km long transmission line, the fault response at the SE is instantaneously and at the RE , the fault response is delayed by about 0.4 milli-seconds which is the travelling time of the fault generated travelling wave to reach the $R E$. If the operating time of the $D D$ is considered as 1 milli-second, then $T_{F}$ will be available at the SE after a delay of 1 milli-second and at the RE after a delay of 1.4 milli-seconds. Before processing the signal $V_{D}$ to the decision logic circuit, the signal is processed through a level detector to get the signal '1' or '0'. The level detector detects the level at a particular time, say $T_{T}$ which is the operating time of the scheme without considering the delay due to the travelling time of the fault generated travelling wave. In such a case the signal $T_{F}$ from the $D D$ is delayed further by the time $T_{T}-1$ millisecond. If $T_{T}$ is 2 milli-second, the $T_{F}$ is delayed further by 1 milli-second. In the reverse fault condition the DD detects the fault as 'forward' at the RE and as 'reverse' at the SE. The signal $T_{F}$ is '1'at the RE and '0' at the

SE. The signal $T_{R}$ is ' 0 ' at the RE and '1' at the $S E$. For the reverse faults , the switch 'S' is in the NO mode at the RE and is not affected by the DD. But at the $S E$, the NO mode of the 'S' is changed over to the NC mode through a signal from the DD which operates reverse at the SE. This is shown in Figure 2.5 under reverse fault condition. At the $S E$, the signal $\mathrm{V}_{\mathrm{D}}$ is '1', the signal $\mathrm{T}_{\mathrm{F}}$ is '0', and the signal $T_{R}$ is ' 0 ' till the stack tuner is under the NO mode. Depending upon the operating time of the DD, the stack tuner changes to the $N C$ mode and provides the signal $T_{R}$ as '1'. Any way at the $\mathrm{SE}, \mathrm{V}_{\mathrm{D}}$ is '1' and the $\mathrm{T}_{\mathrm{F}}$ is '0', so that the decision logic output for the tripping is always '0'. At the RE, $V_{D}$ is '1' till the stack tuner at the $S E$ changes over to the NC mode. Once the stack tuner changes to the $N C$ mode at the $S E$, the signal $V_{D}$ at the $R E$ becomes ' 0 ' due to the short circuiting of the signal at the $S E$. The signal $T_{F}$ is '1' at the $R E$. The signal $T_{R}$ is ' 0 ' at the RE. The decision logic output is always '0'at the RE. Therefore the scheme does not operate under the reverse fault condition. The signal processing of the scheme is explained in Figure 5.3. $V_{o a}, V_{o b}, V_{o c}$ are the output of the stack tuners corresponding to the three phases of the transmission line. These signals have been combined in a suitable Aerial mode (1,0,-1) or (1,-2,1) to get a composite signal, $\mathrm{V}_{\mathrm{F}}$. This signal is passed through a suitable filter i.e. auto-correlator or a band pass filter. The simulated test results have been studied for both the filters: (1) Band Pass Filter of a narrow band width, (2) auto-correlator. The otput signal of this filter is
called, $V_{X}$ which passes through an 'AND GATE'. The 'AND GATE' also receives a signal from the DD after an operating time delay, say typically 1 milli-second. The output of this 'AND GATE', $V_{Y}$ does not indicate fault response for this delay of 1 milli-seconds. This delay is increased if the travelling time of the fault generated travelling wave to reach the $S E$ or the $R E$ is taken into account. Now the signal $V_{Y}$ is squared and integrated with the time. This signal is called as $\mathrm{V}_{\mathrm{Z}}$. This signal $\mathrm{V}_{\mathrm{Z}}$ is passed through a level detector whose level is preset. The setting of the level takes into account noise signals due to filter, stack tuner, etc. If the signal $\mathrm{V}_{\mathrm{Z}}$ is more than the preset value of the level detector, the output $V_{D}$ is '1' otherwise the output $V_{D}$ is '0'. The level of the level detector is set for a particular instant, say typically 2 milli-secods which is agreed as the operating time of the scheme. This has been described earlier in this section. The signal $V_{D}$, delayed by about 2 milli-seconds is passed through another 'AND GATE' which also receives the signal $T_{F D}$. This signal $T_{F D}$ is the signal $T_{F}$ which is delayed by the time $T_{D}$, typically, say an 1 milli-second. This provides total 2 milli-seconds delay for the signal from the $D D$ to match the operating time of the level detector as described earlier. As explained in Figure 2.5, the discrimination for the internal fault and the reverse fault at the $S E$ is looked after by the $D D$, but at the $R E$, the discrimination is looked after by the signal of the fault response. So main emphasis of the simulated test results is based on the discrimination of the signals of the fault responses for
the internal and the reverse faults at the RE. This discrimination is also the basis of the preset value of the level detector. However it is to be ensured that the signal at the $S E$ for the internal fault is reasonably much above the preset value of the level detector for the reliable operation of the scheme.

### 5.3.1 SELECTION OF AERIAL MODE AND FAULT RESPONSE $\mathrm{V}_{\mathrm{F}}$

Figure 5.4 shows the fault responses in the Aerial Mode (1,0,-1) for the internal and the reverse faults at both ends: SE and RE. Figure 5.5 shows the fault responses in the Aerial Mode (1,-2,1) for the internal and the reverse faults at both the ends $S E$ and RE. Based on the positive peak value of the responses at the $R E$ for the internal and the reverse fault, the discrimination factor is about 1.25 for the Aerial Mode (1,0,-1) and 2.5 for the Aerial Mode (1,-2,1). The discrimination factor is defined as the peak response for the internal fault divided by the peak response for the reverse fault. Since the discrimination factor for the Aerial Mode (1,-2,1) is larger than the discrimination factor for the Aerial mode (1,0,-1), the Aerial Mode (1,-2,1) has been selected for the simulated test results. Moreover, Mode (1,0,-1) is independent of the middle phase voltage. This means that the Aerial Mode ( $1,0,-1$ ) response depends on the two phases only which reduces the reliability of the scheme. Further Aerial Mode (1,-2,1) is mostly used in carrier communication system as
it is of a least attenuation. These responses are based on the post fault observation time of 600 micro-seconds. With a view to take consideration of the operating time of the DD, it is necessary to study the fault responses for a period of 5 milli-seconds. Further responses for 600 micro-seconds do not indicate the repeatitive fault responses. The time interval between the responses depends upon the travelling time of the fault generated travelling wave between the fault point and the $S E$ or the RE. Figure 5.6 shows the fault responses for a period of 5 milliseconds in the Aerial Mode (1,-2,1) for the internal and the reverse faults at SE and RE . It can be seen that the responses are repeating at a definite interval depending upon the travelling time of the fault generated travelling wave between the fault point and the RE or the SE. These responses are $V_{F}$ with reference to Figure 5.3 of the signal processing. For the purpose of simulating the internal fault, it is to be noted that the stack tuners are in the NO mode at both the ends: $S E$ and $R E$, and the fault is at a distance of 1 km from the SE . For the purpose of simulating the reverse fault, the stack tuners are in the $N C$ mode at the $S E$ and in the NO mode at the $R E$ with the fault at the same point as for the internal fault i.e. 1 km from the $S E$. This shows the worst case of the reverse fault as the fault has been considered in the zone of the internal fault. This point of the fault has been chosen under the reverse fault condition to study the discrimination factor under the worst case of the fault. This point of the fault can also be considered as the
nearest point of the reverse fault for the purpose of studying the discrimination factor.

Theoretically, reflection coefficient $R_{1}$ is given by:
$R_{1}=\frac{z_{L}-z_{O}}{Z_{L}+z_{O}}$
where $Z_{L}$ is the load impedance and $Z_{O}$ is the surge impedance of the transmission line.

For an open circuit case (NO), $Z_{L} \geqslant Z_{O}$, and the $R_{1}=1$ For a short mode (NC), $Z_{L}<Z_{O}$, and the $R_{1}=-1$

If the reverse fault is considered at $F_{2}$ in
Figure 2.1, $R_{l}$ is 1 at the $R E$ and -1 at the $S E$. In such a case, there is an effective component of the superimposed voltage at the RE. But, if the reverse fault is considered at $F_{1}$ in Figure 2.1, $R_{1}$ is -1 at the $S E$ and the travelling wave is reflected back towards the fault. This ensures that there is no refracted travelling wave towards the RE. In such a case, there is a negligible superimposed voltage at the RE.

### 5.3.2 BAND PASS FILTER

The output signals of the stack tuner with mid-band frequency of $300 \pm 2.5 \mathrm{kHz}$ may have non-band frequency components of very large range i.e. 100 to 500 kHz , and it is necessary to filter these non-band frequency components to improve upon the signals. For this purpose, a band pass filter with a mid-band frequency of $300 \pm 5 \mathrm{kHz}$ has been simulated. This filter has the frequency range of only 290
to 310 kHz . This filter filters all frequency components below 290 kHz and above 310 kHz . The frequency response of such a band pass filter is shown in Figure 5.7. The computer program used for the design of the band pass filter has been given in Appendix D. The design details of the band pass filter are also given in Appendix D. For the purpose of running the program, the software package 'SPICE' has been used. Figure 5.8 shows the filtered signals $\mathrm{V}_{\mathrm{X}}$ (reference Figure 5.3) for the internal and the reverse fault conditions at the SE and the RE. Figure 5.9 shows the signal $V_{Y}$ as explained earlier in this section with reference to Figure 5.3. It can be seen that there is zero response for a period of 1 milli-second at the SE for the internal and the reverse fault conditions. This 1 milli-second is considered as a delay due to the operating time of the DD. At RE, the delay is 1.4 milli-seconds for the internal and the reverse faults. This delay includes: (1) 1 milli-second due to the operating time of the DD, (2) 0.4 milli-second due to travelling time of the fault generated travelling wave from the point of the fault to the RE. Figure 5.10 shows the signals $V_{Z}$ for the internal and the reverse fault conditions at the $S E$ and the RE. The signal $\mathrm{V}_{\mathrm{Z}}$ is given by:
$V_{Z}(k)=\sum_{k=1}^{N} V_{Y}{ }^{2}(k \delta t)$
where $V_{Y}$ is the input signal. $k=1,2, \cdots-N$ where $N$ is total number of the samples. $\delta t$ is the sampling time for each sample.

It can be seen in Figure 5.10 that the discrimination factor is about 5 at the instant of 2 milli-seconds. It can also be seen that the signal at the $S E$ at the instant of 2 milli-secons for the internal fault is very much greater than the signal at the $R E$ for the reverse fault condition. This ensures the reliable operation of the scheme under the internal fault condition if the level detector is preset according to the signal at the instant of the 2 milli-seconds at $R E$ for the reverse fault condition.

### 5.3.3 AUTO-CORRELATOR AND ITS SIMULATION TECHNIOUE

The auto-correlation technique gives the filtered response of a signal. It works on the principle that if there is a signal $P_{x}$ with NS number of samples. A fixed window 'WC' of samples $N$ can be selected from this signal for the purpose of the correlation. The method is based on the principle that the moving window 'WS' of $N$ samples of the signal $\mathrm{P}_{\mathrm{x}}$ is selected. The moving window 'WS' starts from the first sample of the signal. The moving window 'WS' is correlated with the fixed window 'WC' and the first sample of the correlated response $P_{x x}$ is calculated. Now the moving window 'WS' moves by the one sample and, the second sample is calculated by correlating the present moving window 'WS' with the fixed window 'WC'. Thus the moving
window 'WS'is moved by one sample each time and is correlated with the fixed window 'WC', and the sample of the correlated response is calculated. Mathmatically, it can be explained as below:

$$
\begin{equation*}
P_{x x}(T)=\frac{1}{N} \sum_{k=1}^{N} P_{x}(k \delta t+m \delta t) \cdot P_{x}(k \delta t) \tag{5.44}
\end{equation*}
$$

```
where T = m\deltat, m=0,1,2\cdotsN
    k = 1,2, - - . . -N
    N
    N = number of samples of the window
    \deltat = sampling time
```

This method needs an advance information of the fixed window 'WC' of $N$ samples of the signal $P_{x}$, and the correlated response is delayed by the fixed window 'WC'. For this purpose, a method of correlation using a delay of only one sample is used instead of the delay of the complete fixed window 'WC'. In this method, the sample of the correlated response is calculated based on the following equation till the complete fixed window 'WC' becomes available:
$P_{x x}(T)=\frac{1}{N} \sum_{k=1}^{N} P_{x}(k \delta t) \cdot P_{x}(k \delta t)$
where $T=k \delta t$
$\mathrm{k}=1,2, \ldots \ldots \mathrm{~N}$

So, first $N$ samples of the correlated response are calculated based on the above equation. Once the fixed window 'WC' of N samples is available, the moving window 'WS' of the signal $P_{x}$ is moved by one sample each time and the correlated response is calculated using the following equation.

$$
\begin{equation*}
P_{x x}(T)=\frac{1}{N} \sum_{k=1}^{N} P_{x}(k \delta t+m \delta t) \cdot P_{x}(k \delta t) \tag{5.46}
\end{equation*}
$$

```
where T = (N+m)\deltat, m=1,2\cdots-N
    k = 1,2, - - - - -N
```


### 5.3.4. DESIGN ASPECTS OF THE HARDWARE OF THE AUTO-CORRELATOR AND THE SELECTION OF THE NUMBER OF SAMPLES OF THE FIXED WINDOW

In the present scheme, the principle is based on the detection of high frequency components of the fault generated noise. These high frequency components are in the range of $300 \pm 2.5 \mathrm{kHz}$. In order to detect to these frequency components, a sampling frequency of aleast 605 kHz is required. In the present simulated test results, this sampling frequency is 819.2 kHz which can be
considered nearly 800 kHz for the simplified calculation. In the auto-correlator technique, if there are N samples in the fixed window 'WC'(reference Section 5.3.3) then $N$ numbers of multiplications and additions are required for calculation of the each sample. For example, if there are 25 samples in the fixed window 'WC', then 25 multiplications and 25 additions are required for calculation of the each sample of the auto-correlated response. If the hardware component is selected such that it is capable of carrying the multiplication and the addtion in one instruction, then 25 instructions are required per sample. Since the sampling frequency is 800 kHz , the frequency for each instuction is 25 times 800 kHz i.e. 20000 kHz . This means that the hardware should be capable of carrying 200,000,000 instuctions per second i.e. 20 MIPS (million of instructions per second). A microprocessor hardware upto 5 MIPS are available in general (For example: TMS300). Probably, the micro-processor hardwares of higher MIPS, say typically 40 MIPS are on the way of development. However transputers are commonly used for combination of the micro-processor hardwares if design criterion demands a higher MIPS value. Based on the criterion that the hardware components of 20 MIPS are available, the number of samples of the window 'WS' or 'WC' (reference Section 5.3.3) are considered 25.

### 5.3.5 SIMULATED TEST RESULTS OF THE AUTO-CORRELATOR

Figure 5.11 shows the auto-correlated output, $\mathrm{V}_{\mathrm{X}}$ (reference Figure 5.3) for the internal and the reverse faults at $\operatorname{SE}$ and RE. Figure 5.12 shows the response $V_{Y}$ (reference Figure 5.3) for the internal and the reverse faults at the SE and the RE. These responses show zero response for a period of 1 milli-second at the SE and 1.4 milli-seconds at the RE. The delay of 1 milli-second at $S E$ corresponds to the operation time of the DD. The delay of 1.4 millisecond at the RE takes into account: (1) 1 milli-second for the operating time of the DD, (2) 0.4 milli-second delay due to the travelling time of the fault generated travelling wave from the fault point to the RE. Figure 5.13 shows the response $V_{Z}$ (reference Figure 5.3). A discrimination factor of about 35 has been found at the instant of 2 milli-second for the internal and the reverse fault at the RE. It can be also seen that the signal at the instant of 2 milli-second at the $S E$ for the internal fault is very much greater than the response at the RE for the reverse fault. This means that if the level detector is preset marginally above the signal at the instant of the 2 milli-second at the RE for the reverse fault, a very reliable operation of the scheme under the internal fault conditions is ensured.

Since the discrimination factor for the auto-correlator is about 35 which is about 7 times greater than the discrimination factor for the band pass filter (reference Section 5.3.2), the auto-correlator has been selected for the purpose of filtering Aerial Mode (1,-2,1) signals.

### 5.3.6 ESTABLISHMENT OF THE SCHEME

Figure 5.14 shows the signals $V_{F^{\prime}}, V_{X^{\prime}} V_{Y}, V_{Z}$ for the internal fault and the reverse fault at the $R E$ with reference to Figure 5.6, Figure 5.11, Figure 5.12, and Figure 5.13. These signals have been explained earlier in this section. The only difference is that Figure 5.14 shows the signal $\mathrm{V}_{\mathrm{F}}$ and $\mathrm{V}_{\mathrm{X}}$ same as in the internal fault condition for a period of 1.4 milli-second. Delay of the 1.4 milli-second takes into account: (1) 0.4 milli-second delay due to the travelling time of the fault generated travelling wave from the fault point to the RE, (2) 1 milli-second delay due to operating time of the DD.

The discrimination factor (value of the signal for the internal fault divided by the value of the signal for the reverse fault at a particular instant) of about 35 has been found which is a very good factor for establishing the principle of the scheme. Figure 5.15 shows the $V_{F}$ (reference Figure 5.6) and $\mathrm{V}_{\mathrm{Z}}$ (reference Figure 5.13) signals for the internal fault at the $S E$ and the RE. It has been found that the values of the signals at the instant, say typically 2 milli-second for both ends: the SE and the $R E$ are much greater than the value of the signal at
the $R E$ for the reverse fault (reference Figure 5.14) which ensures the safe reliable operation of the scheme under the internal fault condition. Figure 5.16 shows the signal $V_{F}$ (reference Figure 5.6 ) and the signal $V_{Z}$ (reference Figure 5.13) for the reverse fault at the $S E$ and the RE. In the reverse fault condition, the operation of the scheme at the SE is inhibited due to the reverse signal from the DD as explained earlier i.e. the signal $T_{F}$ is ' $O^{\prime}$. At the RE, the operation of the scheme is inhibited due to higher threshold value of the level detector than the signal $V_{\bar{Z}}$ i.e. the signal $V_{D}$ is ' ${ }^{\prime}$ ' (reference Figure 5.3). Since both the signals $T_{F}$ and $V_{D}$ are required to be '1' for the operation purpose, the operation under the reverse fault condition can not take place. Hence the principle of the scheme has been established successfully.

### 5.3.7 SIMULATED TEST RESULTS FOR THE FAULT AT 30 DEGREE OF THE VOLTAGE WAVEFORM

When the fault occurs at a lower angle of the voltage waveform, the travelling wave components are much smaller in the magnitude than the fault at 90 degree point-on-wave. Due to this reason, most of the schemes based on the principle of the travelling wave phenomena are considered failures for the faults at lower inception angles. With this view, the simulated test results for the fault at 30 degree of the voltage waveform are shown in figure 5.17. Figure 5.17 shows the signals $V_{F}$, $V_{Z}$ for the internal fault at the $S E$ and the $R E$. It can be seen that the $V_{Z}$ signal at
the RE at the instant of 2 milli-second is about 2 times of the signal at the $R E$ for the reverse fault (reference Figure 5.16) which gives a discrimination factor of 2 for the safe reliable operation of the scheme for faults at 30 degree of the voltage waveform. This can be also seen that the signal $\mathrm{V}_{\mathrm{Z}}$ at the SE at the instance of 2 milli-second is much greater than the signal $V_{Z}$ at the $R E$ for the reverse fault. This ensures the safe and the reliable operation of the scheme for the faults upto 30 degree of the voltage waveform.

### 5.3.8 MODIFIED SIGNAL PROCESSING SCHEME

In the earlier scheme, the directional detector, DD operating delay of typical 1 ms has been considered after auto-correlating the signal $\mathrm{V}_{\mathrm{F}}$. In the present scheme, the signal $V_{F}$ is first delayed and then, the signal $V_{x}(n)$ is auto-correlated as shown in Figure 5.18. If $n$ is the number of samples in the signal $V_{x}$, then the autocorrelated output $\mathrm{V}_{\mathrm{y}}(\mathrm{n} \delta \mathrm{t})$ is given by

$$
\begin{equation*}
V_{y}(n \delta t)=\frac{1}{N} \sum_{k=1}^{N} V_{x}(k+n) \delta t \cdot V_{x}(k \delta t) \tag{5.47}
\end{equation*}
$$

where $\mathrm{n}=1,2 \ldots \mathrm{n}$ $\mathrm{k}=1,2,-\mathrm{N}$
$\mathrm{n}=$ number of samples in the signal $\mathrm{N}=$ number of samples in the window.

This auto-correlated signal $\mathrm{V}_{\mathrm{y}}$ is processed through another function whose output $\mathrm{V}_{z}$ is given by

$$
\begin{equation*}
V_{z}(n \delta t)=\sum_{k=n-M^{y}}^{n} V^{2}(k \delta t) \tag{5.48}
\end{equation*}
$$

where $M=$ number of samples corresponding to the signal delay $T D$ which is typically 3 ms , $\mathrm{n}=$ number of samples corresponding to the signal time, say typically 4 ms .

In fact this signal is the time integral of the square of the signal $V_{y}$ in a fixed window of $M$ samples which is fed to a level detector. If the signal $V_{z}$ is greater than the threshold value (reference Chapter 2) of the level detector, the output of the level detector $V_{D}$ which is '1', is fed to a decision logic, AND GATE. This AND GATE also receives the signal, $T_{F D}$ from the directional detector, $D D$ after the signal delay $T D$. If both the signals $V_{D}$ and $T_{F D}$ are '1', the tripping signal $T_{S}$, 1 is sent to the local breaker for tripping.

In the earlier scheme, there has been no arrangement for resetting of the signal $V_{\bar{Z}}$ and the scheme may mal-operate under the reverse fault condition at the receiving end if the scheme has earlier operated for the internal faults. In this scheme, the signal $\mathrm{V}_{\mathrm{y}}$ is processed through a window of fixed samples and resets as soon as the window is completed.

Figure 5.19 shows the signals $V_{X}, V_{Y}, V_{z}$ for the internal and the reverse faults at the receiving end, $R E$ for the
fault at maximum of the voltage. Looking into Figure 5.19c for the reverse fault, a threshold value can be set corresponding to a value of $20 \times 10^{6}$ units. Now looking into Figure 5.19c for the internal fault, the discrimination factor (reference Section 5.3.1) of $50 \times 10^{7} / 20 \times 10^{6}=25 \quad$ can be noticed. Figure 5.20 shows the signals $V_{F}$ and $V_{z}$ for the internal and the reverse faults at the sending end, $S E$ for the fault at maximum of the voltage. Looking into Figure 5.20b, the discrimination factor for the internal fault is $20 \times 10^{8} / 20 \times 10^{6}=100$, and for the reverse fault is $12 \times 10^{5} / 20 \times 10^{6}=0.006$. This establishes $V_{D}$ signal as $11^{\prime}$ for the internal fault and '0' for the reverse fault. Figure 5.21 shows the logic conditions of the signals $V_{D}$, $T_{F D}$, and the tripping signal $T_{S}$ for the internal fault and the reverse fault at the SE and the RE. Figure 5.21A shows the $V_{D}$ as 1 for the internal fault and 0 for the reverse fault. Figure 5.21B shows the signal $T_{F D}$ as 1 at the $S E$ and the RE for the internal fault and at the RE only for the reverse fault after a delay of 4.4 ms . Figure 5.21C shows the tripping signal $T_{S}$ as '1' for the internal faults and ' 0 ' for the reverse fault. This establishes the principle. Figure 5.22 shows the signals $V_{x}$ and $V_{z}$ for the internal fault at both ends, SE and RE for the fault at 30 degree of the voltage. The discrimination factors have been found to be $13 \times 10^{7} / 20 \times 10^{6}=6.5$ at SE and $30 \times 10^{6} / 20 \times 10^{6}=1.5$ at RE. This ensures the safe operation of the scheme for fault at 30 degree of the voltage.

### 5.3.9 SIMULATED TEST RESULTS FOR FAULTS AT DIFFERENT POINTS OF THE TRANSMISSION IINE

Figure 5.23 shows the fault responses at the SE and the RE for faults at $1 \mathrm{~km}, 25 \mathrm{~km}$, and 50 km from the SE . The repeatition of the responses at a definite interval is clearly shown. This interval depends upon the distance of the fault from the $S E$ or the $R E$. These responses are based on the post fault observation time of 600 micro-seconds. For the purpose of the reliable operation of the scheme, these responses are to be compared with the response at the RE under the reverse fault condition (reference Figure 2.5). This is due to the fact that the level of the level detector (reference Figure 5.3) is set above the value of the signal at $R E$ for the reverse fault. For the purpose of the comparison, the signals have been squared and integrated with the time, and these responses are shown in Figure 5.24. Figure 5.25 shows the response for the reverse fault and Figure 5.26 shows the signal which is squared and integrated with time for the reverse fault at the $S E$ and the $R E$ (reference Figure 2.5). Comparison of these responses based on the discrimination factor at the instant of 0.4 milli-second is shown in Table 5.4. This comparison show the safe and the reliable operation of the scheme for all the points (i.e. $1 \mathrm{~km}, 25 \mathrm{~km}, 50 \mathrm{~km}$ from the SE) of the line.

## EAULT

Figure 5.27 shows the responses for phase-A fault, phase-B fault, phase-C fault, phase to phase fault, three phase fault. For the purpose of the comparison (reference Section 5.3.9), these responses have been squared and integrated with the time as shown in Figure 5.28. These responses have been compared with the response at the $R E$ for the reverse fault condition for the purpose of the safe and the reliable operation (reference Section 5.3.9). The comparison based on the discrimination factor at the instant of the 0.4 milli-second is shown in Table 5.5. Since the discrimination factor is satisfactory in all the cases, the operation of the scheme under all the fault conditions can be ensured safely and reliably.

| 9 |  | $\begin{aligned} & 9 \\ & 0 \\ & 0 \end{aligned}$ |
| :---: | :---: | :---: |
| a | b | c |
| - | - | $\bigcirc$ |

Figure 5.1 SINGLE CIRCUIT, HORIZONTAL THREE-PHASE EHV LINE CONFIGURATION FOR MODAL ANALYSIS FOR THE PURPOSE OF POWER LINE CARRIER APPLICATION

| Phase | Mode 3 | Mode 2 | Mode 1 |
| :---: | :---: | :---: | :---: |
| a | $\xrightarrow{+}$ | $\xrightarrow{1}$ | $\xrightarrow{\longrightarrow}$ |
| b | $\underline{\text { q }}$ |  |  |
| c | $\xrightarrow{\longrightarrow}$ | -1 | $\xrightarrow{\square}$ |

Figure 5.2 MODE DISRTIBUTION FOR MODAL ANALYSIS


Figure 5.3 BLOCK SCHEMATIC OF RELAY SIGNAL PROCESSING

a: Internal fault

b: Reverse fault
X-Axis: TIME-Seconds
Y-Axis: TUNER OUTPU'T RESPONSE-Volts

Figure 5.4 MODE ( $1,0,-1$ ) RESPONSES 'VF FOR INTERNAL AND REVERSE FAULT CONDITIONS

SE
RE


a: Internal fault

b: Reverse fault

X-Axis: TIME-Seconds
Y-Axis: TUNER OUTPUT RESPONSE-Volts




b: Reverse fault

$$
\begin{aligned}
& \text { X-Axis: } \text { TIME-Seconds } \\
& \text { Y-Axis: TUNER OUTPUT } \\
& \text { RESPONSE-Volts }
\end{aligned}
$$

Eigure $5.6 \operatorname{MODE}(1,-2,1)$ RESPONSES ' $V_{E}$ ' FOR FAULT OBSERVATION TIME 5 ms UNDER THE INTERNAL AND THE REVERSE FAULT CONDITIONS


X-Axis: FREQUENCY-Hz
Y-AXIS: GAIN=OUTPUT/INPUT


_b: Reverse fault

$$
\begin{aligned}
\mathrm{X} \text {-Axis: } & \text { TIME-Seconds } \\
\mathrm{Y} \text {-Axis: } & \text { FILTER OUTPUT } \\
& \text { RESPONSE 'VX Volts }
\end{aligned}
$$

[^0]

a: Internal fault
(20,

Figure $5.9 \frac{\text { RESPONSES 'VY' AFTER AN OPERATING DELAY OF THE }}{\text { DIRECTIONAL DETECTOR,DD }}$


> X-Axis: TIME-Seconds
> Y-Axis: RESPONSE 'V ${ }_{Z}$ '-Units

Eigure 5.10 RESPONSES 'VZ'UNDER THE INTERNAL AND THE REVERSE FAULT CONDITIONS

SE


a: Internal fault


b: Reverse fault

$$
\begin{aligned}
& \text { X-Axis: } \text { TIME-Seconds } \\
& \text { Y-Axis: AUTO-CORRELATED } \\
& \text { RESPONSE 'VX'-Units }
\end{aligned}
$$

Eigure 5.11 AUTO-CORRELATFI) RESPONSES 'VX' UNDER THE INTERNAL AND THE REVERSE EAULT CONDITIONS



b: Reverse fault

$$
\begin{aligned}
& X-A x i s: ~ T I M E-S E C O n d s \\
& Y-A x i s: ~ A U T O-C O R R E L A T E D ~ \\
& \text { RESPONSE 'V } V_{X}-l \text { Inits }
\end{aligned}
$$

Eigure $5.12 \frac{\text { RESPONSES ' } V_{Y} \text { ' AF'TER AN OPERATING DELAY OF THE }}{\text { DIRECTIONAL DETECTOR,DD }}$

SE



b: Reverse fault

$$
\begin{aligned}
& \text { X-Axis: 'rIME-Seconds } \\
& \text { Y-Axis: RESPONSE } ' \mathrm{~V}_{Z} \text { '-Units }
\end{aligned}
$$

Figure 5.13 RESPONSES ' $V_{Z}$ ' UNDER THE INTERNAL AND THE



Figure 5.14 SIGNAL RESPONSES $V_{F}, V_{X}, V_{Y}, V_{Z}$ AT RE FOR THE INTERNAL AND THE REVERSE FAULT

b: Signal ' $V_{Z}$ '.

> X-Axis: TIME-Seconds
> Y-Axis: RESPONSE $V_{F}, V_{Z}-$ Units

[^1]
\[

$$
\begin{aligned}
& \mathrm{X} \text {-Axis: TIME-Seconds } \\
& Y \text { Axis: RESPONSE } \mathrm{V}_{\mathrm{F}}, \mathrm{~V}_{\mathrm{Z}} \text {-Units }
\end{aligned}
$$
\]







$$
\begin{aligned}
& \text { X-Axis: TIME-Seconds } \\
& \text { Y-Axis: RESPONSE } V_{F}, V_{Z} \text {-Units }
\end{aligned}
$$



Figure 5.18 MODIFIED SIGNAL PROCESSING SCHEME

a: Signal ' $V_{F}$ '


b: Siqnal 'Vy'.



Internal fault $c: ~ S i g n a l ~ ' ~ V_{Z}$ ' Reverse fault
X -Axis: TIME-Seconds
Y-Axis: RESPONSE $\mathrm{V}_{\mathrm{F}}, \mathrm{V}_{\mathrm{Y}}, \mathrm{V}_{\mathrm{Z}}$-Units
Figure 5.19 RESPONSE $V_{F}, V_{X}, V_{Z}$ FOR THE INTERNAL FAULT AND THE REVERSE FAOLT AT VMAX AI' RE


Figure 5.20 RESPONSE $V_{F}, V_{Z}$ FOR THE INTERNAL FAULT AND THE REVERSE FAOLT AT VMAX AT SE

A. Signal $V_{D}, 0$ or 1 (1 if $V_{Z} \geq V_{T}, 0$ if $V_{Z}<V_{T} . V_{T}=$ THRESHHOLD VALUE, $20 \times 10^{6}$ units

B. Signal $\mathrm{T}_{\mathrm{FD}}, 0$ or 1 ( 1 if forward fault, 0 if reverse fault)

C. Signal trip TS, 0 or $1\left(1 \mathrm{if} \mathrm{V}_{\mathrm{D}}, \mathrm{T}_{\mathrm{FI}}=1\right)$

Figure 5.21 SIGNALS $V_{D}, T_{F I}$ AND TRIP TS AT SE AND RE



b: Siqnal ' $\underline{V}_{\underline{Z}}$ '.

$$
\begin{aligned}
& \text { X-Axis: TIME-Seconds } \\
& \text { Y-Axis: RESPONSE } \mathrm{V}_{\mathrm{F}}, \mathrm{~V}_{\mathrm{Z}} \text {-Units }
\end{aligned}
$$



Figure 5.23 MODAL 2 RESPONSES AT DIEFERENT POINTS ON THE TRANSMISSION LINE



b:

$\times 10^{5}$


c: 50 km from SE

$$
\begin{aligned}
& \text { X-Axis: TIME-Seconds } \\
& \text { Y-Axis: RESPONSE-Units }
\end{aligned}
$$

Figure 5.24 SQUARE AND INTEGRAL WITH TIME OF RESPONSES CORRESPONDING TO FIGURE 5.23


$$
\begin{aligned}
& \text { X-Axis: TIME-Seconds } \\
& \text { Y-Axis: TUNER OUTPUT RESPONSE-Volts }
\end{aligned}
$$

Figure $5.25 \frac{\text { RESPONSE 'VE' IN REVERSE FAULT CONDITION }}{\text { AT VMAX }}$



X-Axis: TIME-Seconds
Y-Axis: RESPONSE-Units

Figure 5.26 SQUARE AND INTEGRAL WITH TIME OF RESPONSES CORRESPONDING TO FIGURE 5.25


cont.


> X-Axis: TIME-Seconds
> Y-Axis: TUNER OUTPUT voLTAGE-Volts






c: Phase C fault

$$
\begin{aligned}
& \text { X-Axis: TIME-Seconds } \\
& \text { Y-Axis: RESPONSE-Units }
\end{aligned}
$$

Eigure 5.28 SQUARE AND INTEGRAL WITH TIME OF RESPONSES CORRESPONDING TO FIGURE 5.27

Cont.

a: 3 Phase fault


b: Phase-Phase fault

$$
\begin{aligned}
& X-A x i s: ~ T I M E-S e c o n d s \\
& Y-A x i s: ~ R E S P O N S E-U n i t s
\end{aligned}
$$

Figure 5.28 SQUARE AND INTEGRAL WITH TIME OF RESPONSES CORRESPONDING TO FIGURE 5.27

## Table 5.1 MODAL VELOCITIES AND ATTENUATION FOR

 OVERHEAD LINES ( $\mathrm{E}=100 \mathrm{kHz}$ )| Single Circuit | Horizontal line |  | Vertical line |  |
| :---: | :---: | :---: | :---: | :---: |
| Mode | $\begin{aligned} & \text { Velocity } \\ & \mathrm{km} / \mathrm{Sec} \end{aligned}$ | Attenuation $\mathrm{dB} / \mathrm{km}$ | Velocity Attenuation km/Sec <br> dB/km |  |
| 1 | 283395 | 0.55537 | 280294 | 0.64561 |
| 2 | 297392 | 0.5959 | 299380 | 0.02852 |
| 3 | 299402 | 0.03013 | 299529 | 0.02811 |
| Double <br> Circuit |  |  |  |  |
| 1 | 276455 | 0.9575 | 272438 | 1.1842 |
| 2 | 295595 | 0.0858 | 298956 | 0.0351 |
| 3 | 299487 | 0.0158 | 299635 | 0.0099 |
| 4 | 288064 | 0.3372 | 294579 | 0.1209 |
| 5 | 298602 | 0.0248 | 299623 | 0.0132 |
| 6 | 299706 | 0.0142 | 299850 | 0.0094 |

Table 5.2 RESULTS OF MODAL ANALYSIS

| Phase | Mode 3 | Mode 2 | Mode 1 |
| :--- | :--- | :--- | :--- |
| a | 1.0 | 1.0 | 1.0 |
| b | $1.206(\mathrm{q})$ | 0 | $-1.66(\mathrm{p})$ |
| c | 1.0 | -1.0 | 1.0 |
| Modal <br> Impedance <br> (Ohms) | 379.0 | 274.0 | 232.0 |

## Table 5.3 MODE ATTENUATION AND PHASE VELOCITY

| Mode | Attenuation $\mathrm{dB} / \mathrm{mi}$ | Phase Velocity <br> relative to Mode 1 |
| :---: | :---: | :---: |
|  | $30 \mathrm{kHz} \quad 300 \mathrm{kHz}$ |  |
| 1 | 0.01-0.03 0.07-0.009 | 1.0 |
| 2 | 0.09-0.1 0.4-0.5 | 0.98-0.995 |
| 3 | 1.5 to 3.0 at 100 kHz |  |

Table 5.4 COMPARISON OF RESPONSES 'a' AND DISCRIMINATION FACTOR 'd' FOR EAULTS AT DIFFERENT POINTS OF THE LINE AT THE INSTANT OF 0.4 ms (REF. Fiqure 5.24) WITH REFERENCE TO THE RESPONSE AT THE RE FOR THE REVERSE FAULT (REF. Fiqure 5.28)

| Fault at | SE |  | RE |  |
| :---: | :---: | :---: | :---: | :---: |
| Eault at | Response 'a' | Discrimination factor, 'd'=a/b | $\begin{aligned} & \text { Response } \\ & \text { 'a' } \end{aligned}$ | Discrimination factor,'d'=a/b |
| 1 km | $20 \times 10^{5}$ | 5 | $25 \times 10^{5}$ | 6.25 |
| 25 km | $30 \times 10^{5}$ | 7.5 | $20 \times 10^{5}$ | 5 |
| 50 km | $25 \times 10^{5}$ | 6.25 | $24 \times 10^{5}$ | 6 |

Table 5.5 COMPARISON OF RESPONSES ' $a^{\prime}$ AND DISCRIMINATION FACTOR 'd' FOR DIFFERENT TYPES OF FAULT AT THE INSTANT OF 0.4 ms (REF. Fiqure 5.26 ) WITH REFERENCE TO THE RESPONSE AT THE RE FOR THE REVERSE FAULT (REF. Fiqure 5.28)

| Fault Type | SE |  | RE |  |
| :---: | :---: | :---: | :---: | :---: |
|  | Response 'a' | Discrimination factor,'d'=a/b | $\begin{gathered} \text { Response } \\ \text { 'a' } \end{gathered}$ | $\begin{aligned} & \text { Discrimination } \\ & \text { factor,'d'=a/b } \end{aligned}$ |
| Phase'A' | $20 \times 10^{5}$ | 5 | $25 \times 10^{5}$ | 6.25 |
| Phase'B' | $10 \times 10^{6}$ | 25 | $11 \times 10^{6}$ | 27.5 |
| Phase'C' | $10 \times 10^{6}$ | 25 | $8.5 \times 10^{6}$ | 21.25 |
| Three- | $32 \times 10^{6}$ | 80 | $10 \times 10^{6}$ | 25 |
| Phase <br> Phase- <br> Phase | $8 \times 10^{6}$ | 20 | $9.5 \times 10^{5}$ | 2.4 |

## CHAPTER 6

## INVESTIGATION OF THE PRINCIPLE CONSIDERING BUS BAR CAPACITANCE

This chapter investigates the principle as described in Chapter 2 for the transmission line system with effective bus bar capacitance [47]. This bus bar capacitance may vary from line to line depending upon various factors in particular the number of lines terminating at the bus bar. A typical value of $0.1 u F$ has been considered in this chapter for investigation purposes. As explained in Chapter 2, the stack tuner has to be in the normally close mode, NC for the internal faults (reference Figure 2.2) and changeover to the open mode, NO for the external faults as shown in Figure 2.3. The operational principle has been explained in Chapter 2 with reference to Figure 2.6. The digital simulation technique has been explained in Chapter 3. After digital simulation of the stack tuner and the bus bar capacitance, fault responses at the $S E$ and the RE have been studied at the bus bars (primary side of the stack tuner) and have been found to be satisfactorily as described in Chapter 3. In this chapter, fault responses will be investigated at the output (secondary side) of the
stack tuner to establish the principle for faults at the voltage maximum. Further investigations have been carried out for the satisfactory operation of the scheme 1) at lower fault inception angles of the voltage and 2) under the condition of different source capacities. For the simulated test purposes, phase A to earth fault has been considered at 1 km from the SE of a 100 km long 400 kV transmission line with sources of 5 GVA at SE and 20 GVA at RE. The fault observation time is 600 micro-seconds and the sampling frequency is 853.33 kHz . Value of the frequency shift constant $\alpha$ has been adjusted very critically at 1000 or 10000. It has been found that higher the value of $\alpha$, the response has been found to be satisfactory nearly at the end of the fault response period. At the beginning of the fault response period, the response has been found to be diverge very soon. If the lower value of the $\alpha$ is chosen, the response has been found to be satisfactory at the beginning of the fault response period. At the end of the fault response period, the response has been found to be diverge. In the present study, a lower value of $\alpha=$ 1000 has been chosen for the simulated testing the discrimination feature and a higher value $\alpha=10000$ has been chosen for other tests. to get satisfactory responses at the end of the fault response period at the RE. The reason for selecting a satisfactory response at $R E$ is that the discrimination feature of the scheme is decided by the responses at the $R E$ for the internal fault and the reverse fault. The fault responses have been considered for a period of only 600 micro-seconds to establish the basic
principle of the scheme. As explained above, with the higher value of $\alpha=10000$, at the $S E$, the divergence of the responses has been found and therefore the responses have been recorded only for a period of half the fault response period i.e. 300 micro-seconds for other tests: different fault inception angles and different source capacities. It has been established earlier that the transients fault responses (bursts) repeats with attenuation (refernce Figure 4.3) at every interval (depending upon the distance of the fault from the bus bar) for a period of about 5 msec. In such a case if the discrimnation for the reverse fault can be established based upon the simulated fault responses for a period of 600 micro-seconds without considering the effect of the operating delay of the directional detector, $D D$, then the principle based upon the signal processing (reference Chapter 4 and Chapter 5) can be established without any difficulty. The signal processing includes auto-correlation, time delay due to operation of the directional detector, DD, squaring and time integral of the delayed signal for input to the level detector. Based upon the threshold value of the level detector, the output signal $\mathrm{V}_{\mathrm{D}}$ (reference Chapter 2) of the level detector is 1 or 0 . This signal $V_{D}$ goes to a decision logic which also gets a signal 1 or 0 from the directional detector for the reverse or the forward fault. When both signals are 1, then only a tripping signal is sent for operation of the local breaker. This signal processing scheme is same as has been explained in the earlier chapters. So in this chapter, the signal
processing of the output of the stack tuner has not been included.

### 6.1 Establishment of the principle

Figure 6.1 shows the fault responses for the internal fault and the reverse fault at the $S E$ and the RE. Figure 6.1a shows the responses for the internal faults and Figure 6.1b shows the responses for the reverse faults. It has been explained earlier that the stack tuners are in the normally close mode at $S E$ and $R E$ for the internal faults and for the reverse fault at F1 (reference Figure 2.1), the stack tuner changes to the open mode. With reference to figures 6.1, the operation can be explained based on the peak voltages of the faulted phase response. Peak voltages have been found to be 6 kV at the SE and 3 kV at the RE for the internal fault. When considering an external fault, the peak voltages have been found to be 350 Volts at the $S E$ and only 1.5 kV at the RE. This low response at the RE gives an excellent feature of the discrimination of the scheme. With reference to Chapter 2 if the threshold value can be set to a value proportion to say 2 kV , the $\mathrm{V}_{\mathrm{D}}$ can be found to be 1 for the internal faults and 0 for the reverse fault at both ends the $S E$ and the $R E$. For the reverse fault at the $R E, V_{D}$ is 0 . The signal from the directional detector $T_{F}$ is 1 in all cases except in the case of the reverse fault at the $\mathrm{SE} . \mathrm{T}_{\mathrm{F}}$ is 0 at the SE for the reverse fault. Since the tripping signal is initiated only when both signals $\mathrm{V}_{\mathrm{D}}$ and $\mathrm{T}_{\mathrm{F}}$ are 1 , it can be confirmed
that the scheme operates satisfactorily for the internal fault and does not operate for the reverse fault. This establishes the principle of the scheme.

### 6.2 Effect of different fault inception angles

Figure 6.2 shows the fault responses for various fault inception angle of the voltage i.e. $0,2,5,15,30,60,90$ degrees at the sending end, $S E$ and the receiving end, RE. It has been established in Chapter 4 that the responses are proportional to the value of $\sin \Theta_{f s}$ for a particular voltage of the transmission system and the particular midfrequency of the stack tuner. where $\Theta_{\text {fs }}$ is the fault inception angle. If $\overline{\mathrm{V}}_{\mathrm{ff}}(\mathrm{SE})$ is the peak voltage at the SE in Volts and $\bar{v}_{f f}(R E)$ is the peak voltage at the $R E$ in Volts, the following table shows (1) $\Theta_{f s}$, (2) the positive peak responses at the $S E$, (3) its percentage for the fault at the maximum voltage, (4) the positive peak responses at the RE, (5) its percentage for the fault at the maximum voltage, (6) value of $\sin \Theta_{f s}$, (7) its percentage for the fault at the maximum voltage.

## TABLE

| $\Theta_{\mathrm{fS}}$ | $\overline{\mathrm{V}}_{\mathrm{ff}}(\mathrm{SE})$ | $\%$ | $\overline{\mathrm{~V}}_{\mathrm{ff}}(\mathrm{RE})$ | $\%$ | $\sin \Theta_{\mathrm{fS}}$ | $\%$ |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 90 | 25000 | 100 | 25000 | 100 | 1 | 100 |
| 60 | 21000 | 84 | 21000 | 84 | 0.866 | 86.6 |
| 30 | 12500 | 50 | 12500 | 50 | 0.5 | 50 |
| 15 | 6500 | 26 | 6500 | 26 | 0.258 | 25.8 |
| 5 | 2100 | 8.4 | 2100 | 8.4 | 0.087 | 8.7 |
| 2 | 900 | 3.6 | 900 | 3.6 | 0.035 | 3.5 |

This table shows a good agreement between the calculated values and the simulated test results. Comparing the positive peak response at the $R E$ for the reverse fault at maximum voltage (Figure 6.1b) and the internal fault at 2 degree point-on-wave fault of the voltage (Figure 6.2f), it can be concluded that the scheme can be used satisfactorily for all faults for the point-on-wave of 2 degrees or more. In general 30 degrres is the very well accepted value as has been evaluated in Chapter 4.

### 6.3 Effect of different source capacities

Simulated tests have been carried out for (1) 1 GVA at both ends, (2) 50 GVA at both ends, (3) 99 GVA at both ends. Figure 6.3 shows the results for these simulated tests. It has been found that there is no appreciable difference in responses and it can be concluded that the performance of the developed scheme is satisfactory for different source capacities for the transmission line with two sources.


```
VD=1,TF=1
TR=0,TRIP=1
```

$\mathrm{VD}=1, \mathrm{TE}=1$
$T R=0, \operatorname{TRIE}=1$


b: Reverse fault
VD=0,TF=0
VD=0,TF=0
TR=1,TRIP=0
TR=1,TRIP=0
$\mathrm{VD}=0, \mathrm{TE}=1$
$T R=0, T R I P=0$
X axis: TIME - Seconds
Y axis: VOLI'AGES - Volts
(TUNER OUTPUT)

Eigure 6.1 ESTABLISHMENT OF THE PRINCTPLE CONSIDERING BUS BAR CAPACITANCE




e: 5 Deqrees

$$
\begin{gathered}
X \text { axis: TIME - Seconds } \\
\text { Y axis: VOLTAGES - Volts } \\
\text { (TUNER OUTPUT) }
\end{gathered}
$$

Figure 6.2 EFFECT OF DIFFERENT FAULT INCEPTION ANGLES CONSIDERING BUS-BAR CAPACITANCE

Cont.

f: 2 Degrees

R E



$$
\begin{gathered}
\text { X axis: TIME - Seconds } \\
\text { Y axis: VOLTAGES - Volts } \\
\text { (TUNER OUTPUT) }
\end{gathered}
$$

Figure $6.2 \quad$ EFFECT OF DIFFERENT FAULT INCEPTION


## c: 99 GVA at both ends

(TUNER OUTPUT) $Y$ axis: VIME - seconds
Figure 6.3
EFFECT OF DIFFERENT SOURCE CAPACITIES CONSIDERING BUS-BAR CAPACITANCE

## CHAPTER 7

## DIGITAL SIMULATION OF ELECTRIC ARC MODEL AND SUPERIMPOSED ARC VOLTAGES

As explained earlier in Chapter 1, most faults on HV transmission lines involve arcing or some other intermittent discharge. Arc length may be small or large but voltage drop across it, is generally very small relation to transmission system voltages. Typically, for a 20 meter arc length in a 400 kV system, a maximum voltage drop of 20 kV is expected and does not affect a protection scheme which normally works on the basis of steady state fault current. It is only the current which is limited due to arc resistance and can affect the operation of a relay. In the new approach of UHS protection due to fault generated noise, the principle of protection is based on the detection of a particular frequency band from the high frequency voltage waveform. In such a scheme, noise due to other effects in particular arc can not be neglected. With this aim the present chapter describes the digital simulation techniques of the arc model and the transmission system with the arc model.

### 7.1 Digital simulation of electric arc model [42.43]

The static arc characteristic is very well known. In the present scope of work, the dynamic arc characteristic has been considered due to reversal of current at the power cycle 50 Hz or 60 Hz . In the dynamic characteristic, the arc does not follow the same path for increasing the current and decreasing the current, and this effect is known as the arc hystersis. Considering Strom's [35] dynamic characteristic, an approximate linearised arc model has been considered for simplification as shown in Figure 7.1. $I_{p}$ is peak of the steady state arc current in amperes. $I_{1}$ and $I_{h}$ are $0.15 I_{P}$ and $0.38 I_{P}$ respectively. $V_{p}$ is the maximum arc voltage per cm of the arc length which is 10.3 Volts.

The arc cyclogram shown in Figure 7.1 can be obtained repeatedly only if the time for which the current remains zero, is so small that the arc can be reestablished without any appreciable deionization or restriking voltage. Infact the cyclogram shown in Figure 7.1 has negligible reignition voltage. Since the arcing faults may influence the fault generated high frequency waveforms, it is desired to include the effect of reignition voltage to have a reasonably good accuracy in the arc cyclogram modelling. According to Fukunishi [41], the rate at which the reignition voltage increases is a function of the time for which the arc exits. With this view Johns [42] defined the reignition voltage as
$\left|V_{R}(t)\right|=K\left(T_{e}\right)\left(t-T_{e}\right) h\left(t-T_{e}\right) k V / C m$
where $T_{e}$ is arc extinguishing time and $K\left(T_{e}\right)$ is the rate of increase of the reignition voltage. Value of $K\left(T_{e}\right)$ has been found to be as $5 \mathrm{~V} / \mathrm{ms}$ and $15 \mathrm{~V} / \mathrm{ms}$ for $\mathrm{T}_{\mathrm{e}}$ of 0 and 200 ms respectively [53] and the above relationship of Johns [41] can be modified as

$$
\begin{equation*}
\left|V_{R}(t)\right|=\left(5+50 T_{e}\right)\left(t-T_{e}\right) h\left(t-T_{e}\right) k V / C m \tag{7.2}
\end{equation*}
$$

Figure 7.2 shows the effect of reignition voltage $V_{R}(t)$ on the arc cyclogram $0, a, b, c, b, 0, a^{\prime}, b^{\prime}, c^{\prime}, b^{\prime}, 0, ~ a$ piecewise linear characteristic VA (voltage - ampere). Following the arc voltage reversal, the arc current is held zero till the arc voltage remains below the reignition voltage given by above equation. When the arc voltage exceeds the re-ignition voltage, the arc voltage follows the arc cylogram. The broken line ${ }^{\prime} V_{R}(t)$ - $a$ ' shows the lower value of the reignition voltage before following the arc cyclogram. The continuous line ${ }^{\prime} V_{R}(t)$ - a' shows the higher value of the reignition voltage before following the arc cyclogram. This arc voltage $\bar{v}_{\text {arc }}$ can be evaluated using Thevenin equivalent circuit as shown in Figure 7.3.

$$
\begin{equation*}
\overline{\mathrm{V}}_{\mathrm{arc}}=\dot{\mathrm{E}}_{\mathrm{T}}-\dot{\mathrm{I}}_{\mathrm{arc}} \cdot \mathrm{z}_{\mathrm{T}} \tag{7.3}
\end{equation*}
$$

where $\bar{E}_{T}$ is Thevenin source voltage at the fault point, $\bar{I}_{\text {arc }}$ is the arc current and, $Z_{T}$ is Thevenin source impedance. In the present work it has been asumed that (1) the time for which the arc current is being held zero, has been assumed to be very small, (2) the reignition voltage shoots up to a value greater than the arc voltage and follows the normal arc cyclogram, (3) the rate of rise of the reignition voltage is proportional to the time from the instant of the fault. Since the arc exits nearly half a cycle all times, $\mathrm{T}_{\mathrm{e}}=0.010$ seconds and the above relationship for the reignition voltage can be simplified as:

$$
\begin{equation*}
\left|V_{R}(t)\right|=5.50 \mathrm{th}(\mathrm{t}) \mathrm{kV} / \mathrm{Cm} \tag{7.4}
\end{equation*}
$$

Now the simulation technique is followed as below: The simulation techinque is being explained for a positive half cycle $0, a, b, c, b, o$ and it shall follow exactly the same for $a$ negative half cycle $0, a^{\prime}, b^{\prime}, c^{\prime}, b^{\prime}, \quad 0$. Simulation process begins with the comparision of the arc voltage against the reignition voltage for 1 cm length of the arc. Thus the current is held at zero until the arc voltage exceeds the re-ignition voltage. Mathematically,

$$
\begin{equation*}
I_{\text {Farc }}(t)=0 \quad V_{\text {farc }}(t)<V_{R}(t) \tag{7.5}
\end{equation*}
$$

and when the arc voltage reaches the value $\mathrm{V}_{\mathrm{R}}(\mathrm{t})$, Equation (7.6) is applied till the value is reached $V_{p}$.

$$
\begin{equation*}
V_{\text {farc }}(t)=\left[\left(V_{p}-V_{R}(t)\right) I_{\text {Farc }}(t) / 0.15 I_{p}+V_{R}(t)\right] \tag{7.6}
\end{equation*}
$$

Now for any current in excess of $0.15 I_{p}$, the arc voltage assumes a constant value $\mathrm{V}_{\mathrm{p}}$ and the arc voltage is given by

$$
\begin{equation*}
V_{\text {farc }}(t)=V_{p}, I_{\text {Farc }}(t)>0.15 I_{p} \tag{7.7}
\end{equation*}
$$

Most of the fault current exceeds $0.38 I_{p}$. Subsequent reduction in arc current follows the position of arc extinction given by equation (7.8)

$$
\begin{equation*}
V_{\text {farc }}(t)=\left[V_{p} I_{\text {Farc }}(t) / 0.38 I_{p}\right], I_{\text {Farc }}(t)<0.38 I_{p} \tag{7.8}
\end{equation*}
$$

It is important to note that current may reduce before reaching $0.38 \mathrm{I}_{\mathrm{p}}$ for arc extinction and in such case arc voltage and current would be constrained to reach extinction along the line for increasing current where $V_{R}(t)=0$. This path can be defined as

$$
\begin{equation*}
\mathrm{V}_{\text {farc }}(\mathrm{t})=\left[\mathrm{V}_{\mathrm{p}} \mathrm{I}_{\text {Farc }}(\mathrm{t}) / 0.15 \mathrm{I}_{\mathrm{p}}\right], 0<\mathrm{I}_{\text {Farc }}(\mathrm{t})<0.15 \mathrm{I}_{\mathrm{p}} \tag{7.9}
\end{equation*}
$$

The above procedure is repeated again and agian after each extinction. The piecewise linear characteristic is cyclically traversed for both positive and negative half cycles until final extinction occurs when $\left|V_{f a r c}(t)\right|$ remains permanently below the arc reignition voltage $\left|V_{R}(t)\right|$. Appendix $E$ shows the simulated test results for (1) the linearised arc cyclogram, and (2) the linearised arc cyclogram with the reignition effect.

### 7.2 Digital simulation of superimposed arc voltages of the transmission line

As explained in Chapter 3.2, the fault current $I_{F}(t)$ at a fault point $F$ as shown in Figure 7.4, can be evaluated. Assuming the fault current as the arc current, it can be written as

$$
\begin{equation*}
I_{\text {Farc }}(t)=I_{F}(t) \tag{7.10}
\end{equation*}
$$

now using the arc cyclogram simulation, the total arc voltage $V_{\text {farc }}(t)$ can be found out and it can be expressed as

$$
\begin{equation*}
V_{\text {farc }}(t)=K_{\text {arc }}(t) I_{F}(t) \tag{7.11}
\end{equation*}
$$

where $K_{\text {arc }}(t)$ is an arc cyclogram constant. To get more realistic value of the arc voltage, a more realistic value of the arc current is necessary and therefore the arc current has been evaluated considering 0.5 Ohm arc resistance.

It is also known that the arc voltage at the fault point is equal to the sum of steady state voltage $\mathrm{V}_{\text {ISS }}(\mathrm{t})$ and superimposed arc voltage $\mathrm{V}_{\text {ffarc }}(\mathrm{t})$ which can be expressed as

$$
\begin{equation*}
V_{f a r c}(t)=V_{f S S}(t)+V_{f f a r c}(t) \tag{7.12}
\end{equation*}
$$

Now superimposed arc voltage can be evaluated using following equation:
$v_{\text {ffarc }}(t)=v_{\text {farc }}(t)-v_{f s s}(t)$

Using the FFT subroutine, the superimposed arc voltage at the fault point can be evaluated in the frequency domain. As explained in Chapter 3, the superimposed arc voltages for all three phases at the sending end $P$ and the receiving end $Q$ can be evaluated. Now using again the FFT subroutine, these superimposed arc voltages at two ends can be evaluated in time domain.


ARC CURRENT

Figure 7.1 APPROXIMATE LINEARISED ARC-CYCLOGRAM


ARC CURRENT

Figure 7.2 APPROXIMATE LINEARISED ARC-CYCLOGRAM WITH REIGNITION EFFECT


Eigure 7.3 THEVENIN EQUIVALENT FAULT POINT CIRCUIT


Eigure 7.4 REPRESENTATION OF THE FAULTED LINE APPLYING SUPERIMPOSED ARC VOLTAGE AT THE FAULT POINT

## CHAPTER 8

## INVESTIGATION OF THE SCHEME BASED ON THE SUPERIMPOSED <br> ARC VOLTAGES

Since most faults involve electric arc, the proposed scheme has to be investigated for the transients due to electric arc. With this view, this chapter investigates the discrimination feature of the scheme with and without effective bus bar capacitance. Other simulated test results include: (1) the performance at zero point-on-wave faults, (2) superimposed arc voltages for different arc lengths, (3) investigation of the superimposed arc voltages (burst) repeating every half cycle.

For simulation of the superimposed arc voltages, a 400 kV 100 km long transmission line with sources 5 GVA at SE and 20 GVA at RE has been considered. Arc has been considered across a 400 kV insulator which is usually 2 meter long. Therefore the arc length of 2 meter has been accounted in the present work. All the simulated results are based on the superimposed voltages of the faulted phase A. Fault has been considered at 1 km from the SE at maximum of the voltage. The stack tuner has been tuned to the midfrequency of $300 \pm 2.5 \mathrm{kHz}$. All fault responses have been
studied at the output of the stack tuner. It is worth to mention that the simulation of the fault responses for a period of 5 ms has been found to be possible at the sampling frequency of 820 kHz to detect 300 kHz components with the available computer Gould 9095. In order to investigate the fault response at an interval of half cycle, the computer program has to be optimised for the fault observation time of 13 ms with sampling points of 8192. In order to confirm the repeat feature of the response at every half cycle, it has been found that the fault observation time of about 2 cycles will be desirable. With this objective, the simulated tests have been carried out for the stack tuner mid-frequencies (1) $50 \pm 40 \mathrm{kHz}$ to detect 50 kHz components for a period of fault observation time of 40 ms with 4096 sampling points, (2) $100 \pm 2.5 \mathrm{kHz}$ to detect 100 kHz components for a period of fault observation time of 40 ms with 8192 sampling points, and, the result has been found to be satisfactory.

### 8.1 Discrimination feature of the scheme with and without effective bus bar capacitance

Figure 8.1 shows the discrimination feature of the scheme without considering the bus bar capacitance. Figure 8.1.a shows the superimposed arc voltages for the internal fault with stack tuners in open mode $N O$ at both ends, $S E$ and $R E$ (reference Chapter 2). Figure 8.1.b shows the superimposed arc voltages for the reverse fault with stack tuners in the close mode at the $S E$ and in the open mode at
the RE. After studying the responses, it can be concluded that the positive peak at the $R E$ for the reverse fault is 1) about half the positive peak response at the RE for the internal fault, 2) at least eight times smaller than the responses at the SE for the internal fault and the reverse fault. Considering $V_{F}, T_{F}, T_{R}$, and TRIP with reference to the Chapter 2, it has been found that the TRIP $=0$ for the reverse fault and $\operatorname{TRIP}=1$ for the internal fault at both ends. This has been shown in Figure 8.1. When these responses are compared with the responses of Figure 4.2 (responses without arcing faults), it has been observed that these responses are about twice in magnitude of the responses of Figure 4.2.

Figure 8.2 shows the transient responses considering the effective bus bar capacitance (typically 0.1uF). The stack tuners are in normally close mode for the internal fault. For the reverse fault, the stack tuner at the SE is in the open mode and the stack tuner at the RE is in the close mode (reference Chapter 2). It can be observed that the TRIP $=0$ for the reverse fault and TRIP=1 for the internal fault at both ends. Figure 6.1 shows the transient responses without considering the arcing faults. In both cases, the frequency shift constant $\alpha$ (reference Chapter 3 ) is 1000 and the responses can be compared to study the effect of the arcing faults. Comparing the responses of Figure 8.2 and Figure 6.1, it can be concluded that the responses with the arcing faults are about twice of the responses without the arcing faults with the discrimination feature as explained for Figure 8.1. This establishes the
discrimination feature of the scheme for the arcing faults. The signal processing of the scheme has not been included in this chapter as it has been explained earlier.

### 8.2 Performance at zero polnt-on-wave faults

It is a well known fact that the faults at zero point-onwave responds superimposed voltages of nearly zero value due to negligible travelling wave effect and therefore the discrimination for the reverse fault can not be guaranteed below a minimum angle of the fault inception. Under the scope of the present work, as most faults involve the electric arc, it is possible to get superimposed voltages of significant value due to the arc effect. With this view, simulated tests have been carried out (1) without arcing fault (2) with arcing fault as shown in Figure 8.3. Now comparing these results, it can be concluded that the response is improved by ten times with the arcing fault and it is possible to improve the discrimination for the reverse faults. Another significant feature can be seen in Figure 8.4 which shows the repeatitive arc effect (burst) after half a cycle. This effect has given a revolutionary idea in the field of power system protection for the travelling wave relays for the successful discrimination even at zero degree fault inception angle. So far relays based on travelling wave phenomena have been considered a failure.

### 8.3 Superimposed arc voltages for different arc lengths

Figure 8.5 shows the responses for different arc lengths:(1) 2 meters, (2) 20 meters, (3) 40 meters and significant improvement in the responses can be noticed for long arcs. The fault observation time is only 600 microseconds to study the effect in the responses due to the different arc lengths.

### 8.4 Repeatitive arc effect every half a cycle

Figure 8.6 shows the repeatitive feature of the arcing fault response (bursts) at every half cycle. The fault has been considered at the maximum of the voltage. Figure 8.6a shows the responses for the stack tuner mid-frequency of $300 \pm 2.5 \mathrm{kHz}$ with the maximum limit of the fault observation time of 13 ms . A burst at an interval of 10 ms i.e. half a cycle can be observed clearly. Figure 8.6b shows the fault responses for the stack tuner mid-frequency of $100 \pm 2.5 \mathrm{kHz}$ with the maximum limit of the fault observation time of 40 ms i.e. 2 cycles and the sampling points 8192.

Figure 8.7 shows the fault responses for the stack tuner mid-frequency of $300 \pm 2.5 \mathrm{kHz}$ with the maximum limit of the fault observation time of 13 ms but with the arc length of 80 meter. A very clear burst at an interval of 10 ms i.e. half cycle can be observed. Figure 8.6a shows the burst at an interval of half cycle for an arc length of 2 meter only. Comparing the magnitude of the bursts it can
be observed that the positive peak of the burst for 80 meter long arc is 1200 Volts which is about 40 times the positive peak of the burst with 2 meter long arc. It can be concluded that the responses are proportional to the arc length. Figure 8.8 shows the repeating effect of the arcing faults (bursts) at every half cycle. Figure 8.8a shows the responses without the arcing faults and Figure 8.8b shows the responses with the arcing faults. These responses are for the stack tuner mid-frequency of $50 \pm 40$ kHz for a fault observation period of 2 cycles with the sampling points of 4096. The comparison of Figure 8.8a and Figure 8.8b clearly shows the bursts for the arcing faults repeating every half cycles.

SE


RE

a: Internal fault
VF=1,TE=1
VF=1,TE=1
TR=0, TRIP=1
TR=0, TRIP=1

$$
\begin{array}{r}
\mathrm{VE}=1, \mathrm{TE}=1 \\
\mathrm{TR}=0, \mathrm{TRIP}=1
\end{array}
$$



b: Reverse fault
VF=1,TF=0
VF=1,TF=0
VF=1,TF=0
TR = 1, TRIP = 0
TR = 1, TRIP = 0
TR = 1, TRIP = 0
$\mathrm{VF}=0, \mathrm{TF}=1$
$\mathrm{VF}=0, \mathrm{TF}=1$
$\mathrm{VF}=0, \mathrm{TF}=1$
$T R=0$, TRIP $=0$
$T R=0$, TRIP $=0$
$T R=0$, TRIP $=0$
X axis: TIME - Seconds
X axis: TIME - Seconds
X axis: TIME - Seconds
Y axis: VOLTAGES - Volts
Y axis: VOLTAGES - Volts
Y axis: VOLTAGES - Volts
(TONER OUTPUT)
(TONER OUTPUT)
(TONER OUTPUT)
Eigure 8.1 DISCRIMINATION TEST FOR REVERSE FAULT

a: Internal fault

```
VD=1,TE=1
TR=0, TRIP= = 1
```

$$
\begin{array}{r}
\mathrm{VD}=1, \mathrm{TF}=1 \\
\mathrm{TR}=0, \mathrm{TRIP}=1
\end{array}
$$



b: Reverse fault

```
VD=0, TF = 0
TR=1, TRIP=0
```

$\begin{aligned} & V D=0, T F \\ & T R=0\end{aligned}$
$T R=0, \mathrm{TRIP}=0$

$$
\begin{gathered}
\mathrm{X} \text { axis: TIME - Seconds } \\
\text { Y axis: VOLTAGES - Volts } \\
\text { (TUNER OUTPUT) }
\end{gathered}
$$

Figure 8.2 DISCRIMINATION TEST FOR REVERSE FAULT CONSIDERING BUS BAR CAPACITANCE

SE
I?



b: With arc effect

$$
\begin{gathered}
X \text { axis: TIME - Seconds } \\
\text { Y axis: VOLTAGES - Volts } \\
\text { (TUNER OUTPUT) }
\end{gathered}
$$

## Eigure 8.3 COMPARISON OE KESPONSES FOR ARC EFFECT AT ZERO FAUL'T INCEPTION ANGLE

$\underline{S E}$


X axis: TIME - Seconds
Y axis: VOLTAGES - Volts
(TUNER OUTPUT)

Figure 8.4 ARC EFFECT REPEATING EVERY HALF CYCLE AT ZERO EAULT INCEPTION ANGLE





$$
\mathrm{b}: 100 \pm 2.5 \mathrm{kHz}
$$

$$
\begin{aligned}
& \mathrm{Y} \text { axis: TIME - Seconds } \\
& \mathrm{Y} \text { axis: VOLTAGES - Volts }
\end{aligned}
$$

(TUNER OUTPUT)

Figure 8.6 ARC EFEECT REPEATING EVERY HALF CYCLE AT 90 DEGREES FAULT INCEPTION ANGLE

SE


R E

X axis: TIME - Seconds
Y axis: VOLTAGES - Volts
(TUNER OUTPUT)

Figure 8.7 ARCING FAULT RESPONSE AT HALF CYCLE FOR VOLTAGE MAXIMUM FAULT WITH 80 METER LONG ARC


> X axis: TIME - Seconds
> Y axis: VOLTAGES - Volts

Figure 8.8 COMPARISON OF ARCING FAULT RESPONSES AT TUNER MID-FREQUENCY $50 \pm 40 \mathrm{kHz}$

## CHAPTER 9

## CONCLUSION

This thesis concludes with a successful approach to a new technique of UHS protection of EHV transmission lines based on detection of high frequency signals of the fault generated noise. The scheme is of non unit type having features of unit type of protection i.e. independent of the communication link. For detecting the reverse faults, a high speed directional detector (dd) has been used which inhibits the operation of the breaker. At the same time, this dd makes a change in the stack tuner such that the response at other end is always below a threshold value which inhibits the operation of the breaker at the other end also. For internal faults, response is always higher than the threshold value and goes to a logic circuit linked with a forward direction of $a$ dd which gives a tripping signal for operation of the breakers at both the ends. Theoretical, operational and design aspects have been established successfully for different applications: (1) transmission lines with negligible bus bar capacitance, (2) transmission line with considerable bus bar capacitance, typically 0.1 microfarads. Since most faults involve electric arc or some other discharge, effect of the
electric arc in both the cases have been considered and the proposed scheme has been established for both the applications based on faulted phase voltage responses. The threshold value has been recommended on the basis of correlation output function as it gives better discrimination for internal and reverse faults. With a view of cancelling out the affect of other noises, three phase outputs have been combined in the modal form 2 (1, -2, 1) and the principle has been established successfully. Responses have been found of considerable amount for various other faults: (1) at different points on the 100 km long transmission line i.e. $1 \mathrm{~km}, 25 \mathrm{~km}$ and 50 km, (2) faults in different phases, (3) three phase faults, (4) two phase fault.

Effect of different source capacities has been also studied and it has been established that the scheme is practically independent of source capacities.

The effect of different fault inception angles has been also studied and satisfactory responses have been noticed for fault inception angles of 15 degrees or more. As it is known that most faults take place near 90 degrees of fault inception angle, the response for faults below 15 degrees of fault inception angles is not relevent practically.

Response at different stack tuner frequencies tuned in narrow band at $300 \mathrm{kHz}, 200 \mathrm{kHz}, 100 \mathrm{kHz}$, and wide band in $100-300 \mathrm{kHz}, 10-90 \mathrm{kHz}$ shows the responses at 200 kHz about twice of that of $300 \mathrm{kHz}, 100 \mathrm{kHz}$ about twice of that of 200 kHz and considerable responses for 300 - 100 kHz and 10 - 90 kHz . This explains that the stack tuner can be
tuned to any frequency which may not interfere with other carrier frequencies, if used. For the present work, 300 kHz has been used as stack tuner frequency due to consideration of size of components.

The system has been digitally simulated to establish various aspects of the scheme. In order to study the effect of electric arc, a linearised arc cyclogram with reignition effect has been considered. Reignition voltage has been calculated approximately to check its effect. As the UHS protection schemes based on travelling wave phenomena are considered faliures at zero degree fault inception angle, responses due to electric arc have been studied and two important observations have been found: (1) response is about ten times more than the case of without electric arc, (2) considerable amount of response has been noticed repeating every half a cycle. With this observation, a new concept has been revolutionized for the success of travelling wave based UHS protection schemes at zero degree fault inception angle.

Tests have been also carried out for different arc lengths and higher responses have been noticed for longer arc lengths.

With a view to study the effect of electric arc repeating every half a cycle at 90 degree fault inception angle, various tests have been carried out and significant responses repeating every half a cycle have been noticed. This observation has given birth to a new concept of UHS protection of EHV transmission lines based on arcing faults.

## CHAPTER 10

## SUGGESSIONS FOR FUTURE WORK

Based on the experience gained in the present thesis, the following works are suggested which may be carried out in the future.
[1] a stack tuner may be designed and fabricated for a transmission line system with a view of field trial for different applications: (1) with negligible bus bar capacitance, (2) with considerable bus bar capacitance.
[2] In the present work, the reignition voltage for considering the electric arc effect has been evaluated using a very simplified expression to enable to study its effect. Future work may be carried out using a more accurate expression for the evaluation of the reignition voltage.
[3] The effect of the frequency shift constant in the Modified Fourier Transform (reference Chapter 3) for evaluating the time domain components from the time varying components has been noticed considerably for the case of the bus bar capacitance. This makes
difficult to evaluate the discrimination factor (reference Chapter 4) for the purpose of the discrimination of the forward and the reverse fault. Probably some more work can be done to identify the appropriate value of the frequency shift constant to get more realistic values of the responses for the purpose of the establishment of the scheme.
[4] A field trial of the fabricated model (suggested for future work) is suggested to confirm an improvement in the operation at lower fault inceptional angles of the voltage waveform due to the concept of the travelling waves due to the arcing faults.
[5] A relaying scheme based on the arcing faults is suggested to be designed and fabricated. A field trial of the scheme is also suggested to confirm the operational principle based on the arcing faults.

## CHAPTER 11

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## CHAPTER 12

## APPENDIX A

## POWER IINE CARRIER

In Chapter 3, a principle of Power Line Carrier Communication and the design criteria have been explained. In this appendix, Power Line Carrier components and Power Line Characteristics have been described.

## POWER LINE CARRIER COMPONENTS

The major carrier components are :

1. Line Trap
2. Coupling Capacitor
3. Drain Coil
4. Line Tuner
5. Coaxial Cable

These components are shown in Figure A-1

## Iine Trap

The line trap consists of a parallel tuned resonent circuit which presents a high impedance to the operating carrier frequency but a negligible impedance at system frequency. The line trap serves two functions:

1. Prevents carrier energy from flowing into the station bus.
2. Prevents an external ground fault (F1) behind the protective relays from short circuiting the carrier signal on the unfaulted line.

Three types of line traps are available: single frequency, double frequency, and broad band. Typical carrier frequency impedance characteristics of the three types of traps are shown in Figure A-2.

## 1. Single-frequency traps

The MS single-frequency line trap (Figure A-2a) covers the 30-to-300 kHz range in four divisions: 30-90, 50-150, 70200, and 90-300 kHz. This trap has a minimum impedance of 400 ohms over a bandwidth of $\pm 5 \%$ of the centre frequency $( \pm 2 \%$ for the $30-$ to -90 kHz range). Both low- and high-Q tuning are available; the high-Q trap is used for a singlefrequency relaying channel. The low-Q trap, which has a lower but relatively constant impedance over a moderate frequency band, is used for a group of closely spaced frequencies.

## 2. Double-frequency Traps

The MD double-frequency line trap (Figure A-2b) covers the 30-to-300 kHz range in the same four ranges as the singlefrequency trap. It may be tuned to any two frequencies between 30 and 300 kHz , but both the upper and lower frequencies must be in the same tuning range. The impedance will be greater than 400 ohms over a bandwidth of $\pm 5 \%$ of the centre frequency $( \pm 2 \%$ for the $30-$ to- 90 kHz range), provided the upper and lower frequencies are separated by 25 kHz or 25 percent of the upper frequency, whichever is greater. Both low- and high-Q tuning are available and are applied as for single-frequency traps.

## 3. Broad-band Traps

The broad-band trap presents a moderate but fairly constant impedance over a wide frequency range and is used to isolate a group of widely spaced carrier frequencies from a station bus. There are three varities of broad-band traps: fixed, field adjustable, and factory adjusted. These cover a wide range of applications:

1. Fixed 90-200 kHz Type MW Trap for applications in the 50-205 and 100-300 kHz range where a 400-ohm trap impedance (0. 265 mH ) is adequate. The characteristics are shown in Figure A-2c. This trap is not adjustable.
2. Field-adiustable Type MWA Traps- for applications in the 50-205 and 100-300 kHz frequency bands with impedance levels of 600,750 or 1000 ohms at moderate bandwidths. Typical characteristics are shown in Figure A-2d and A-2e. This trap has a 0.265 mH main coil.
3. Factory adtusted Type MWA Traps- for applications involving wider frequency bands and higher trap impedance levels than are available with the field adjustable traps. typical characteristics of two of these traps are shown in Figures A-2f through A-2i. These curve show that, for a given geometrical mean frequency (GMF), the bandwidth at a desired impedance level increases with the increase in main coil inductance. The inductance range is from 0.265 to 1.590 mH .

A typical example of the tuning of a Type MWA trap is shown in Figure A-2f, where a GMF of 154 kHz is chosen for a 0.265 mH line trap. A minimum impedance ( $\mathrm{R}_{\mathrm{O}}$ ) of 600 ohms gives an upper frequency $\left(F_{U}\right)$ of 198 kHz and a lower frequency ( $F_{L}$ ) of 118 kHz . The corresponding curve of impedance versus frequency (Figure A-2g) shows both the impedance and resistive components, $Z$ and $R$, respectively. The bandwidth, $F_{U}-F_{L}$ or $198-118 \mathrm{kHz}$ is 80 kHz .

## Coupling Capacitor

The carrier signals are coupled to the power line by the coupling capacitor, $C_{C}$. This unit, a series of stacked capacitors mounted inside procelain insulators, has a
capacitance of the order of 0.002 to 0.02 uF , depending on system ac voltage and potential burden requirements. Standard coupling capacitor ratings are listed in Table A1.

## Drain Coil

The drain coil, mounted in the base of the coupling capacitor, provides a low-impedance path for the flow of power-frequency current through the capacitor to ground, thus minimizing the system-frequency voltage developed from point X to ground (Figure A-1). At carrier frequencies, on the other hand, the drain coil has a high impedance, minimizing r-f losses to ground at this point. The carrier loss introduced by the drain coil does not exceed 0.5 dB .

## Line Tuner

Applied in conjunction with the coupling capacitors, the line tuners (Figure A-1) provide an impedance match with low losses between the coaxial cable and the transmission line. Both resonent and broad-band line tuners are used. Two types of resonant tuners are available: single frequency and double frequency (Figure A-3). In the single-frequency tuner (Figure A-3a), the inductive reactance of $L 1$ cancels the capacitive reactance of the coupling capacitor $C_{C}$ and isolating capacitor $C_{S}$ at one frequency, thus providing a low-loss coupling circuit for the carrier-frequency energy. The transformer (T1) matches
the impedance of the coaxial cable to the surge impedance $\left(z_{o}\right)$ of the power line.

The two-frequency resonant tuner (Figure A-3b) provides low-loss coupling at two separate frequencies from a single coaxial cable to the power line. The two-frequency tuner shown in Figure $A-3 c$ is used for isolating two different sets of carrier equipment. The path from each coaxial cable includes a matching transformer, ( $T 1$ or $T 2$ ) tuning inductance, and a trap circuit tuned to the other carrier frequency. In both cases, the isolating capacitor, $C_{S}$, prevents the system-frequency current ( 50 or 60 Hz ) from the coupling capacitor from flowing through the tuning coil and matching transformer. Such a current flow might mask the relatively small carrier-frequency current.

For the two-frequency tuners, the minimum frequency separation is 25 percent of the lower frequency. Frequency response curves of the single- and double-frequency line tuners are shown in Figure A-4.

Where more than two frequencies are coupled to a power line, a wide band coupling device known as a Hi-Coupler is used in place of a resonant tuner. The Hi-Coupler is a high-pass filter (Figure A-5) in which the coupling capacitor ( $C_{C}$ ) servers as part of the filter. The filter is terminated by the line impedance $Z_{0}$. Radio-frequency hybrids or series L-C circuit, which are part of the associated transmitter output filters (Figure A-5a), are used to isolate channel equipment connected to the coaxial cable. The frequency range of the Hi-Coupler is 40 to 300 kHz . The capacitance of the coupling capacitor should be
0.002 uF or higher. Figure $\mathrm{A}-5 \mathrm{C}$ shows the Hi-Coupler insertion loss characteristics.

## Coaxial Cable

Low-loss concentric cable is used to connect the line tuner and the carrier transmitter-receiver assemblies. Type RG8/U coaxial, recommended for this purpose, has a characteristic impedance of 52 ohms and an attenuation of about 0.4 to $0.9 \mathrm{~dB} / 1000 \mathrm{ft}$, increasing gradually over the 30-to-300 kHz frequency range (Figure A-6). The coaxial cable is generally grounded at the terminal equipment end only to prevent system-frequency 50 or 60 Hz current from flowing over the coaxial cable shield during a ground fault. If both ends of the coaxial cable shield must be grounded to improve the shielding effect, a 4/0 copper cable should be run in the same duct to interconnect the ground points at the two ends of the coaxial cable. This arrangement will eliminate the possibility of current burning the coaxial cable shield open in the event of a ground fault.

## POWER-IINE CHARACTERISTICS

Carrier frequencies exceed power frequencies by a factor of 1000 or more. as a result, a transmission line's response to carrier frequencies will be appreciably different from its response to power frequencies. At 50 or 60 Hz , all power lines are electrically short. At carrier
frequencies, however, most lines are several wavelengths long because of the much shorter wavelength of the higher frequencies.

The relative efficiency of power- and carrier-frequency transmission also differs significantly. Although the conductor size is relatively large on transmission lines, the carrier-frequency energy flows mainly on the surface of the conductors. This skin effect increases with higher frequencies. Other factors influencing the carrier losses include shunt leakage losses, conductor spacing, radiation, and ground resistance.

In carrier application work, the transmission characteristic of a system (or channel) can be conveniently considered in terms of attenuation-the decrease in power level along a transmission line. The ratio between the voltages, currents, or power at any two points is a measure of the attenuation between these two points. Losses expressed in these ratios, however, cannot be added to give the total loss but must be multiplied. It is more convenient to express losses in decibels (dB), which can be added directly. Decibels are defined as follows:

$$
\mathrm{dB}=10 \log _{10} \frac{\mathrm{P}_{1}}{\mathrm{P}_{2}}
$$

$$
\begin{array}{r}
\mathrm{dB}=20 \log _{10} \frac{\mathrm{~V}_{1}}{\mathrm{~V}_{2}}  \tag{A2}\\
\mathrm{~dB}=20 \log _{10} \begin{array}{l}
\mathrm{I}_{1} \\
\mathrm{I}_{2}
\end{array}
\end{array}
$$

Equations (A2) and (A3) are valid only if the impedance levels of the circuits are identical where the two voltages or currents are measured.

Various power and voltage or current ratios, with the corresponding $d B$ values, are listed in Table A-2.

Another term widely used in the field of communications is dBm , a level of dB relative to 1 milliwatt. Thus, a power level of one milliwatt is zero dBm. Other power levels in watts and dBm are:

## Watts dBm

$10.0+40$
$1.0+30$
$0.1+20$
$0.01+10$
$0.001 \quad 0$
$0.0001 \quad-10$
0.00001 -20

Most electronic voltmeters are calibrated in dB relative to a zero-voltage reference level that would produce one milliwatt of power with a 600 ohm load. Most carrier signal levels, however, are measured across a coaxial cable of 50-to-70 ohm impedance level. Assuming a 60-ohm coaxial cable impedance, 10 dB must be added to the scale dB value to give a true dBm reading. Assuming a 50-ohm coaxial cable impedance, $10.8 \mathrm{~dB}(11 \mathrm{~dB}$ is usually close enough) would have to be added. Similarly, for a 70-ohm impedance, 9.34 dB would have to be added.

For example $10-\mathrm{W}(+40-\mathrm{dBm})$ transmitter will develop 24.5 V rms across 60 ohms. A voltmeter calibrated in $d B$ for $a$ 600-ohm impedance will read +30 dB . To obtain a true dBm, 10 dB must be added to the scale reading giving +40 dBm . To avoid confusion, actual scale readings should be recorded, and the values converted to true dBm later.

## Line Attenuation

As indicated above, line attenuation depends on a number of factors. Higher voltage lines usually have a lower loss, since the longer insulator strings reduce carrier leakage and dielectric losses in the insulation. The use of bundled conductors on EHV and UHV lines also reduces the series resistance loss.

Typical fair-weather losses for transmission lines from 34.5 kV to 765 kV are shown in Figure $\mathrm{A}-7$. As indicated in Table A-3a, foul-weather losses are estimated by adding 25 percent to the values shown for lines 230 kV or higher, and

50 percent for lines less than 230 kV . The corrections for transpositions are shown in Figure A-3b. The coupling corrections are shown in Table A-4.

## Characteristic Impedance

Another important line characteristic for carrier is characteristic impedance, or input impedance of an infinitely long line for a given design. If a line of finite length is terminated in its characteristic impedance, there will be no energy reflected from the termination when an ac signal is applied to the line. That is, the characteristic impedance, designated $Z_{o}$ is the ratio of the applied voltage to the resulting current flowing into the line. Carrier terminals and line coupling equipment must be matched to $\mathrm{Z}_{\mathrm{o}}$ for best power transfer. Table A-5 indicates the range of characteristic impedance values for a variety of lines.

## Power-line Noise

The main factor limiting the allowable attenuation on a power-line carrier channel is the noise level of the line itself. Receiver filters will accept a fraction of the total noise voltage, yielding a certain signal-to-noise ratio (rms-signal-to-average noise). The channel becomes unsatisfactory when this ratio drops below a designated value, about 20 dB for on-off carrier relaying and as low as 10 dB for frequency-shift carrier.

Most power-line consists of a mixture of impulse and random noise, in which noise predominates and random noise occupies the space between the pulses. (Impulse noise consists of sharp, well separated pulses caused by specific electrical discharges, random noise has a continuous spectrum.) The average value of line noise is the measure of its interfering properties with a function such as protective relaying.

Figure $A-8$ shows typical fair and foul-weather average noise levels at 230 kV in a $3-\mathrm{kHz}$ bandwidth. The dBm figures are based on a reference level (zero dBm) of one milliwatt. For convenience, a millivolt scale is also included. These curves, based on field data and calculated values from several sources, show that noise decreases with increasing frequency. This phenomena partially compensates for the increase in line attenuation at the higher frequencies.

To determine line-noise levels at other system voltages, the appropriate correction factor (Table A-6) must be added to the value read from the curve in Figure A-8. In this way, the line noise in a $3-\mathrm{kHz}$ band at any carrier frequency and line voltage can be estimated.

The average noise-power accepted by a carrier receiver is proportional to its bandwidth. for a receiver in the standard type TC carrier set, for example, the noise response will be 3 dB less than the values given in Figure A-8, since the $T C$ receiver has a $1500-\mathrm{Hz}$ bandwidth. The correction factor for receivers of different bandwidths are given in Table A-7

For example, the expected foul-weather noise level at 100 kHz on a $500-\mathrm{kV}$ line is calculated as follows for a type TCF wide-band receiver.

From Figure A-8:
Average noise level at $100 \mathrm{kHz} \quad=-17 \mathrm{dBm}$
From Table A-6
Correction factor for $500-\mathrm{kV}$ line voltage = +5 dB
From Table A-7
Correction factor for receiver bandwidth $=-8 \mathrm{~dB}$
Therefore,
Total noise level accepted by receiver $=-20 \mathrm{dBm}$
The attenuation and noise level values given here, although typical of field values, are intended only as a guide. Specific line values may show some deviation for any number of reasons: the age of the line, type of line construction, conductor size, transpositions, insulator leagage, ground wire, ground conductivity, weather conditions, and the test setup. The accuracy of the attenuation and noise level figures is sufficient to determine the type and rating of carrier terminal equipment required for a given application.


Figure A-1 BASIC DIAGRAM OF ONE LINE SECTION SHOWING MAIN CARRIER COMPONENTS FOR PROTECTIVE RELAYING CHANNEL

'a' Typical impedance values 100 kHz Single Frequency Type MS Traps

'b' Typical impedance values 112 and 150 kHz Double Frequency Type MD Traps

'c' Typical impedance values $|z|$ and resistive component $R$ for Fixed Type MW Trap ( 0.265 mH )


Center Frequency, $\mathrm{F}_{\mathrm{O}}(\mathrm{kHz})$
'd' Bandwidth characteristics of the 0.265 mH field Adjustable Wide Band Line Trap 50-205 kHz


Geometric Mean Frequency,kHz
'e' Bandwidth characteristics of the 0.265 mH field Adjustable Wide Band Line Trap $100-300 \mathrm{kHz}$


Geometric Mean Frequency, kHz
'f' Upper and Lower Frequencies vs Geometeric Mean Frequency $(0.265 \mathrm{mH}$ Factory Adjusted Type MWA Trap)


Frequency, kHz
' $\mathrm{g}^{\prime}$ Typical impedance vs frequency ( 0.265 mH )
Factory Adjusted Type MWA Trap)

'h' Upper and Lower Frequencies vs Geometeric Mean Frequency ( 1.590 mH Factory Adjusted Type MWA Trap)


Erequency, kHz
'i' Typical impedance vs frequency (1.590mH) Factory Adjusted Type MWA Trap)

Figure A-2 TYPICAL LINE TRAP CHARACTERISTICS

'a' Single frequency tuner

'b' Double frequency tuner for one coaxial cable

'c' Double frequency tuner for two coaxial cables

Figure A-3 SCHEMATIC DIAGRAM OF SINGLE AND DOUBLE


Frequency response Type jz 716 Tuner
curve 1 tuned to 30 kHz
curve 2 tuned to 100 kHz curve 3 tuned to 300 kHz
coupling capacitor 1870 MMF , Load resistance 300 Ohms 'a' Response curves of jz 716 Single Frequency Tuner

Frequency in kHz


Frequency response Type jz 726 Line Tuner
Coupling Capacitor 1870 MMF, Load resistance 300 OHMS
'b' Response curves of jz 726 Two frequency tuner

Figure A-4 TYPICAL FREQUENCY RESPONSE CURVES OF SINGLE AND DOUBLE FREQUENCY RESONANT LINE TUNERS

'a' Wide Band High Pass Filter (Hi-Coupler)

'b' Equivalent circuit of Hi-Coupler

'c' Hi-Coupler Insertion Loss for a 500 Ohm Load

Figure A-5 SCHEMATIC DIAGRAM, EQUIVALENT CIRCUIT, TYPICAL RESPONSE CURVES FOR A HI-COUPLER UNIT


Figure A-6 ATTENUATION OF RG $8 / \mathrm{U}$ COAXIAL CABLE AT CARRIER FREOUENCIES

'a' Typical Attenuation curves for Power Lines at $34.5,115,345$, and 765 kV

'b' Typical Attenuation curves for Power Lines at $69,138,230$, and 500 kV

Figure A-7 LINE ATTENUATION VS CARRIER FREQUENCY FOR DIFFERENT SYSTEM VOLTAGES


Figure A-8 TYPICAL AVERAGE NOISE LEVELS ON A 230 kV LINE IN A 3 kHz BANDWIDTH

## Table A-1

## Coupling capacitor rating

| System Voltage (kV) | $\begin{array}{c}\text { Capacitance (uF) } \\ \text { PCA-5 }\end{array}$ |  |
| :---: | :---: | :---: |
|  |  | PCA-7 |$]$|  | 0.01 | - |
| :--- | :--- | :--- |
| 69 | 0.006 | 0.025 |
| 115 | 0.005 | 0.021 |
| 138 | 0.0043 | 0.018 |
| 161 | 0.003 | 0.012 |
| 230 | 0.002 | 0.008 |
| 345 | 0.0015 | 0.006 |
| 500 | - | 0.00376 |

Table A-2

Power Voltage or Current Ratios,
and Decibels equivalents

| Power Ratio | Voltage or | Decibels |
| :---: | :---: | :---: |
| Current Ratio | (dB) |  |


| 1.26 | 1.12 | 1.0 |
| :--- | :--- | :--- |
| 1.58 | 1.26 | 2.0 |
| 2.00 | 1.41 | 3.0 |
| 4.00 | 2.00 | 6.0 |
| 10.00 | 3.16 | 10.0 |
| 100.00 | 10.00 | 20.00 |
| 1000.00 | 31.60 | 30.00 |
| $10^{4}$ | 100.00 | 40.00 |
| $10^{5}$ | 316.2 | 50.00 |
| $10^{6}$ | 1090.00 | 60.00 |
| $10^{8}$ | $10^{4}$ |  |

Table A-3a
Attenuation Correction for Foul Weather

$$
\begin{array}{ll}
34.138 \mathrm{kV} & \text { Add } 50 \text { percent } \\
230.765 \mathrm{kV} & \text { Add } 25 \text { percent }
\end{array}
$$

## Table A-3b

Transposition Correction in dB for 345 kV \& higher

| Number $<10 \mathrm{mi}$ | $>100 \mathrm{ml}$ |  |
| :--- | :---: | :---: |
|  |  |  |
| 1 | 0 | 6 |
| 2.4 | 0 | 8 |
| 5 or more | 0 | 10 |

## Table A-4

Coupling Corrections for Two Terminals - in dB

| Type of Coupling | $>50-\mathrm{ml}$ <br> line |
| :--- | :---: |
| Mode 1 <br> Center-to-outer | -2 |
| Phase <br> Center-to-ground | 0 |
| Al or Cu gnd Wire +1 <br> Steel gnd wire  <br> Outer-to-outer  <br> Phase (Push-  <br> Push Op.)  | +4 |

## Table A-5

Range of Characteristic Impedance of
Power-line Carrier Circuits

Table A-6

Noise Level Correction Factor for other Ine Voltages

| Voltages | Correction Factor |
| :---: | :---: |
| $(\mathrm{kV})$ | $(\mathrm{dB})$ |

66-115 -8
138-161 -4
230** 0
$345^{*}+2$
$500^{\star}+5$
765* +12

* bundled conductors

Table A-7
Correction Factor for other Receiver Bandwidths

## APPENDIX B

## STACK-TUNER FREOUENCY RESPONSE DATA

In Chapter 3.1, the design aspects of the stack tuner have been explained. The output gain versus frequency characteristics have been plotted in the open mode and the close mode of the stack tuner when tuned in for the different frequencies $300 \pm 2.5 \mathrm{kHz}, 200 \pm 2.5 \mathrm{kHz}, 100 \pm 2.5$ $\mathrm{kHz}, 200 \pm 100 \mathrm{kHz}$, and $50 \pm 40 \mathrm{kHz}$. Similarly, the impedance versus frequency characteristics have been plotted. Since most of the simulated test results are based on the stack tuner mid-band frequency of $300 \pm 2.5$ kHz , the actual data of the stack tuner when tuned in for the mid-band frequency of $300 \pm 2.5 \mathrm{kHz}$ are presented in Table B-1. The column 1 shows the frequencies. The column 2 shows the short/close mode impedance. The column 3 shows the open mode impedance. The column 4 shows the open mode gain. The column 5 shows the short/close mode gain. The gain has been defined in Chapter 3.1.

Table B－1
Stack tuner frequency response data

## Frequency Close mode Open mode Open mode Close mode impedance impedance gain gain

9世6日，क0 $\therefore 000.00$ 4000.00 $\therefore \theta \theta \theta \cdot \theta \theta$ K00日． 00 $7000.0 日$ ब000． $0 \theta$ $9000.0 日$ 10000.00 1．1000， 00 1． 2000,00 1． 3000.00 1． 700 回 $1 \% 000.00$ 16000.00 $\therefore 7000.0 日$ 1.0000 .00 1． 9000.00 $20 日 \theta O . A 0$ $\because 1000,00$ 22000.00 $\therefore 3000.00$ $\therefore 4000.00$ $\therefore \therefore 000.00$ $\therefore 600 日 .00$ $\therefore 7000.00$ 28000.00 $\therefore 9000.00$ 30000.00 $\because 1.900 .00$ 32000.00 3.3000 .00 34000.60 $\therefore 3000.00$ 36000．00 $\because フ 000.00$ 38000.00 $\because 9000.00$ 40000.00 41000.00 42000． 00 43000.00 44000.00 45000.00 46000． 00 47000.00 48000．00 49000.00 50000.00 $\because 1.000,00$ 52000,00 $\because 3000.00$ 54000.00
 56000．00
 58000.00 $\because 9000.00$ $\therefore 0000.00$

| \％99\％，16 | $\therefore 968$ ¢ |
| :---: | :---: |
| $26 \times 6.6$ | 30984.66 |
| $19 \% 069$ | ，\％4\％\％ |
| $16919.1 \%$ | 29467 ，16 |
| 13264．36 | 20660.31 |
| 11367．82 | 19496.39 |
| 9946 | 1973.26 |
| 8830.5 | 181．39．03 |
| 7959.97 | 17796 |
| 7200.31 | 17411.31 |
| 6624.27 | 17169．36 |
| 6113.02 | 16973，6 |
| \％674．69 | 1．60\％ |
| －294．67 | 1．6701， $4 \%$ |
| 9962.0 | I6\％9\％，旬品 |
| 4668.46 | 1650.6 |
| 46976 | 1． 6441.74 |
| $4.73,71$ | 16360.6 |
| 3963.31 | 16328，3！ |
| 3772.86 | 1． 6 ¢35．L4 |
| 3599.6 | 16243.93 |
| 3441.45 | 16209，5 |
| 3295.31 | 16177， 67 |
| 3162．74 | $161=6.67$ |
| 3039.37 | 16198．99 |
| 909．07 | 1． 6107 ， 83 |
| 2818.80 | ．1． 6888.9 L |
| $2719.9 \%$ | 1． 607 l 1． 87 |
| 2697.6 | 1． 6 \％ 6 ， 6 |
| 254 L ，67 | 1．6042．60 |
| 9459， 93 | 1．6020．3 |
| 2383.45 | 16018.41 |
| 2311.81 | 1．6007， 28 |
| 2244,193 | 15990.24 |
| 2179.96 | 11：9989， 58 |
| $21.18,31$ | 上 9981.20 |
| 2061.90 | 1．6973．7！ |
| 2007.90 | 15966.76 |
| 1．956， 38 | 15960，32 |
| 186玉．86 | 15954，35 |
| 1858.74 | 15948．30 |
| 1913.78 | 15943.63 |
| 1770． 32 | 15938，80 |
| 1729.73 | 15934.30 |
| 1690.38 | 15930.98 |
| 1652．60 | 15926．13 |
| 1616.51 | 1592つ， 43 |
| 1581．79 | 15918．95 |
| 1548．40 | 15916．67 |
| 15.16 .30 | 1591．5．59 |
| 1485．40 | 1590968 |
| 1455.63 | 15906.94 |
| 1426.93 | 159\％4．35 |
| 1．399，25 | 15904．94 |
| 1，372． | 1．5899．58 |
| 1.346 .70 | 1：397．38 |
| 1321.73 | 1589\％． 29 |
| 1297.5 | 17893.32 |
| 1274．22 | $1 \div 891.43$ |

$\therefore 69 \% \quad 9$ － 29467,10 20660,31 19496.99 1076,26 16159.0 17411,31 17169,36 0.776 1．6701， $4 \%$ ज65\％， 0 164017 16600,6 ， 21 16243.68 16909.5 $161 \div 2.67$ 16 LQ .99 1． 6107.86 1．6888， 9 L 16tion ：
1．6042． 6
1．6029．93
16016．41
1．6007， 28

1：988
士气゙981，2ず
15973．7．1
06.7
15954.35
15948.30
$15943 \cdot 63$
15938,80
15934.30
15926.13

1 5タつつ，43
15918.95

159世㤩，67
15915.59
15909.68
15906.94
15904.35

13901．9
15897.38
15895.29
17893.32 1274． 22 1

| 日，ल10］ | 1， 0 |
| :---: | :---: |
| （0．（0） | （9， 8 |
| （1）．69 | （1）同 |
| 0．（0） | （0，0） |
| 0.00 | （b） 0 |
| （3）A | （－9） |
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Table B－1
Stack tuner frequency response data
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| 26306.64 | 30804.66 | 0.00 | Q． 00 |
| 19900．94 | 25435.26 | 9.90 | 0．00 |
| 15919.17 | 29467.18 | 0.00 | 0.00 |
| 13264.36 | 20660.31 | 0．00 | 9， 08 |
| 11367,80 | 19496．99 | （a）00） | （－3） 0 |
| 9945.20 | 18703.26 | 0.00 | a，00 |
| 8039． | 19139， 03 | （1）．00 | 0.00 |
| $79 \% 29$ | 17794，43 | 0.00 | Q． $0^{0}$ |
| 7208.31 | 17411.31 | 0.00 | Q， 6 |
| 6624.27 | 17169，34 | 0.00 | （a）0 |
| 6113.02 | 16978．69 | 0.00 | 0， 00 |
| W6y7．63 | 1682\％． 90 | 0.00 | 0.00 |
| 5298.67 | 1．6701，47 | 0.00 | （1）．00 |
| 4962．06 | 16599．02 | 0.00 |  |
| 4668.46 | 16513．64 | 0．00 | Q． 08 |
| 4407.89 | 16441.74 | 0.00 | （3） 013 |
| 4173.71 | 16300．6\％ | 0.00 | 0． 00 |
| 3963.31 | 16328．31 | 0.00 | $\theta \cdot 0$ |
| 3772.86 | 16283．14 | 0．00 | （0．00 |
| 3599 | 16243.68 | 0.00 | Q， 0 |
| 3441.43 | 16209.55 | 0.00 | 0.00 |
| 3296.31 | 16179.37 | 0.00 | 0.00 |
| 3162，74 | 16162．69 | 0．00 | 0.00 |
| 3039.37 | 1．6129．99 | 0.0 | 0． 00 |
| 2929.07 | 16107．96 | 0． 00 | 0.00 |
| 2818.90 | 16088．9\％ | 0． 00 | 0．00 |
| 2719.90 | 16871.87 | （0．00 | 0.00 |
| 2627.6 | 16056.6 | Q． 00 | （8） 8 |
| 254\％， 97 | 1．6042，50 | 日． 00 | Q． 00 |
| 2459.93 | 16029，93 | 0.00 | Q，00 |
| 2383.6 | 16018.41 | 0． 00 | 0.00 |
| 2311.81 | 1．6007．88 | 0． 08 | （1） B |
| 2244.6 | 15998．24 | 0.00 | （2）00 |
| 2179.96 | 15989．38 | Q． 00 | （0， 00 |
| 2119.31 | 15981， 23 | 0．00 | Q． 00 |
| 2061.80 | 15973．71． | Q， 0 | 0.00 |
| 2007.20 | 15966.76 | 0．00 | 0．00 |
| 1955．28 | 15960.32 | 0.00 | 0.00 |
| 1905.86 | 15954．36 | 0．00 | 0.00 |
| 1859.74 | 15948．90 | 0.00 | 0.00 |
| 1813.78 | 15943．63 | 0.00 | 0.00 |
| 1770．82 | 15938．80 | 0.00 | 0．00 0 |
| 1729.73 | 15934．30 | 0．00 | 0.00 |
| 1690．38 | 15930．98 | 0.00 | 0．00 |
| 165\％．68 | 15926．13 | 0.00 | 0.00 |
| 1616.51. | 15922．43 | 0.00 | 0．00 |
| 1581.78 | 15918．95 | 0.00 | 0.00 |
| 1548.40 | 1591567 | 0.00 | 0.00 |
| 1516.30 | 15912． 58 | 0.00 | 0.00 |
| 1485．40 | 15909．68 | 0.00 | 0.00 |
| 1455.63 | 15906.94 | 0.00 | 0.00 |
| 1426.93 | 15904.35 | 0．00 | 0.00 |
| 1399．25 | 1\％901．90 | 0.00 | 0.00 |
| 1372． 52 | 15899．58 | 0．00 | 0.00 |
| 1．346．70 | 15897.38 | 0.00 | 0.00 |
| 1321.73 | 15895 | 0.00 | 0.00 |
| 1297.99 | 1．5893．32 | 0.00 | 0.00 |
| 1974．23 | 16891.43 | 0.00 | 0.00 |

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| 674.13 | 15855．34 |
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| 111.06 | 1507743 |
| 109.66 | 16878.9\% |
| 189.46 | 1 1980.49 |
| 109.97 | 15082.16 |
| 103.69 | 15683.93 |
| 180.93 | 16885. 61 |
| 98.77 | 1)997.80 |
| 96.6 | 15869.92 |
| 94.49 | 1589\%, 17 |
| 92.36 | 15894.67 |
| 90. 20 | 1.6897. 1.3 |
| 63.14 | 15399.06 |
| 06.04 | 1.5902.79 |
| 83.96 | $1596 \%$ |
| 91.98 | 15909.27 |
| $7 \% .81$ | 15\%12.80 |
| 77.76 | 1.7916.76 |
| 76.71 | 15920.94 |
| 73.67 | 1.592546 |
| 71.64 | 1.5930, 34 |
| 69.61 | 15935.63 |
| 67.60 | 1.5941.36 |
| 65.60 | 1594763 |
| 63.68 | 159\%4.450 |
| 61.61 | 15961.90 |
|  | 15970.08 |
| 57.66 | 15979.06 |
| \%9,70 | 15998.97 |
| 63.75 | 15999.92 |
| W1. 31 | 16012.06 |
| 49,87 | 16025.6\% |
| 47.94 | 16040.73 |
| 46.02 | 1.60 .57 .69 |
| 44.11 | 16076.82 |
| 42.31 | 16093.47 |
| 40.31 | 16133.11 |
| 39.43 | 16151.30 |
| 36. 5 | 16183.76 |
| 34.68 | 16201.36 |
| 32.82 | 16265.24 |
| 30.97 | 1.6316 .84 |
| 29.12 | 16370.05 |
| 27.29 | 16451.36 |
| 25.47 | 16540.08 |
| 23.65 | 16648.73 |
| 21.85 | 16783.56 |
| 20.06 | 1.6953 .38 |
| 1.8.99 | 17170.88 |
| 16.53 | 17454.71 |
| 14.80 | 17833.03 |
| 13.69 | 1.8349.93 |
| 11. 4.1 | 19071. 5 |
| 9.78 | 20108.03 |
| 8.23 | 2162020 |
| 6.79 | 33846.17 |
| \%, | 26906.46 |
| 4.69 | 30293.45 |


| (1).04 | (ia) $0 \times$ |
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| (a) 04 | 6. 04 |
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| 0, 0\% | 0.04 |
| Q. 0 | 0.05 |
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| 9, 96 | O. 0 \% |
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| Q. 07 | 日. 0.6 |
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| 0.07 | 0, 06 |
| 0.87 | 日, 97 |
| Q. 08 | 0.07 |
| 0.08 | 0.07 |
| 0.08 | 0. $0 \%$ |
| 0.88 | 0.08 |
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| 0.10 | 0, 09 |
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| (3. .1. 1 | (0.10 |
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| (0.12 | (0.11 |
| 0.12 | (0.12 |
| 0.1 .3 | 0.12 |
| (1).1.4 | 0.13 |
| 9.13 | 0. 14 |
| 0.16 | (0. 1.3 |
| 0.17 | 0.16 |
| 0. 18 | Q. 17 |
| 0.19 | 0.18 |
| 0.21 | 0.20 |
| 0.22 | 0.22 |
| 0.24 | 0.24 |
| 0. 27 | 0. 27 |
| (1.29 | 0, 31 |
| Q.32 | 0.35 |
| 0. 36 | (1, 41 |
| 0.39 | 0.49 |
| 0.43 | 0.6 |
| 8. $4 \%$ | 0.72 |
| 9.49 | 0, $8 \%$ |

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$\therefore .79$ 2966\％ 6 6，71 26462． 1
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 ：\％．04 17982， 2 W8．62 15974．60 60.2015867 .67 $61.77 \quad 15961.29$ $63.34 \quad 15955.36$ $64.91 \quad 15949.91$ 66.4715944 .83 68.0315940 .11 $69.58 \quad 16935,71$ 71.13 15931．61 72.68 15927．77 74．22 1．5924．18 $75.76 \quad 15920.82$ 77.30 15917．66 $79.83 \quad 15914.69$
$89.36 \quad 15911.90$ 81.89 15909．27
83.4115906 .78
$84.93 \quad 15904.44$
86.4415902 .22

87．95 15900．12
$89.46 \quad 15898.13$
$90.96 \quad 15896.24$
$92.46 \quad 15894.45$
$93.96 \quad 15892,75$
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| 160，7\％ | 1689 |
| 119．21 | $159 \%$ |
| I．11． 1.6 | 1\％9\％ |
| 113.12 | 15 |
| 11．4．67 | 1587\％ 7 |
| I I 6，02 | 15874．6\％ |
| 117.47 | 19874.8 |
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| 120．3世 | 158724 4 |
| 121.79 | 1 1－ |
| 123.2 | 159\％ 16 |
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| 127．\％1 | 15869．02 |
| 128.93 | 15968．40 |
| 130．3 | 15867．8\％ |
| $1.31 .7 \%$ | 15867．24 |
| 133.18 | 15866.69 |
| 1．34． 99 | 15966． |
| 136．00 | 15863．64 |
| 37.41 | 15965．14 |
| 138.81 | 15864.66 |
| 148.21 | 15964.19 |
| 1．4．6． | 15963.74 |
| 1.43 .00 | 16863 |
| 144.40 | 15962． 69 |
| 45.79 | 1.5862 |
| 147.17 | 15962．08 |
| 48,6 | 1：861．70 |
| 149.94 | 1\％861．32 |
| 151，32 | 15860.96 |
| 152， 70 | 15960．61 |
| 1． 1.6407 | 16860.27 |
| 155．45 | 15859.94 |
| 156，82 | 15859.62 |
| 158．18 | 15859．31 |
| 159.55 | 1585901 |
| 60.91 | 15858.71 |
| 1．62．27 | 15858．43 |
| 1．63．63 | 15858．15 |
| 164.99 | 15857．88 |
| 166．34 | 15857．62 |
| 167.69 | 15857.37 |
| 169．04 | 15857．12 |
| 170．39 | 15856.88 |
| 171.73 | 15856.64 |
| 173.07 | 1．5856．41 |
| 174．41 | 15856．19 |
| 176．70 | 15859 |
| 177．09 | 15855.76 |
| 178．42 | $168 \%$ \％6 |
| 179．7\％ | 1585936 |


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1． 6,4 $16 \% . \%$ $16 \%, 06$ 186.30 $10 \% .69$ 189.01 190.33 191.64 192.96 194．26 196,6 1.96 .67 190.17 199.47 20． 77 202.07 203.36 204． 6 20… 9 207.23 203.6 209.01 211.09 210.37 213.65 214.93 216.21 217.48 218.76 20． 221.30 202． 66 223.93 226． 09 226.36 227．62 228．80 230.14 231.39 232． 63 233.90 235.15 236.40 237.69 238．89 240.14 241． 38 242．62 243． 86 245.10 246． 34 247.58 248.81 250.04 254.27 252.50 253.73 254.96 256.18
$168 \% 4$ $169 \% 4.79$ $16 \% 4.61$ $150 \% 4.43$ 1565426 $158 \%$ ， 10 15053.93 $1 w 963.77$ 1685 ． 2 $168 \% 3.4$ 15953.32 1585.18 15853.03 1．68\％．90 1 世\％\％．76 1．56\％，63 $1 \% \%$ ． 15652.36 $1 \% 8 \% 2.26$ 1． $58 \%, 14$ $1 \% 8 \% 202$ 16891．91 15851．90 15851.69 15851.59 1．5651．43 15851， 38 $1585 \% .20$ 169世\％，19 $156 \% \% .09$ $158 \%$ 15360.9 1580.82 15050,74 15850，6＂ $15860, \% 7$ 15850.49 $150 \%$ ． 41 15950.34 15850.26 15850.19 15850.12 15850.05 15849.93 15849．91． 15849.85 15849.78 15849.72 15849．66 15849.60 15849．54 15849.49 15849.43 15849.38 15849.32 $1: 849.29$ 15849．22 1． 5849,17 15849.12
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| $259.55$ | $15848$ |
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| 261.67 | 15948.94 |
| 26.29 | 15040.90 |
| 263.51. | 15348.06 |
| 264.72 | 15843.81 |
| $26 \% .93$ | 15848,77 |
| 267.15 | 15848.74 |
| 263.36 | 15043.70 |
| 269.57 | 15849.66 |
| 270.78 | 15040.62 |
| 271.93 | 15848.59 |
| 273.19 | 1.5843.5\% |
| 274.40 | 15040.52 |
| 275.60 | 1.6843.49 |
| 276.80 | 15043.46 |
| 278.00 | 15043.42 |
| 279.20 | 15843.39 |
| 2ea. 40 | 1.5848.36 |
| 231.60 | 1.5949.34 |
| 282.79 | 15040.31 |
| 283.99 | 15048.28 |
| 285.18 | 15943.25 |
| 286.37 | 15848.23 |
| 287.36 | 15843.29 |
| 283.7\% | 15943.18 |
| 289.94 | 15848, 15 |
| 291.13 | 15848.13 |
| 292.31 | 15848.11. |
| 293.51 | 15848.09 |
| 294.48 | 15848.97 |
| 295.86 | 15848.04 |
| 297.04 | 15648. 92 |
| 298.22 | 15948.01. |
| 299.40 | 15847.99 |
| 300.50 | 15947.97 |
| 301.76 | 15847.95 |
| 302.93 | 15847.93 |
| 304.10 | 15347.92 |
| 305.28 | 15847.90 |
| 306.45 | 15847.89 |
| 307.62 | 15847. 67 |
| 308.79 | 15847.86 |
| 309.96 | 1.5847.84 |
| 311.13 | 15847.83 |
| 312.29 | 15847. 82 |
| 313.46 | 15847. 80 |
| 314.62 | 15847.79 |
| 31.5 .78 | 15847.78 |
| 316.95 | 1.5847.77 |
| 318. 11. | 15847.76 |
| 319.27 | 15847.75 |
| 320.43 | 15847.74 |
| 321.58 | 15847.73 |
| 322.74 | 1.5847 .72 |
| 323.90 | 15847.72 |
| 325.95 | 15847.71 |
| 326.20 | 15847.70 |
| 327.36 | 1.6847 .69 |

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|  | 16847.66 |
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| ब1 | $1584 \%$ 60 |
| 31.96 | 13647.67 |
| 33, 11 | 16847.66 |
| 34.25 | 1584\%.60 |
| $3 \%$ | 15647.66 |
| 336 | $1:$ |
| 337.69 | 1. |
| 370.63 | 13 |
| 339.97 | 15047.64 |
| 34.1 .11 | 15847.64 |
| 342, 2 | 1564\%.64 |
| 343.39 | 15947,64 |
| 344. 3 | 1.6847.64 |
| 346.67 | 15847.64 |
| 346.90 | 1684\%.63 |
| 347.94 | 16847.63 |
| 349.0\% | 16847.63 |
| 360.20 | 158447.64 |
| 361.34 | 16047.64 |
| 3924 | $1584 \% .64$ |
| 353.60 | 15047 |
| 354.73 | 1.6847.64 |
| 3'5.86 | 15847.64 |
| 356.99 | 15847.64 |
| 358.11 | 15847.60 |
| 369.24 | 16847.6 |
| 360.37 | 15847,6\% |
| 361.49 | 15947.66 |
| 362.61 | $11584 \% .66$ |
| 363.74 | 16847.66 |
| 364.86 | 15647.67 |
| 365.98 | 16847.67 |
| 367.10 | 115047.68 |
| 360.29 | 15847.68 |
| 369.34 | 15947.69 |
| 370.46 | 15947.69 |
| 371.57 | 1.5947 .76 |
| 372.69 | 15847.71 |
| 373.81. | 15847,71. |
| 374.92 | 15847.72 |
| 376.03 | 15847.73 |
| 377.15 | 1.5847 .73 |
| 378.26 | 15847.74 |
| 379.37 | 15847.75 |
| 380.48 | 15847.76 |
| 381.59 | 1.5847 .76 |
| 392.70 | 15847.77 |
| 383.81 | 15847 |
| 384.92 | 15847.79 |
| 386.02 | 15847.80 |
| 387.13 | 15847.81 |
| 388, 23 | 15847.82 |
| 389.34 | 1.584\%.83 |
| 390.44 | 15847.84 |
| 391.54 | 15947.85 |
| 392.65 | 1.5847.86 |
| 393.75 | 15847.87 |
| 394.86 | 1.5847 . |


| 6. bi | (4.3) |
| :---: | :---: |
| 0, 01. | (1) 61 |
| 0.0. | (0.01 |
| 9. 01 | 0, 01 |
| 0.01 | 0. 010 |
| (2) 01 | Q, 01 |
| 0.01 | (0) 01 |
| Q. 01 | Q, 01 |
| a. ${ }^{0}$ | (a) 01 |
| 0.0 l | Q, 01 |
| a, 01 | Q, 01 |
| Q, 01 | 0.01 |
| 0.01 | (3) 01 |
| 0.01 | Q. 01 |
| 9.01 | Q, 0 |
| 0.01 | 0.01 |
| 0.01 | 0.0.01 |
| a. 01 | Q, 01. |
| 0.0 | (1) , \% |
| a. 01 | (1), 0.1 |
| 0.01 | O. 0.1 |
| 0.01 | 0.01 |
| 0.01 | (0) B |
| 0.01 | 0.0.1 |
| 0.01 | (0.0) |
| 6, 0.1 | (0, 01 |
| 0.01 | (0) 0 |
| 6. 01 | Q. 01 |
| 0.01 | (0.0) |
| 0.01 | 0.01 |
| 0.01 | 0, 01 |
| 0.01 | (3) 01 |
| Q. 8 | Q, 01 |
| (a) 8 | Q. 81 |
| 0.01 | Q. 01 |
| 0.01 | Q. 01 |
| Q, Q1 | (0.0) |
| 0.01. | 0.01 |
| 0.01 | Q. $0 . \mathrm{L}$ |
| 0.0.1 | 0.01 |
| 0.01 | $0.0) \mathrm{l}$ |
| 0.01 | 0.01 |
| 0.01 | Q, 01 |
| 0.01 | 0.01 |
| 0.01 | 0.0.1 |
| Q. 01 | 0. 0.1 |
| Q. 01 | O. 0.1 |
| (0.01 | 0.01 |
| 0.01 | 0.01 |
| O. 0. | Q. 01 |
| Q, 0.1 | Q. 01 |
| 0.01 | 0.01 |
| Q. 01 | 0.01 |
| 0.01 | 0.0 .1 |
| (1, 0] | 0. 01 |
| Q, 0] | 0.01 |
| 0.01 | 0.01 |
| 0. 01 | $\theta \cdot \theta 1$ |
| 0.01. | 0.01 |

## APPENDIX C

## CAICULATION OF TRANSMISSION LINE BASIC PARAMETERS

The basic line parameters are the series impedance [Z] and the shunt admittance [Y]. These are usually considered as being uniformally disrtibuted along the whole length of the line. For a multiconductor line, the basic line parameter matrices have been formulated [49] for realistic transient studies and are presented in this appendix.

## SERIES IMPEDANCE MATRIX

The series impedance matrix [Z] consists of 3 main components:
1.the impedance due to physical geometry of the conductor [ $Z_{g}$ ] which is due to electromagnetic coupling between them,
2. the internal or self impedance $\left[Z_{S F}\right]$,
3. the impedance which accounts for the earth return path $\left[Z_{e}\right]$.

Thus, the series impedance matrix of a multiconductor line is of the form;

$$
\begin{equation*}
[z]=\left[Z_{g}+Z_{S F}+Z_{e}\right] \tag{C1}
\end{equation*}
$$

The series impedance matrix [Z] and each of its component matrix are of the order $3 p+q$, where $p=$ number of circuits and $q$ = number of earth wires.

## Impedance due to physical geometry:

The impedance matrix due to the physical geometry of the line conductors as shown in Figure C-1:

$$
\begin{equation*}
\left[z_{g}\right]=j X_{g}=j w u / 2 \pi \quad[B] \tag{C2}
\end{equation*}
$$

[B] is known as the charge-coefficient matrix and its elements are defined as:

$$
\begin{equation*}
B_{i j}=\log _{e}\left(D_{i j} / d_{i j}\right) \tag{C3}
\end{equation*}
$$

where
$D_{i j}=$ distance between the $i$ th conductor and image of the $j$ th conductor,
$d_{i j} \neq$ distance between the $i$ th conductor and $j$ th conductor (for i = j)

```
                        = radius of the i th conductor for ( }i=j\mathrm{ )
```


## The self impedance of the conductor:

Consider a single circuit EHV line with two earth wires, the self impedance matrix of the line conductor is one where the diagonal elements represent the self impedance of the phase conductors and earth wires, and whose offdiagonal elements are zero as given by Equation (C4)
$\left[Z_{S F}\right]=\operatorname{Diag}\left[Z_{s f}+Z^{S f}+Z^{s f}+Z^{s e}+Z_{S e}\right]$
Since it is a common practice to bundle the conductors of EHV lines, the self impedance of the bundle conductors is:

$$
\mathrm{Z}_{\mathrm{sf}}=\mathrm{Z}_{\mathrm{SSf}} / \mathrm{N}_{\mathrm{b}}
$$

where,
$Z_{\text {sSf }}=$ self-impedance of a sub-conductor of the bundle $\mathrm{N}_{\mathrm{b}}=$ number of sub-conductors consistig the bundle.

The calculation is difficult due to various factors: nonuniform current density, steel core, stranding of conductors etc.

Different methods have been suggested for low frequency (below 500 Hz ) and high frequency (above 500 Hz ) applications.

For low frequency .applications, uniform current
disrtibution is assumed and influence of steel core or stranding of conductors is neglected. In such case elements of matrix are given by:

$$
\begin{equation*}
\left[Z_{S F}\right]=R_{d c}+j w u / 2 \pi \quad \log _{e}\left(r / r_{g m}\right) \tag{C5}
\end{equation*}
$$

where, $R_{d c}=$ dc resistance
r = phase conductor or earth-wire radius
$r_{g m}=$ phase conductor or earth wire geometric mean radius given by the manufacturer.

$$
\mathrm{R}_{\mathrm{dc}}=\frac{\mathrm{R}_{\mathrm{c}}}{\mathrm{n} \pi \mathrm{r}_{\mathrm{s}}^{2}}
$$

$$
\text { where, } \begin{aligned}
\mathrm{R}_{\mathrm{C}} & =\text { conductor resistivity } \\
\mathrm{n} & =\text { number of conductor strands } \\
r_{S} & =\text { radius of earth strand }
\end{aligned}
$$

For high frequency applications, the skin effect plays an important role and is responsible for non-uniform current distribution. In such a case, resistance increases and inductance decreases with the increase of frequency. Figure C-2 shows a typical variation of resistance and self inductance with the increase of frequency. This information is normally stored in a piece-wise linearised form for computational purposes [6].

For very high frequency applications(above 2.5 kHz ), Galloway [49] has suggested a method:

$$
z_{s f}=R_{s f}+j X_{s f}
$$

where,

$$
\begin{equation*}
R_{s f}=X_{s f}=k_{1} R_{c} m /\left[2 r_{o}\left(n_{o}+2\right) \pi\right]^{1 / 2} \tag{C7}
\end{equation*}
$$

and, $m=\left[w u / R_{c}\right]^{1 / 2}$

$$
\begin{aligned}
& r_{0}=\text { radius of each outer strand } \\
& n_{0}=\text { number of strands in the outer layer } \\
& k_{1}=a \text { factor accountig for conductor stranding }
\end{aligned}
$$

$$
(=2.5)
$$

In the present work, the standard conductor is substituted by a solid aluminium conductor of the same overall radius, and the self impedance is evaluated on the basis of nonuniform current density due to skin effect. This method is commonly used for the evaluation of the self impedances of ACSR conductors [50].

## Impedance due to earth return path:

Carson [51] has suggested a method to find out the impedance due to earth return path by using an infinite
series. Galloway [49] has expressed his method for computatioal purpose as:

$$
\begin{equation*}
\left[Z_{e}\right]=2 w \mu / 2 \pi[P+j Q] \tag{C8}
\end{equation*}
$$

where $\mu=$ permeability
$P$ and $Q$ are calculated in terms of $r_{i j}$ and $\Theta_{i j}$ as indicated in the Figure C-1 where,

$$
r_{i j}=\left[\left(\begin{array}{ll}
\left(w / R_{c}\right) & D_{i j}
\end{array}\right]^{1 / 2},\right.
$$

and have different formulas [49] according to the value of $r_{i j}$ greater than $5,\left(r_{i j}>5\right)$, or smaller or equal to 5 $\left(r_{i j} \leq 5\right)$.

## SHUNT ADMITTANCE MATRIX

Neglecting the air conductance of free-space, the shuntadmittance matrix contains only imaginary terms corrosponding to the capacitance coupling between the conductors, and conductors and earth. It is a matrix of the order of $3 p+q$, where $p$ is number of circuits and $q$ is number of earth-wires conductors. It is expressed as:

$$
\begin{equation*}
[Y]=j 2 \pi w \in \quad[B]^{-1} \tag{C9}
\end{equation*}
$$

where, [B] ${ }^{-1}$ is the inverse of the matrix [B] given earlier

$$
\text { Let }[C]=2 \pi \epsilon \quad[B]^{-1}
$$

Then, the elements of the matrix [C] define the various capacitances as below:

$$
\begin{gathered}
C_{i j}=-C_{i j}, \text { for }(i \neq j) \\
C_{i j}=\begin{array}{l}
j=3 p+q \\
\sum C_{i j}, \text { for }(i=j) \\
j=1
\end{array}
\end{gathered}
$$

where $C_{i j}=$ capacitance between conductors $i$ and $j$, for $i \neq$ j
$C_{i j}=$ capacitance between the conductor $i$ and the earth, for i $=$ j

## ELIMINATION OF EARTH WIRE

In the present study, voltage and current in the earth wire have not been considered , and therefore, it is suggested to eliminate the rows and columns corrosponding to the earth wire both in the series impedance matrix and shunt admittance matrix to make the analysis simplified. Moreover the performance of the conductors does not change [6,50]. For elimination, the following method has been used:
since earth-wires are earthed at regular intervals,
standing voltage between towers can be neglected and zero potential can be assumed in the earth wire. In such a case, the elements of the shunt admittance matrix corrosponding to the earth wire in the rows and columns can be discarded directly. In the series impedance matrix, the matrix is first inverted and then the elements in columns and rows corrosponding to the earth-wire are eliminated. Finally, the matrix is again re-inverted back.


FIGURE $\quad \mathbf{C - 1}$ SCEMATIC DIAGRAM OF CONDUCTOR POSITIONS


FIGURE C-2, VARIATION OF LINE INDUCTANCE (L) AND
RESISTANCE (R) WITH FREQUENCY

## APPENDIX D

## PROGRAM FOR BAND-PASS FILTER

In Chapter 5.3.2, the application of the band-pass filter has been explained. The band-pass filter has been used to filter most out of the frequency band $300 \pm 2.5 \mathrm{kHz}$ components. For this purpose the band-pass filter has been designed with nine sections. For the purpose of the simulation of the band-pass filter, the computer program has been shown in Figure $\mathrm{D}-1$. The design details are given in Figure D-2.
\$ TY BPFTLTER.FOR
$C$ Proaram to calculate the element values for
C Buttervorth band-pass filters
C for operation betveen equal source and load resistances. DIAENSION AR (10). $\dot{A} \mathrm{~L}(10), \dot{A C}(10) . \mathrm{B}(10), \mathrm{F}(10)$
10 CONTINUE WRITE $\mathbf{i 6 , *}$; How many sections in the filter?'
READic,*iN
IF (N.GT. 20i GOTO 500
IF (N.LT.ll) GOTO 20
URITE( $\epsilon, *)$ 'ERROR! No. of sections exceeds 10 '
GOTO 10
CONTINUE
WRITE( $6, *)$ 'Enter the Centre Freq. of the band-pass filter'
READ (5.*) FO
GRITE(6,*) 'Enter the band-width of the band-pass filter' READ(5,*) BU WRITE(G.*) 'Enter the surqe impedance of the line' READ(5,*)RL. $\mathrm{RS}=\mathrm{RI}$.
$\mathrm{PI}=3.1415926$
$\mathrm{HC}=2.0 * \mathrm{PI} * \mathrm{BH}$
$H 0=2.0 * P I * F O$
$F C 2=0.5 *(S O R T(i B H * B H)+4 * F O * F 0)+B H)$
$\mathrm{FCl}=\mathrm{FC} 2-\mathrm{BW}$
$C$ Hhere $F C 2$ and $F C 1$ are the upper and lover $3 d B$ cut-off freqs. CかLL BUTll(B, H ) $\mathrm{I}=1$
100 CONTINUE
$r$. The element values of the lov-pass prototype are nov beina calculated
$C$ The prototvpe network is of the form of Fig. 10.4 and Fig. 10.7
C
$A L(I)=B(I)$

$A R(I)=0.1$

$I=I+1$
IF (I.ET. H) GOTO 105
$A C(I)=B(I)$
$A C(I)=(A C(I) I /(H C * R L)$
$A R(I)=1.0 E+9$

$I=I+I$
IF (I.GT.N) GOTO 105
GOTO 100
105 CONTINUE
WRITEiG,*) 'The element values for a'
WRITE(G,*)N, 'Section band-pass Buttervorth Filter'
HRITE(6,*) ' '
URITE(G,*) 'RL $=$ ', RL,' ohms'
URITEIG.*i 'RS = ', RS,' ohms'
HRITE(G.*)
URITE(E,*i 'Centre-freq. = ',FO,' Hz'

```
WRTTE(6,*) 'Upper cut-off freq. = ',FC2.' Hz'
URITEf6.*' 'Lover Gut-off freq. = ',FCl,'Hz'
WRITF,(G,*) 'Baral-viath= ', BH.' Hz'
HRITE,GG.*)
WRITEIG.*।
GRITEiG.*` 'The elements Lili and Clli are'
HRITE!6.*) in series vith each other'
GRITE(G,*i ' ard the series combination of L(l) and C(l)'
URITE(G.*) 'is ar series vith RI,'
HRITE(6,*) '
```



```
HFTTE(6.*) 'in sifiem with each other'
WRITE{G,*i 'and the elements wnth even subscripts are'
HRITE(G,*) 'in parallel wath each other'
HRITE(G,*)
WRITEiE.*) 'The element values are: "
HRITEi6,* ।
HPITE|2t.*। 'The element values for a'
HRITEi24.*i N. 'Sectaon band-pass Buttervorth filter'
HRITE(24.*)
HRITE(2*,*i 'RL = ',RL.' ohms'
WRITE.(2\pm.*; 'RS = ',RS,' onms'
HRITE(2&.*)
HRITE(24,*) 'Centre-freq. = ',F0,' Hz'
HRITES24,*; 'Unper cut-nff freq. = '.FC2,' Hz'
HRITE(24,*) 'Lower こut-off freq. = ',FCl,'Hz'
HRITE(24.*) 'Band-vidth= '. BH.' Hz'
HRITE(2\pm.*)
HKITE(2G,*)
HRITEi2G.*i ' * ',N, ' Section band-pass Buttervorth Fidter'
WRITE(26,*i
HRITEI26.*; ' * RL = '.RL,' ohms'
HRITE(2G.*) ' * RS = '.RS;' ohmS'
URITE(2G.*)
MRITEi2G.*i ' * Centre-freq. = '.FO.' Hz'
WRITE,(26.*)' * Upper cut-off freq. = ',FC2.' Hz'
HRTTEiz.G.*; ' * Lover cut-off freq. = ',FCl.' Hz'
WRTTE(26,*) ' * Band-vidth= ',BH,' Hz'
HRITE(26,*) ' *
HRITE(24.*) 'The elements L(l) and C(l) are'
HRTTEi24,*i 'in serles with each other'
HRITE(2t,*; , and the series combination of L(l) and C(l)'
HRITE(24,*) 'is irl series vith RL'
WRITE(2\pm,*) '
HRITE(2t.*; 'The elements with odd subscripts are '
WRITE(2t,* ' 'IM series vith each other'
HRITE(24.*) 'and the elements with even subscripts are
WRITE(2.4,*) 'in parallel with each other'
HRITE(24.*)
HRTTE(24.*)
```

HRITE(24,*) 'The element values are:
DO $200 \mathrm{I}=1 \mathrm{~N}$
MRITE(6.*) 'I」'. I.' = ' ALfI), ' henrys'
URITE(6,*) 'Ci'.I.') = ' $\quad$ 'C(I), ' farads'
WRITE(6, *)
HFITEiG.*)
HRITE(2t,*) 'Li'.I,') = ' nLifi, ' nenrys'
WRITEi2t.*) 'C:'t.' = ' AC(I), farads'
VRITE(24, *) *
WRITFi2t,*)

CONTIHUE
HRITEfG.*i
HRITEI6.*
URITEi2t.*)
URITET2.**
IN $=1$
$[\mathrm{Nl}=3$
IN2 $=4$
Continue
WRITE(6.901)IN.INI.EN2. $\dot{\text { IC }}$ (IN)
WRITE (26.901)IN. THI, IN2.AC: (IN1
HRITE(24.901)IN,IN1.IN2, AC(IN)
FORHATi' C'.II.IX,I3.1X,I3.1X.EIO.4)
IN3=IN2+1
WRITE(6.902)TN.IN2.ING, AL(IN)
WRITE(26.902)IN.IN2, IN3, ALIIN)
HRITE:24.902iIH.IH?. IH3. ALITH
90.2

THA=THS+1

HRITER2t.903)IN, TH3, iNt, ARGINi
WRITE(26,903)1N, (H3, IN4.AR(IN)

IFIIN.EO.N GOTO 960
$I N=I N+1$
$A R(I N)=1.0 E+9$
WRITE(6.904)IN, IN4. AC(IN)
WRITE(26.904itN, IN4.AC(IN)
HRITE(24. 9041 TN. INt. AC(IN)
904
FORMAT('C', IL, IX, IB, IX, '0', 1X, E10.4)
URITE(6.305)IN.IN.t,ALITN,
WRITE(26.905)IN.IN4.AL(JN)
HRITE(24. 9051 IN .INt, AL (IN)
FORMAT' L', II, 1X, I 3, 1X, '0', 1X, E10.4)
HRITEI6,906iIN,INt,AR(IN)
WRITEi26.906iIN.INt.AR(IN)
WRITE(2t.906)IH. INt.nR(TN:
906
FORMAT:' R', IL, IX,I3, 1X, 'O', 1X, ElO.4
THT $=$ IN 4
$\operatorname{IN} 2=I N 4+1$
$I N=I N+1$
TF (IN.GT.N) GOTO 955
GOTO 950
955 CONTINUE
URITE(6,*) 'Error $n$ is even'
URITE(26,*i 'Error n is even'
960
CONTINUE
WRITE (6.907)IN4, RS
WRITE $26.90 \%$ )INt.RS
HKITE(24.907)IN土.RS
907 FORHAT('RS'.1X.I3.2X.'1', E10.4)
WRITEi6.908)RL
HRJTEi2 6.908 )RL
WRTTEM24.908IRL
908
FORAMT'' RL', LX,'3', 2X,'0', ElO. 4)
GOTO 10
500 COHTINUE
EMD
SUBROUTINE BUTII(A,N)
C This subroutine is to be used vhen $r=1$ in Table 10.1
DIMENSION A(N). B(10.10)
Dati $B / 2.0 .1 .4142 .1 .0 .0 .7654,0.618,0.5176,0.445,0.3902,0.3473$.
$10.3129 .0 .0 .1 .4142,2.0,1.8478,1.6180 .1 .4142,1.247,1.1111,1.0$.
$20.908 .2 * 0.0,1.0 .1 .8478 .2 .0,1.9319,1.8019 .1 .6629,1.5321,1.4142$,
$3 \quad 3 * 0.0,0.7654 .1 .618,1.9319 .2 .00 .1 .9616,1.8794,1.7820$,
$4 \quad 4 * 0.0,0.6180,1.4142,1.8019 .1 .9616 .2 .00,1.3754,5 * 0.0,0.5176$.
$5 \quad 1.2470,1.6629 .1 .8794,1.9754,6 \star 0.0 .0 .4450$.
$6 \quad 1.1111 .1 .5321,1.7820 .7 * 0.0 .0 .3902,1.00,1.4142,8 * 0.0,0.3743$,
$70.9080 .9 * 0.0 .0 .31291$
DO $10 \mathrm{~J}=1 \mathrm{~N}$
$A(T)=B(N . I)$
10
CONTINUE
RETURH
END

```
$ TY lOBPOUT.DAT
    * The element values for a
    * G Siection band-pass Butterworth Filter
    * Rr = 450.0000 0hms
    * RS = 150.0000 ohmes
    * Centre-freq. = 300000.0 Hz
    * Unper cut-off frea. = 30504l.7 Hz
    * Lower cut-off freq. = 29504l.7 Hz
    * Band-viath = 10000.00 Hz
C1 % + 0.1132E-09
LI t 5 0.2487E-02
R1. 5 G O.1000E+OO
C2. % 0.35.3715-07
1,2 & 0 0.7058E-05
R2. © 0 0.1000E+10
0. F, 7 0.25&5E-10
L3 7 8 0.]097E-01
R3 & G 0.1000E+00
C.t G 0 0.6E.47E-07
T& O 0. 0. 23+E-05
[12 O 0.1000E+10
B5 9 i.0 0.1965F-10
LE 10 11 0. [132E-0]
R.5 11 1 2 0.1000E+00
Cも 12 0 0.GG+7E-07
L6 12 0 0.423\pmE-05
RG 12 O 0.1000E+10
C7 12 13 0.2565E-10
L7 13 l. 0.1097E-01
F7 it 15 0.1000F+00
C8 15 0 0.3537E-07
L8 1500.7958E-05
FB I5 O 0.1000E+10
C. 15 16 0.1050E-09
L9 16 3.7 0.2681F:02.
R9 17 18 0.1000E+00
RS 18 3 0.4500E+0.3
RL. 3 O O. H5OOE10.3
```


## ARPENDIX E

## SIMULATED TEST RESULTS OF THE ARC CYCLOGRAM

## IINEARISED ARC CYCIOGRAM

The simulation technique for the arc cyclogram has been explained in Chapter 7. For the purpose of the simulated test of the arc cyclogram, the arc current of the sinusoidal waveform has been selected. Mathematically, this can be represented as:

$$
\begin{equation*}
i_{\text {arc }}=100 \sin (100 * P i * T) \tag{E1}
\end{equation*}
$$

The waveform for a period of one cycle has been shown in Figure E-1. Figure E-2 shows the arc cyclogram as explained in Chapter 7. For a positive half cycle, curve 1-2-3 is followed. For the negative half cycle, the the curve 1'$2^{\prime} 3^{\prime}$ is followed. $\mathrm{V}_{\mathrm{P}}$ is the peak voltage which is 10.3 Volts. The arc voltage waveform $v_{\text {arc }}$ has been shown in Figure E-1. The peak voltage 10.3 Volts and different gradients at the point of the voltage reversal are clearly shown.

## ARC-CYCLOGRAM WITH REIGNITION EFFECT

For the purpose of the simulated test of the arc cyclogram with reignition effect, the arc current
$i_{a r c}=100 \sin (1000 * P i * T)$
with holding the current zero near the cycle reversal has been considered.

Figure E-3 shows the current and voltage waveforms of the arc with re-ignition effect for a period of 10 cycles. At each point of the current reversal, 1, 1', 2, 2', 3, 3', 4, 4', 5, 5', 6, 6', 7, 7', 8, 8', 9, 9', 10, the effect of zero current holding has been clearly shown. The reignition voltage corresponding to the each point is evaluated as explained in Chapter 7. As such a current waveform with holding current zero near the voltage reversal has been found to be difficult. Some criterion is required for the time of the current to remain zero. For the purpose of simulated testing, a simplified criterion has been selected. This criterion is that the moment the arc voltage drops to a value equal to the reignition voltage, the current is zero and it remains zero upto the time of the reversal of the current. Figure E-4 shows the reignition voltage $V_{r}$ at each current reversal by the points 1 to 10 and $1^{\prime}$ to $9^{\prime} . V_{P}$ is the maximum voltage which is 10.3 Volts. For the positive half cycle, the curve $a-b-c$ is followed and for the negative half cycle the curve a'-b'-c' is followed.


Figure E-2 ARC CYCLOGRAM



Figure E-4 ARC-CYCLOGRAM WITH REIGNITION EFFECT



[^0]:    Figure 5.8 BAND PASS EILTER OUTPUT RESPONSES 'VX FOR THE INTERNAL AND THE HEVERSE EAULTT

[^1]:    - Figure 5.15 SIGNAL 'VE' AND 'V $\underset{\underline{Z}}{ }$ ' FOR THE INTERNAL FAULT AT VMAX

