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Robust Adversarial Attacks Detection for Deep Learning based Relative Pose Estimation for Space Rendezvous

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Abstract

Research on developing deep learning techniques for autonomous spacecraft relative navigation challenges is continuously growing in recent years. Adopting those techniques offers enhanced performance. However, such approaches also introduce heightened apprehensions regarding the trustability and security of such deep learning methods through their susceptibility to adversarial attacks. In this work, we propose a novel approach for adversarial attack detection for deep neural network-based relative pose estimation schemes based on the explainability concept. We develop for an orbital rendezvous scenario an innovative relative pose estimation technique adopting our proposed Convolutional Neural Network (CNN), which takes an image from the chaser's onboard camera and outputs accurately the target's relative position and rotation. We perturb seamlessly the input images using adversarial attacks that are generated by the Fast Gradient Sign Method (FGSM). The adversarial attack detector is then built based on a Long Short Term Memory (LSTM) network which takes the explainability measure namely SHapley Value from the CNN-based pose estimator and flags the detection of adversarial attacks when acting. Simulation results show that the proposed adversarial attack detector achieves a detection accuracy of 99.21%. Both the deep relative pose estimator and adversarial attack detector are then tested on real data captured from our laboratory-designed setup. The experimental results from our laboratory-designed setup demonstrate that the proposed adversarial attack detector achieves an average detection accuracy of 96.29%.

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Keywords: Pose Estimation; Deep Learning; Adversarial Attack ; Adversarial Attack Detection; Explainable Artificial Intelligence

1. Introduction

The growth of deep learning-based techniques has drawn 2 increasing attention in various domains of application, such 3 as image processing, speech recognition, and many other challenging Artificial Intelligence (AI) based tasks (Guo 5

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et al., 2016). Vision-based autonomous orbital space rendezvous (Wie et al., 2014), is an application for which adopting deep learning approaches to spacecraft position and attitude estimation is continuously gaining interest within the research community and the space agencies (Song et al., 2022; Kisantal 10 et al., 2020). 11

The state-of-the-art achievements in deep learning (DL) re-12 search demonstrate that the Convolutional Neural Networks 13 (CNNs) have successfully gained outstanding performance in 14 computer vision applications, such as object detection and tar-15 get localisation (Ren et al., 2017; Redmon & Farhadi, 2018; 16

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Cebollada et al., 2022). Determining the pose of a spacecraft's 17 relative state by processing input images is typically achieved 18 through the six Degree-of-Freedom (6 DOF) pose estimation 19 of the target object frame relative to the camera (onboard the 20 spacecraft) frame. These vision-based pose estimation meth-21 ods are traditionally computed by matching relative features on 22 images captured by the camera to relative locations in the tar-23 get frame. Different from the traditional approaches, the CNNs 24 can be trained to detect features from raw image data and es-25 timate the relative pose by regressing the position and attitude, 26 without the need for manual feature engineering which is often 27 required in traditional computer vision methods. The advan-28 tages of CNN-based pose estimation approaches are that they 29 can potentially lead to better performance in complex orbital 30 scenarios and more robustness to variations in lighting, view-31 point, and cluttered background. 32

Recent achievements in DL-based pose estimation demon-33 strate outstanding accuracy performance (Phisannupawong 34 et al., 2020; Oestreich et al., 2020; Rondao et al., 2022; 35 Chekakta et al., 2022). However, the vulnerability of such 36 deep learning scheme can be questionable (Chawla et al., 37 2022; Nemcovsky et al., 2022; Tian et al., 2024).Indeed, mi-38 nor changes in the spacecraft onboard camera acquired images 39 that is used by the CNN-based pose estimator can cause CNNs 40 to make wrong predictions due to their reliance on low-level af-41 fected features, such as edges and textures, and their high sen-42 sitivity to slight variations in the input space. Those changes in 43 the input images and thus on features that the CNN-based pose 44 estimation relies on can be caused by adversarial attacks (Lin 45 et al., 2020). Adversarial attacks aim to make small perturba-46 tions to the input images that are imperceptible to human vi-47 sion and can significantly affect the CNN's prediction (Grabin-48 ski et al., 2022). For real-world applications where CNNs are 49 applied to estimate the relative pose of spacecraft, applying an 50 adversarial attack to the input images can potentially make the 51 CNNs output the wrong position or attitude of the target. This 52 could seriously damage the autonomous rendezvous operation 53 system if wrong pose data are involved to generate any further 54

actions, such as guidance commands for the spacecraft to rendezvous and/or dock to the target satellite.

55

One of the significant challenges associated with deep neu-57 ral networks is that these models usually lack of transparency, 58 which means people cannot understand how the deep neural 59 networks achieve their decisions. To address this issue, eX-60 plainable AI (XAI) aims to provide an understandable explanation for the AI models' decision-making process. By apply-62 ing XAI methods to CNNs, such as Class Activation Mapping 63 (CAM) (Pope et al., 2019), Layer-wise Relevance Propagation 64 (LRP) (Nazari et al., 2022) and SHapley Additive exPlanation (SHAP) values (Lundberg & Lee, 2017), users can understand 66 how CNNs work and why models output their relative pose 67 predictions. This nice characteristic of XAI methods can po-68 tentially be adopted in detecting adversarial attacks on CNN 69 models. 70

This work aims to present an innovative demonstration of 71 the vulnerability of CNN-based spacecraft rendezvous relative 72 pose estimation scheme to digital adversarial attacks on camera 73 input images and proposes a novel method for detecting those 74 adversarial attacks when they may occur. In this paper, a vision 75 based orbital autonomous rendezvous dynamic scenario is sim-76 ulated. A CNN-based pose estimator is designed and trained to 77 estimate the relative position and attitude of the target satellite 78 involving a modified Darknet-19 (Redmon & Farhadi, 2017) as 79 a feature extractor. The Fast Gradient Sign Method (FGSM) is 80 employed to introduce small perturbation attacks to the input 81 images. Various configurations of the FGSM attack are devel-82 oped to demonstrate the impact of digital adversarial attacks 83 on the CNN-based pose estimator. An LSTM-based detector 84 exploiting the explainable Shap values of the CNN based esti-85 mator is then proposed to detect the adversarial attacks acting 86 on the input images and thus the CNN based estimator outputs. To this end, this paper makes the following contributions: 88

 Firstly, a CNN-based relative pose estimator for closerange rendezvous is introduced, which is subsequently formulated as the target DL-based navigation system against adversarial attacks. Secondly, the Fast Gradient Sign Method (FGSM) Goodfellow et al. (2014) is utilised to generate invisible perturbations in the input images, introducing a range of FGSM attack configurations to illustrate the effects of digital adversarial attacks on the CNN relative pose estimator.

Then, an LSTM-based adversarial attacks detection mechanism is proposed, leveraging the explainable (SHAP)
 value (Lundberg & Lee, 2017) from the CNN-based navigation system to identify adversarial attacks affecting the input images.

 Subsequently, the CNN-based relative pose estimator and LSTM-based adversarial attacks detection mechanism have been evaluated in both synthetic data and real-world data obtained from our laboratory to demonstrate the performance of proposed frameworks.

The paper is organised as follows: Section 2 provides an 108 overview of current DL-based spacecraft pose estimation ap-109 proaches and discusses existing methods for detecting adver-110 sarial attacks. Section 3 outlines the proposed design of the 111 CNN-based pose estimator, how to adopt FGSM attacks to the 112 pose estimator, and the design of the LSTM-based adversarial 113 attack detector. Section 4 presents the test experiments that are 114 conducted on both simulation data and real-world data obtained 115 from our laboratory. Finally, Section 5 concludes the paper and 116 discusses future work. 117

118 2. Background and Related Works

119 2.1. DL-based Spacecraft Relative Pose Estimation

Sharma et al. (2018) proposed a relative pose classification 120 network which is based on AlexNet (Krizhevsky et al., 2012) 121 architecture for non-cooperative spacecraft. In their design, the 122 convolutional layers in AlexNet are initially trained on Ima-123 geNet dataset (Deng et al., 2009) as feature extractors. The pre-124 trained feature extractors are adopted with two fully-connected 125 layers and one classification layer with training on ten sets of 126 synthetic images that were created from Tango spacecraft flown 127 in the Prisma mission (Persson et al., 2006). Their work shows 128

that the CNN-based relative pose classification outperforms the accuracy of an architecture based on classical feature detection algorithms. However, this network is designed to output a coarse pose classification and cannot meet the precision requirements for fine position and attitude estimation missions.

Yang et al. (2021) have proposed a CNN-based pose estima-134 tion method to estimate the relative position and orientation of 135 non-cooperative spacecraft. In their approach, the pre-trained 136 ResNet-50 (He et al., 2016) is adopted as the feature extrac-137 tor, and two fully-connected layers are concatenated after the 138 feature extract to output the relative position and orientation of 139 the target spacecraft, respectively. To adapt the network to esti-140 mate the relative pose of other similar spacecraft, an additional 141 output layer is concatenated with the output of position and ori-142 entation to predict the category of the target spacecraft. Dif-143 ferent from previous work introduced by Sharma et al. (2018), 144 this work can output the relative position and orientation of the 145 target spacecraft, instead of a coarse pose classification. Sim-146 ilarly, pre-trained ResNet has also been used as the backbone 147 by Proença & Gao (2020). In this work, the estimation of po-148 sition is achieved by two fully-connected layers with a simple 149 regression, and the relative error is minimised based on the loss 150 weight magnitudes. Then, the continuous attitude estimation is 151 performed via classification with soft assignment coding (Liu 152 et al., 2011). 153

Rather than estimating the relative pose of spacecraft by us-154 ing a single input frame, consecutive image inputs have been 155 considered by group previous work, named ChiNet (Rondao 156 et al., 2022). The ChiNet featured a Recurrent Convolutional 157 Neural Network (RCNN) architecture, which involves a mod-158 ified Darknet-19 (Redmon & Farhadi, 2017) as an image fea-159 ture extractor and followed by LSTM units to deal with the se-160 quences of input images. The ChiNet takes 4-channels input 161 which not only includes the RGB image but also a thermal im-162 age of the spacecraft that has been stacked to the fourth channel 163 of input. The ChiNet also proposed a multistage optimisation 164 approach to train the deep neural network to improve the per-165 formance in spacecraft relative pose estimation. 166 4

167 2.2. Explainability in CNNs

While recent approaches to DL-based spacecraft relative 168 pose estimation demonstrate outstanding performance in terms 169 of prediction accuracy, understanding how these models predict 170 relative pose is essential for providing robust solutions for fu-171 ture space rendezvous missions. As a new approach solution, 172 eXplainable AI (XAI) techniques offer the possibility to anal-173 vse gradients in DL models to indicate the significance of input 174 variables in the estimation decision-making process. 175

Lundberg & Lee (2017) proposed the SHAP values to in-176 terpret complex machine learning models. The SHAP value 177 is based on a concept from game theory called Shapley val-178 ues. These are used to fairly distribute the payoff among the 179 players of a cooperative game, where each player can have dif-180 ferent skills and contributions. Similarly, SHAP values assign 181 each feature an importance value for a particular prediction and 182 provide insights into the contribution of each feature. By ex-183 amining the SHAP values of machine learning models, we will 184 able to understand the predictions of complex machine learning 185 models. 186

Contrastive gradient-based (CG) saliency maps (Simonyan 187 et al., 2013) are visual explanation methods for deep neural 188 networks. They produce a heat map where the norm of the 189 model's gradients indicates the significance of input variables. 190 The heat map highlights the areas in the input image that would 191 change the output class if they were changed. By accessing the 192 heat map, users can identify the most relevant features for the 193 model's prediction. 194

Class Activation Mapping (CAM) (Zhou et al., 2016) gen-195 erates visual explanation maps by finding the spatial locations 196 in the input image that contribute the most to a specific pre-197 diction. The CAM is particularly helpful in image classifica-198 tion tasks through CNNs. Similarly, gradient-weighted CAM 199 (Grad-CAM) (Selvaraju et al., 2017) extends the work of CAM 200 and provides visual explanations for decisions made by a wide 201 range of CNN-based methods. Grad-CAM utilises the gradients 202 of any target concept, flowing into the final convolutional layer 203 to produce a localisation map that highlights the important re-204

gions in the input image for predicting the concept. These XAI 205 methods interpret the CNNs, making people understand how 206 and why CNNs make certain predictions. However, since then, 207 there has been no specific analysis on interpreting the DL-based 208 spacecraft relative pose estimation to improve their explainability. 210

2.3. Adversarial Attacks 211

Adversarial attacks for CNNs aim to make small perturba-212 tions on the original input images where original and perturbed 213 images look similar in human vision but can significantly im-214 pact the CNNs' predictions. However, very limited research 215 works are investigating how adversarial attacks can impact DL-216 based pose estimation systems. Chawla et al. (2022) demon-217 strate the effect of different types of adversarial attacks on the 218 predictions of the DL-based pose estimation system. Their 219 work shows that adversarial attacks can significantly impact 220 monocular pose estimation networks, leading to increased tra-221 jectory drift and altered geometry. Similarly, Nemcovsky et al. 222 (2022) illustrate that the physical passive path adversarial at-223 tacks can seriously increase the error margin of a visual odom-224 etry model which is used in autonomous navigation systems 225 leading onto potential collisions. 226

The impacts of adversarial attacks have garnered significant 227 attention in the DL-based autonomous systems. Ilahi et al. 228 (2021) provide an extensive overview of recent methodologies 229 for adversarial attacks on Deep Reinforcement Learning mod-230 els applied to autonomous systems, as well as the existing tech-231 niques for mitigating these attacks. Wang & Aouf (2024) ex-232 amine the effects of perceptual perturbations on vision-based 233 autonomous driving systems and propose an optimised pol-234 icy to counter adversarial attacks on observation perturbations. 235 Tian et al. (2024) explore multi-label adversarial example at-236 tacks targeting multi-label False Data Injection Attacks for lo-237 cational detectors, highlighting significant security vulnerabili-238 ties in DL-based smart grid systems. 239

To protect the DL-based system from adversarial attacks, ²⁴⁰ Liu et al. (2020) proposed a detection method based on the robustness of the classification results. Their results show that ²⁴²

the detector performs well against gradient-based adversarial 243 attacks. Our group work, Hickling et al. (2023) proposed a 244 CNN-based adversarial attack detector and an LSTM-based 245 adversarial attack detector for Deep Reinforcement Learning 246 (DRL) based Uncrewed Aerial Vehicle guidance. The simula-247 tion results show that the LSTM-based adversarial attack detec-248 tor leads to 90% detection accuracy on the DRL model. It also 249 suggests that the LSTM-based detector performs much more 250 accurately and quicker than the CNN-based adversarial attack 251 detector. Indeed, the LSTM-based detector is demonstrated to 252 meet the real-time requirement in DRL based guidance. 253

To the best of our knowledge, as of yet, there is no literature 254 looking at the impact of adversarial attacks in spacecraft relative 255 pose estimation and how to detect those adversarial attacks in 256 DL-based spacecraft relative pose estimation systems and this 257 work first time proposes this. Our objective is to ultimately 258 create an adversarial attack detector for the space navigation 250 system, which employs SHAP values explainability mechanism 260 to detect and flag potential adversarial attacks. 261

262 **3. Methodology**

In this section, a CNN-based spacecraft relative pose esti-263 mator is newly designed with the aim of providing a reliable 264 estimated position and attitude of the target spacecraft in as 265 rendezvous scenario. Then, the FGSM attacks are adopted on 266 the spacecraft onboard camera resulting in an adversarial im-267 age to evaluate the impacts on the proposed deep pose estima-268 tor. Next, SHAP values are introduced to generate XAI sig-269 natures for both adversarial and normal input images. Finally, 270 an LSTM-based adversarial detector is proposed and trained, 271 which learns normal and adversarial SHAP values to detect the 272 adversarial attacks on the spacecraft relative pose estimator. 273

3.1. CNN-based Spacecraft Relative Pose estimator 3.1.1. Overall architecture design

Similar to most DL-based spacecraft relative pose estimation
algorithms, CNN is applied to extract features in the proposed
pose estimator. The overall design of the pose estimator follows
the design methodology in ChiNet (Rondao et al., 2022). The

Darknet-19 (Redmon & Farhadi, 2017) is originally trained in ImageNet (Deng et al., 2009) dataset which has an input size of $244 \times 244 \times 3$. In our design, input images of the pose estimator 282 have a larger size than ImageNet images. Therefore, the first 283 convolutional layer in Darknet-19 is configured with a kernel 284 size of 7×7 . Following the approach of Darknet-53 (Redmon 285 & Farhadi, 2018), the maxpooling layers in the Darknet-19 are 286 replaced by 3×3 convolution operation with a stride of 2. Sim-287 ilarly, as the Darknet-53 approaches, the residual connections 288 are also adopted to the proposed pose estimator. Batch Normal-289 isation (Ioffe & Szegedy, 2015) layers are applied after each convolutional layer. 291

Our deep spacecraft relative pose estimator aims to output 292 the relative position and attitude of the target directly. There-293 fore, two separate FC layers are applied. The first FC layer involves 3 output nodes to output the relative position in (x, y, z)295 and the second FC layer adopts a 6-dimensional (6-D) vector to 296 represent the relative attitude of the target spacecraft. Finally, 297 two FC layers are concatenated together to output the relative 298 6-DOF pose. In the second FC layer, 6-D vectors are applied to represent the relative attitude of the target spacecraft, instead of 300 using quaternion representation. The reason is that the relative 301 pose estimator is designed as a regression problem where the 302 output has to be continuous. However, the normal attitude rep-303 resentation of quaternion is discontinuous, due to its antipodal 304 ambiguity, i.e. -q = q. Therefore, the proposed pose estimator 305 applies the 6-D vector formulated by Zhou et al. (2019), which 306 mapped the 3-dimensional rotations into a 6-D continuous rota-307 tion. The overall design of the spacecraft relative pose estimator 308 is presented in Fig. 1. 309

3.1.2. Synthetic data generation

To train and test the spacecraft relative pose estimator, synthetic datasets are generated in Blender, which is an opensource 3D modelling software. The spacecraft target model used in the synthetic dataset generation is the Jason-1 satellite, which was downloaded from the NASA 3D Resources website (Jason-1 3D Model). Dynamic simulation of the rendezvous is developed to generate the synthetic dataset in which



Fig. 1. The overall architecture of the proposed spacecraft relative pose estimator. The blue blocks represent the convolutional layers, which are formatted as (*layer size, kernel size, stride*)). Each convolutional layer is followed by a batch normalisation layer and LeakyReLu activation. The yellow block indicates the Global Average Pooling (GAP) layer that downsamples the exacted features to a fixed 1D vector of 1000 units. The green blocks represent FC layers that will output the estimated relative position and attitude, respectively.

the camera onboard the chaser spacecraft starts at 60 metres 318 away in z - axis from the target and end at 10 metres away 319 from the target in z - axis, i.e. (0,0,10). Random rotation of 320 the camera and target is considered in the synthetic data gen-321 eration. Many trajectory sequences are generated and each se-322 quence contains 2,500 RGB images. Each image has a size 323 of 744×480 . To prevent overfitting in the deep relative pose 324 estimator network, random rotation of the target spacecraft is 325 applied to the model, and the camera is initialised at various 326 positions in the synthetic data generation. Table 1 illustrates 327 the synthetic dataset generated for training and validating the 328 deep pose estimator. 329

Ta	b	le	1.	Examp	le of	synt	hetic	data	generated	from	B	lend	lei
----	---	----	----	-------	-------	------	-------	------	-----------	------	---	------	-----

Sequence ID	Start Position	Target Rotation
0	(0,0,60)	0
1	(-15,-25,60)	0
2	(-15,25,60)	0
3	(15,25,60)	0
4	(15,-25,60)	0
5	(-15,-10,60)	± 10 deg
6	(-15,10,60)	± 10 deg
7	(15,10,60)	± 10 deg
8	(15,-10,60)	± 10 deg
9	(-15,-10,60)	± 10 deg
10	(-15,10,60)	± 10 deg
11	(15,10,60)	± 10 deg
12	(15,-10,60)	± 10 deg

3.1.3. Loss Function

Training the spacecraft relative pose estimator can be for-331 mulated as a regression problem, where the total loss function 332 combines the loss in position and loss in attitude. These are 333 computed by Eq. (1) and Eq. (2), respectively, which were orig-334 inally proposed by Kendall et al. (2018). Followed by Rondao 335 et al. (2022), a trainable weight is attributed to each loss, which 336 corresponds to the task-specific variance of the Gaussian distri-337 bution. The total loss is then formulated in Eq. (3). 338

$$L_p = \sum_{i=0}^{B} (||p_{pred}^i - p_{gt}^i||)$$
(1)

330

344

$$L_r = \sum_{i=0}^{B} (||r_{pred}^i - r_{gt}^i||)$$
(2)

$$L_{total} = exp(-2\sigma_p)L_p + exp(-2\sigma_r)L_r + 2(\sigma_p + \sigma_r)$$
(3)

where the p_{pred} and r_{pred} indicate the predicted position and attitude, and p_{gt} and r_{gt} indicate the ground truths position and attitude, respectively. *B* is the batch size and $\|\cdot\|$ donates the L_2 norm. σ_p and σ_r represent the learnable weights for position and attitude, respectively.

3.2. Adversarial Attacks

In this work, the adversarial examples are generated by ³⁴⁵ FGSM attacks (Goodfellow et al., 2014). The FGSM attacks ³⁴⁶ aim to add small perturbations to the input images which will ³⁴⁷



Fig. 2. An example of applying FGSM attacks to the input image. (a) the original input image. (b) perturbation patch with $\epsilon = 0.05$. (c) resultant adversarial image.

maximise the network's loss. The efficacy of adversarial at-348 tacks, including the FGSM attacks used in this work, can be 349 influenced by the backbone neural network employed in per-350 ception systems. Different neural network architectures may 351 exhibit varying levels of robustness and vulnerability to spe-352 cific types of adversarial attacks. Therefore, the effectiveness 353 of these adversarial patches is inherently related to the specific 354 CNN architectures employed. The equation in Eq. (4) describes 355 how to generate an adversarial example for a given input image 356 x by FGSM attack. 357

$$x' = x + \epsilon \times sign(\nabla_x L(\theta, x, y))$$
(4)

where ϵ is a value of the perturbation effect which describes how strong the attack is. *L* is the loss of the input *x* with ground truth of *y*. The (∇_x calculates the loss gradient, *L* for input image *x* with relative ground truth *y*, and θ indicates the trained network's parameters. Depending on the quality of input images and the attack strength, the result of the FGSM attack can be modified by changing the ϵ value.

In real implementation, the ϵ needs to be small enough to 365 ensure the perturbations on the input image are seamless and 366 cannot be visible by human vision but still significantly change 367 the deep model's predictions. The ϵ value should be in the range 368 of (0,1), where a value of 0 means the adversarial image will 369 be the same as the input image without any perturbation and 370 a value of 1 means the adversarial image will be perturbed as 371 significant distorted image to human vision. Fig. 2 illustrates 372 an example of applying FGSM attacks to input images of the 373 spacecraft relative deep pose estimator. 374

3.3. Explanability and Adversarial Attacks Detector

3.3.1. Explanability via DeepSHAP

The black-box nature of deep neural networks makes users can only observe the prediction of these models, but do not 378 know the reasons for getting correct or wrong predictions. XAI 379 techniques are developed to interpret the DL models. When the 380 model's prediction is changed, the XAI will generate relative 381 explanations to explain why the model is getting the prediction. In this work, we proposed a novel approach that adopts XAI 383 techniques by applying the change in SHAP values of the input 384 images as a measure to determine whether an adversarial attack 385 happens on input images. 386

Originally, SHAP is proposed based on the idea of Shap-387 ley values, which are designed to assign a credit to every in-388 put feature for a given prediction. Generating SHAP values for DNNs can be computationally expensive, as the DNNs nor-390 mally contain a huge amount of features. Thanks to the work of 391 DeepLIFT (Shrikumar et al., 2017), Shapley values for DNNs 392 can be estimated by linearising the non-linear components of a 393 neural network, a method referred to as DeepSHAP (Lundberg & Lee, 2017). This is achieved by utilising a reference input 395 distribution, which can be linearly approximated, to estimate 396 the expected value for the entire model. 397

However, directly generating SHAP values for the spacecraft relative pose estimator still requires a large amount of computational resources. The pose estimator is based on CNNs with image inputs that contain thousands of pixels. Using DeepSHAP for image input requires generating Shapley values for each single pixel for every output neuron. Therefore, in this work, we

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consider computing the SHAP values for the subsampling layer 404 in the pose estimator, instead of computing them for the input 405 image. As demonstrated previously, the spacecraft relative pose 406 estimator contains a GAP layer that downsamples feature maps 407 from the prior convolutional layer to 1000 samples. For exam-408 ple, computing SHAP values for a 744 × 480 RGB image needs 409 to compute 1,071,360 pixels, instead, the GAP layer in the pose 410 estimator only employs 1000 neurons. As a result, SHAP val-411 ues are generated for the outputs of the GAP layer that only 412 need to compute 1000 features. This saving in the computation 413 makes the generation of SHAP values for the deep pose estima-414 tor could potentially meet the implementation time constraints. 415

416 3.3.2. Adversarial attack detector

To detect any incoming adversarial attacks on the spacecraft 417 deep relative pose estimator trhough the onboard camera, an 418 LSTM-based adversarial attacks detector is proposed. The de-419 tector aims to monitor the SHAP values generated from the out-420 put of the GAP layer and detect any slight anomaly changes that 421 could result based an adversarial attack. The LSTM is a type 122 of Recurrent Neural Networks (RNNs) that is widely used in 423 learning from time-series data, such as speech recognition (Yu 424 et al., 2019). The LSTM architecture was originally proposed to 425 address the long-term dependency issue in conventional RNNs. 426 It can enable the propagation and representation of information 427 over a sequence without causing useful information from dis-428 tant past time steps to be ignored. 429

In our approach, the SHAP values are generated based on 430 the prediction of each output neuron in the proposed deep pose 431 estimator. Different from applying adversarial attacks on a clas-432 sification CNN that only change the output label, when an at-433 tack occurs on the deep pose estimator, it could affect all output 434 neurons to estimate for wrong position and attitude. Therefore, 435 it can be assumed that there might exist a certain level of depen-436 dencies among those output neurons. From this point of view, 437 building an LSTM-based adversarial attack detector can poten-438 tially achieve high detection accuracy. 439

Fig. 3 introduces the architecture of the proposed adversarial attack detector. The detector takes the SHAP values that are



Fig. 3. Proposed adversarial attack detector. The yellow block indicates the LSTM layer which has an input shape of (9,1000) and an output space of 100. ReLu is applied as the activation function for the LSTM layer. The blue blocks are FC layers in the format of (*units*, *activation*). The green block indicates the output layer of the adversarial detector, which is also formed from the FC layer and outputs a Boolean to detect adversarial attacks.

computed from the GAP layer of the deep pose estimator. As there are nine output neurons in the proposed deep pose estimator, the shape of the SHAP values is (9, 1000). To input SHAP values to the detector, the SHAP values are formatted as a sequence data with a length of 9. The detector outputs a Boolean, True/False, to indicate the result of detecting adversarial attacks. 448

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4. Experimental Results

To validate our adversarial detection approach, two experi-450 ments are performed. The first experiment is built on the sim-451 ulation environment with synthetic data as mentioned in Sec-452 tion 3. The second experiment is built on our lab environment 453 to testing our approach with real data. For both sets of exper-454 iments, the spacecraft deep relative pose estimator and the ad-455 versarial attack detector are tested for their relevant accuracy, 456 and then the two systems are integrated to test the overall suc-457 cessful detection rate of adversarial attacks. 458

4.1. Results on Synthetic Data

4.1.1. Accuracy of the Spacecraft Deep Relative Pose Estimator 460

To train the deep relative pose estimator, image data are collected from the Blender 3D model. There are 13 sequences of images generated from Blender with the relevant trajectories that are mentioned in Table 1. By following the trajectories in Table 1, 2,500 images are generated for each trajectory, resulting in a dataset with 32,500 images for training and testing in
total. Fig. 4 shows two examples of synthetic data generation
in Blender.



Fig. 4. Examples of synthetic data generated from Blender. (a) image captured at a camera position of (0, 0, 60m). (b) image captured at a camera position of (0, 0, 10m). Random rotation is applied to the target spacecraft.

The synthetic dataset is split by a train-test ratio of 0.8, i.e. 470 80% of images in the dataset are used for training the deep rel-471 ative pose estimator, and 20% of images are used to test the 472 model's accuracy. Each image is associated with a ground truth 473 label in the format of $(x, y, z, w, x_i, y_i, z_k)$. The first three ele-474 ments in the ground truth label represent the relative position of 475 the chaser onboard camera to the target and the rest 4 elements 476 represent the target attitude in quaternion representations in the 477 chaser camera frame. The deep relative pose estimator outputs 478 the attitude in a 6-D vector. Therefore, to calculate the loss 479 in attitude, the quaternion representations are converted to the 480 6-D vector representation by following the approach in (Zhou 481 et al., 2019). A dropout rate of 0.2 is applied to the GAP layer in 482 the training process. Multiple data augmentation techniques are 483 considered in training the deep relative pose estimator, includ-484 ing Gaussian blur, Gaussian noise, image compression, random 485 brightness and so on. These techniques help to prevent the 486 model from overfitting the training dataset. The deep relative 487 pose estimator is trained by stochastic gradient descent with an 488 Adam optimiser. The triangular2 (Smith, 2017) policy is ap-489 plied for cycling learning rate with base learning of 2.5e-5. 490

After training the deep relative pose estimator for 50 epochs with the training batch size of 32, The model's average prediction accuracy for both training and test datasets is reported in Fig. 5. In this experiment, the position error is measured by Eq. (5) and the attitude error is measured by Eq. (6).

$$p_{err} := \|p_{pred} - p_{gt}\| \tag{5}$$

$$\dot{r}_{err} := 2 \arccos(q_{pred}^{-1} \otimes q_{gt}) \tag{6}$$

where p_{pred} and p_{gt} represent the prediction of position and the ground truth of position magnitude. The q_{pred} and q_{gt} indicate the prediction of attitude and the ground truth of attitude in quaternion representation. The \otimes denotes the quaternion multiplication and $\|\cdot\|$ denotes the L_2 norm.



Fig. 5. The prediction accuracy of the proposed pose estimator on training and test dataset after 50 epochs. The blue bar presents the average error on training data and the orange bar represents the average error on test data

The proposed spacecraft relative pose estimator achieves an 502 accuracy of around 0.49m in position error and 0.68 deg in attitude error on the test dataset. Table 2 reports a comparison 504 between the proposed deep relative pose estimator and state-505 of-the-art performance of other DL-based space relative pose 506 estimation approaches based on their datasets. The comparison 507 here aims to show that the proposed spacecraft deep relative 508 pose estimator can achieve relatively good performance on the 509 synthetic data and can be applied as a baseline model to imple-510 ment the adversarial attack algorithm on and test the adversarial 511 attack detector. The comparison is not meant to be a quantita-512 tive benchmark evaluation of our approach relative to existing 513 performing approaches. 514

4.1.2. FGSM Adversarial Attacks

As discussed in Section 3, the perturbation made by FGSM $_{516}$ attacks can be adjusted by changing the ϵ value. To investi- $_{517}$

495

Model Dataset Position Error (m) Attitude Error (deg) SPEED (Kelvins - ESA's Advanced Concepts Competition Website) Proenca & Gao (2020) 0.56 8.0 Rondao et al. (2022) Synthetic 1.73 6.62 Yang et al. (2021) Synthetic [0.052, 0.039, 0.077] [0.213, 0.233, 0.097] Ours Synthetic 0.49 0.68

Table 2. Comparison with other approaches in DL-based space relative pose estimation

gate the impact of adversarial attacks on DL-based space rela-518 tive pose estimation, different ϵ values are selected to generate 519 adversarial onboard camera image input to the proposed deep 520 relative pose estimator. Typically, the ϵ applied in this experi-521 ment are 1, 0.5, 0.3, 0.1, 0.05 and 0.01. The larger value of ϵ is, 522 the more perturbations are made to images. The FSGM attack 523 is applied to all synthetic test data, where all images in the test 524 data. Then, the perturbed images are fed to the deep relative 525 pose estimator for testing the impact of the FGSM attack. The 526 average prediction relative pose errors of applying different ϵ 527 values are reported in Fig. 6.



Fig. 6. Comparison of the prediction error of pose estimator under FGSM attack on test data with various ϵ values. The blue bar indicates the average position error and the red bar indicates the average attitude error on test data. The error magnitude for the position is metres and the error magnitude in rotation is measured by degrees.

528

We can see that as the ϵ value increases, the deep model's 529 prediction error becomes larger. The attitude error is quite sta-530 ble on $\epsilon = 0.1, 0.05$ and 0.01, but has a dramatic increase if the 531 $\epsilon > 0.3$. 532

To assess well how the adversarial attack can impact the 533 DL-based navigation system in a space rendezvous scenario, 534 a simple guidance scheme is implemented with the proposed 535 deep relative pose estimator. The guidance scheme takes the 536

estimated relative pose from the proposed deep relative pose 537 estimator and then provides relative control actions to move 538 the camera (spacecraft) to the target position. In the guidance 539 scheme, the camera has an initial position of (0, 0, 60) and a target position of (0, 0, 10) with $\pm 1m$ tolerance. The guidance 541 scheme updates the camera position with a maximum velocity 542 of 1m/s, as described in Eq. (7) and Eq. (8) 543

$$p_{new} = \begin{cases} p_{est} - 1 & \text{if } diff \ge 1\\ p_{est} - diff & \text{otherwise} \end{cases}$$
(7)

$$diff = p_{est} - p_{tar} \tag{8}$$

544

547

where p_{new} , p_{est} , p_{tar} present the updated position, estimated 646 position and target position of the camera, respectively. The 546 test system is implemented as shown in Fig. 7.



Fig. 7. Test system for proposed pose estimator on Blender in simulated space rendezvous scenario.

In this experiment, the test system is continuously attacked 548 by FGSM on image data with various acquired camera frames. 549 The success attack is defined as the camera (spacecraft) missing 550 the target position while the *failure* attack means that the cam-551 era (spacecraft) can still reach the target position under continu-552 ous FGSM attack. Experimental results are reported in Table 3 553 - 7. 554

Table 3. FGSM attacks on the simulated space rendezvous scenario with ϵ =0.5

	$\epsilon = 0.5$						
Continuously		5	10	15	20		
Attacked Frame		5	10	15	20		
	60	failure	failure	failure	Success		
Attack start point	50	failure	failure	failure	Success		
Attack start point	40	failure	failure	Success	Success		
(m)	30	failure	failure	Success	Success		
(111)	20	failure	Success	Success	Success Success Success Success Success Success		
	10	Success	Success	Success	Success		

Table 4. FGSM attacks on the simulated space rendezvous scenario with ϵ =0.3

	$\epsilon = 0.3$						
Continuously		5	10	15	20		
Attacked Frame		5	10	15	20		
	60	failure	failure	failure	Success		
Attack start point	50	failure	failure	Success	Success		
Attack start point	40	failure	failure	Success	Success		
(m)	30	failure	failure	Success	Success		
(111)	20	failure	Success	Success	Success		
	10	failure	Success	Success	Success		

Table 5. FGSM attacks on the simulated space rendezvous scenario with ϵ =0.1

	$\epsilon = 0.1$					
Continuously Attacked Frame		5	10	15	20	
	60	failure	failure	failure	failure	
Attack start point	50	failure	failure	failure	failure	
Attack start point	40	failure	failure	failure	Success	
(m)	30	failure	failure	failure	failure	
(111)	20	failure	failure	failure	failure	
	10	failure	failure	failure	Success	

Table 6. FGSM attacks on the simulated space rendezvous scenario with ϵ =0.05

$\epsilon = 0.05$						
Continuously		5	10	15	20	
Attacked Frame		5	10	15	20	
	60	failure	failure	failure	failure	
Attack start point	50	failure	failure	failure	Success	
Attack start point	40	failure	failure	failure	Success	
(m)	30	failure	failure	Success	Success	
(111)	20	failure	Success	Success	Success	
	10	failure	Success	Success	Success	

Table 7. FGSM attacks on the simulated space rendezvous scenario with ϵ =0.01

		$\epsilon = 0.0$	1		
Continuously		5	10	15	20
Attacked Frame		5	10	10	20
	60	failure	failure	failure	failure
Attack start point	50	failure	failure	failure	failure
Attack start point	40	failure	failure	failure	failure
(m)	30	failure	failure	failure	failure
(111)	20	failure	failure	failure	failure
	10	failure	failure	failure	failure

From Table 3 - 7, we can clearly see that the adversarial attack can result in a significant impact on the guidance scheme if DNN-based relative navigator is attacked, typically when the 557 distance between the camera and the target is less than 30m. 558 In most cases, continuously attacking the deep model for more 559 than 15 frames after the camera approaches less than 30m to the 560 target, the camera (spacecraft) will fail to reach the target posi-561 tion. In a real space rendezvous mission where a chaser relies 562 on a DL-based relative pose estimation system, an adversarial 563 attack has the potential to cause the chaser to fail in approaching the target position, resulting in mission failure. Therefore, detecting adversarial attacks on DL-based pose estimators be-566 comes critical. 567

4.1.3. LSTM-based Adversarial Attack Detector

The proposed adversarial attack detector is designed based 569 on the LSTM architecture. It aims to detect the change in SHAP 570 values when an adversarial attack occurs on the input image. As 571 mentioned in Section 3, the SHAP values are computed at the 572 output of the GAP layer in the proposed deep relative pose es-573 timator. The GAP layer contains 1000 neurons, therefore, 1000 574 values are calculated for each output neuron, resulting 9×1000 575 output SHAP values. 576

In our approach, the SHAP values of the GAP layer are cal-577 culated by DeepSHAP (Lundberg & Lee, 2017) algorithm. The 578 DeepSHAP algorithm computes SHAP values for inputs by in-579 tegrating over background samples. It then estimates approx-580 imate SHAP values in a manner that sums up the difference between the expected deep model's output on the background 582 samples and the current model's output. In this work, 1000 583 images are randomly selected from the training dataset to com-584 pute the downsampled features at the GAP layer. These samples serve as the background samples for the DeepSHAP ex-586 plainer. To train the adversarial attack detector, we generated 587 15,000 sets of SHAP values for normal samples and an addi-588 tional 15,000 sets of SHAP values for adversarial samples. The normal samples consist of the entire test dataset, which is used 590 for testing the deep pose estimator, along with a random selec-591 tion of images from the training dataset. This random selection 592

was made to reach a total of 15,000 samples, thereby bridging 593 the gap between this number and the number of images in the 594 test dataset by the deep relative pose estimator. The adversar-595 ial instances are crafted by launching attacks on the DRLs at 596 arbitrary time steps with random ϵ values: 0.5, 0.3, 0.1, 0.05, 597 and 0.01. Subsequently, 3,000 perturbed images are randomly 598 selected from each ϵ value for calculating the corresponding 599 SHAP values. 600

The SHAP values for both normal and adversarial samples 601 are split into a training and testing set using a 0.8 train-test ra-602 tio, resulting in 24,000 samples for training and 6,000 samples 603 for testing. The adversarial attack detector is trained using the 604 Stochastic Gradient Descent (SGD) method with the Adadetal 605 optimiser for 1,000 epochs. After training the adversarial at-808 tack detector, it achieved a training accuracy of 99.98% and a 607 test accuracy of 99.90% on the test dataset. In this case, the 608 detection accuracy is calculated by Eq. (9) 609

$$accuracy = \frac{successful \ Detection}{Total \ No. \ of \ Frames} \times 100\% \tag{9}$$

where the successful Detection is defined by that the input 610 frames with the adversarial attack are detector as True and 611 frames without adversarial attack are detector as False. The 612 experimental results show that the proposed detector can suc-613 cessfully detect adversarial attacks on the DL-based relative 614 pose estimator with high accuracy. The adversarial attack de-615 tector is integrated with the deep relative pose estimator and the 616 DeepSHAP explainer to enhance accuracy in space rendezvous 617 scenarios. The overall system is presented in Fig. 8. 618

The adversarial attack detector is then tested with three tra-619 jectories. In each trajectory, the camera (spacecraft) starts 60 620 meters away from the target, positioned at various points in the 621 x and y directions within the range of $[\pm 25, \pm 15]$. The camera 622 is oriented directly toward the target, with an attitude repre-623 sented as quaternion (1, 0, 0, 0). The end position is (0, 0, 10). 624 The camera (spacecraft) moves linearly at a rate of 0.25 meters 625 per frame along the z-axis. It follows a projectile trajectory in 626 the x and y directions, resulting in a total of 2,500 frames for 627 each trajectory. The FGSM attack is applied to test trajectories 628

Table 8. The average accuracy of the adversarial attack detector in test trajectories with various ϵ values.

ϵ	Trajectory	Detection Accuracy
	1	100%
0.5	2	100%
	3	100%
	1	100%
0.3	2	100%
	3	99.98%
	1	99.96%
0.1	2	99.98%
	3	99.96%
	1	100%
0.05	2	99.98%
	3	99.98%
	1	97.06%
0.01	2	96.94%
	3	99.02%
Average		99.21%

with an attack probability of 0.2. Once FGSM is initiated, at-620 tacks continue for the subsequent 5 frames. The results of the 630 proposed adversarial attack detector are presented in Table 8. 631 From the test results, the proposed adversarial attack detector 632 successfully detects all incoming FGSM attacks when the ϵ = 633 0.5. As the ϵ value goes small, i.e. fewer perturbations are made 634 to input images, the detection accuracy has slightly dropped. 635 For these three test trajectories, the proposed adversarial attack 636 detector achieves a detection accuracy of 99.21% on average. 637

638

4.2. Experimental Results on Real Data

In previous experiments, both the proposed deep relative 639 pose estimator and the adversarial attack detector exhibited 640 high accuracy on synthetic data. To further evaluate the per-641 formance of both systems, we tested them with real-world im-642 ages obtained from the Autonomous Systems and Machine In-643 telligence Laboratory (ASMI Lab) at City, University of Lon-644 don. These data include sensor noise, camera calibration noise, 645 ground truth measurement noise, and different lighting condi-646 tions that are not present in the training synthetic images. 647

4.2.1. Accuracy of the Spacecraft Dee Relative Pose Estimator 648

At the ASMI Lab, a scaled mock-up model of the Jason-1 spacecraft is constructed. This mock-up model is 1/9 the size of the actual Jason-1 spacecraft. The vision sensor ap-



Fig. 8. The experimental system includes the integration of an adversarial attack detector with the relative pose estimator and SHAP values generator.

plied for real data acquisition is the ZED 2 camera, which out-652 puts images with a resolution of 1920×1080. The deep relative 653 pose estimator is retrained on new synthetic data, referred to 654 as the Synthetic-Lab Dataset, with an input RGB image size of 655 480×270 to match the aspect ratio of the camera used in the 656 ASMI Lab. As before, the Synthetic-Lab Dataset is generated 657 using Blender, where the target was replaced with a 3-D model 658 of the ASMI Lab mock-up Jason-1. To simulate the space ren-659 dezvous scenario over a distance range from 60m to 10m, the 660 3-D model is scaled up by a factor of 9 in Blender data genera-661 tion. An example of the re-training images is shown in Fig. 9 662



Fig. 9. Am example of images generated from Blender for training the pose estimator.

663 Similar to the previous synthetic data experiment, multiple 664 trajectories are generated to collect images from the Blender, resulting in a total of 32,500 images on Synthetic-Lab Dataset for training and testing. The hyperparameter settings for training are the same as the settings applied in previous synthetic data experiment, including the learning rate, optimiser, batch size, and data augmentation methods. The pose estimator was trained for 100 epochs with a train/test split of 0.8.

There are three sets of images captured from the ASMI 671 Lab, referred to as the ASMI Dataset, with each set contain-672 ing a total of 750 images. To acquire images for the ASMI 673 Dataset, the camera movement is controlled by the Rethink 674 Robotics Sawyer (Sawyer Rethink Robotics) moving along the 675 z-axis, and the ground truths relative poses of the images in 676 ASMI Dataset are recorded by the OptiTrack Motion Capture 677 Systems (OptiTrack). The OptiTrack Motion Capture System 678 records the position and attitude of the ASMI Lab mock-up 679 Jason-1 and the ZED camera at a frame rate of 120 frames 680 per second and assigns a timestamp to each frame. Images are 681 acquired by the ZED camera at a resolution of 1920×1080 682 and a frame rate of 30 frames per second, with relevant times-683 tamps. The ground truth pose for each frame acquired by the 684 ZED camera are assigned by matching the corresponding times-685 tamps from the OptiTrack Motion Capture System. Then, the 686

(13)

relative ground truth position is calculated by the difference between the actual positions of the ZED camera and the ASMI
Lab mock-up Jason-1, as shown in Eq. (10),

$$Pos_{lab} = Pos_{camera} - Pos_{target}$$
(10)

where *Pos_{lab}* donates relative ground truth position in ASMI Dataset. The *Pos_{camera}* and *Pos_{target}* donate the actual position of the ZED camera and ASMI Lab mock-up Jason-1 recorded by OptiTrack Motion Capture System, respectively.

⁶⁹⁴ Due to different camera intrinsic matrices applied between ⁶⁹⁵ the Synthetic-Lab Dataset and ASMI Dataset, to represent the ⁶⁹⁶ relative position in the trained model, the position ground truths ⁶⁹⁷ of the ASMI Dataset are collaborated with the camera view by ⁶⁹⁸ the following processing:

$$K_{Blender} = \begin{bmatrix} 640 & 0 & 240 \\ 0 & 360 & 135 \\ 0 & 0 & 1 \end{bmatrix}$$
(11)

699

$$K_{zed} = \begin{bmatrix} 1400.41 & 0 & 956.29 \\ 0 & 1400.41 & 557.258 \\ 0 & 0 & 1 \end{bmatrix}$$
(12)

700 701

$$Pos_{real} = 1400.41 \times \frac{240}{956.29} \times \frac{Target_{lab}}{Target_{real}} \times \frac{1}{640} \times Pos_{lab}$$
(14)

 $Target_{real} = 9 \times Target_{lab}$

where $K_{Blender}$ and K_{zed} represent the camera intrinsic matrices 702 for the camera used in Synthetic-Lab Dataset collection and the 703 ZED camera that is used to acquire images in the ASMI Lab, 704 respectively. Target_{real} and Target_{lab} indicate the target sizes 705 in the Blender 3-D model and the actual size in the ASMI Lab. 706 Posreal and Poslab denote the relative position of the target in 707 the pose estimator and the ground truth position in the ASMI 708 Lab, respectively. Table 9 illustrates the range of relative posi-709 tions in the ASMI Dataset and representative relative positions 710 in trained pose estimator. Furthermore, all images in the ASMI 711 Dataset are segmented with a black background and resized to 712 480×270 to fit the input image size of the trained pose estima-713 tor. An example of images captured in ASMI Lab is shown in 714 Fig. 10. 715

Once the deep relative pose estimation model is trained, it is initially tested on the test set of Synthetic-Lab Dataset, followed then by testing its prediction accuracy on real world data

Table 9. Camera moving range on ASMI Dataset and its representative range on trained pose estimator. The camera is moving along the z - axis. The representative range is calculated by Eq. (14).

trajectory ID	ASMI Lab Range (z-axis)	Representative Range(m)
ASMI-1	3.122 - 2.569	51.180 - 42.11
ASMI-2	2.296 - 1.748	37.64 - 28.66
ASMI-3	1.564 - 1.015	25.64 - 16.64



Fig. 10. Examples of images captured in ASMI Dataset. (a) Original image captured in ASMI Lab (b) Segmented image with back background.

captured from the ASMI Lab, i.e. ASMI Dataset. The predic-719 tion accuracy of the deep relative pose estimator is reported in 720 Fig. 11. Similar to the previous synthetic testing, position error 721 and attitude error are calculated by Eq. (5) and Eq. (6), respec-722 tively. Compared with the prediction accuracy on the Synthetic-723 Lab Dataset, the position error of the ASMI Dataset is slightly 724 higher. This could be attributed to variations in the illumina-725 tion conditions compared to the Synthetic-Lab Dataset, as well 726 as factors such as ground truth measurement noise and camera 727 calibration noise. On the other hand, the predicted attitude er-728 ror in the ASMI Dataset is much smaller than the synthetic data. 729 One possible reason could be that the target remains stable at a 730 fixed position with rotation effects during the images capture. 731

4.2.2. FGSM Attacks on ASMI Dataset

To evaluate how the pose estimator can be impacted by ad-733 versarial attacks on real data, the FGMS attack is then applied 734 to the ASMI Dataset. In this case, the FGSM is configured as 735 the same ϵ as previously applied in synthetic, including 1, 0.5, 736 0.3, 0.1, 0.05 and 0.01. In this experiment, all images are per-737 turbed by the FGSM attack. The model's average prediction 738 error under FGSM attacks with various ϵ values on the ASMI 739 Dataset are illustrated in Fig. 12. 740

732

As shown in Fig. 12, FGSM has a significant impact on position estimation but only slight impacts on attitude estimation. 742 In comparison to the previous experiment with synthetic data, 743



Fig. 11. The prediction accuracy of the proposed pose estimator on Synthetic-Lab Dataset and ASMI Dataset after 100 epochs. The blue bar presents the average error on training data and the orange bar represents the average error on test data on Synthetic-Lab Dataset. The green bar indicates the average error on the ASMI Dataset.



Fig. 12. Comparison of the prediction error under FGSM attack on ASMI Dataset with various ϵ values. The blue bar indicates the average position error in meter and the orange bar represents the average attitude error in degrees.

the FGSM attack has a more pronounced effect when ϵ is less than 0.05 on the predicted position in the ASMI Dataset. However, the attitude error remains quite stable, typically less than 1 degree, for all tested ϵ values.

748 4.2.3. LSTM-based Adversarial Attack Detector

To evaluate the adversarial attack detector on the ASMI Dataset, SHAP values are obtained by processing the pose estimator on the Synthetic-Lab Dataset. Similar to the previous synthetic data experiment, the SHAP values are obtained from the output of the GAP layer in the deep relative pose estimator by DeepSHAP algorithm. 1,000 images from the training 754 data on Synthetic-Lab Dataset are randomly selected to generate background data. A total of 30,000 SHAP value sam-756 ples, consisting of 15,000 normal samples and 15,000 adver-757 sarial samples, are used to train the adversarial attack detec-758 tor. The 15,000 normal samples consist of all images from the 759 test data on the Synthetic-Lab Dataset and randomly selected 760 images from the training data to account for the difference be-761 tween 15,000 and the total number of images in the test data. 762 Adversarial samples are generated by applying FGSM attacks 763 to the normal sample images with randomly selected ϵ values 764 from [0.5, 0.3, 0.1, 0.05, 0.01]. 765

The SHAP values are shuffled and split by a train-test ratio 766 of 0.8, i.e. 24,000 samples for training and 6,000 samples for 767 testing. The adversarial attack detector is trained by SGD with 768 an Adadelta optimiser for 2000 epochs. Early termination is im-769 plemented to reduce the training time. To do that, the training 770 data are further split into 80% for training and 20% for valida-771 tion. If the validation loss does not decrease over 20 epochs, the 772 training process will be terminated. After the early termination 773 condition, the proposed adversarial attack detector achieves a 774 detection accuracy of 99.18% on training data and 98.8% on 775 test data. 776

Subsequently, the pose estimator, FGSM attacks, and ad-777 versarial attack detector are integrated to evaluate the overall 778 performance on the ASMI Dataset. The integrated system is 779 identical to the one shown in Fig. 8, with the exception that 780 the 'Blender Image Generation' part is replaced by the ASMI 781 Dataset. In the ASMI Dataset, a random attack probability of 782 0.2 is applied to FGSM attacks. When an attack occurs, input 783 images are continuously perturbed by FGSM for the next 10 784 frames. The detection accuracy is calculated by Eq. (9). Ta-785 ble 10 presents the detection accuracy on the ASMI Dataset for 786 various ϵ values. 787

As shown in Table 10, the proposed adversarial attack detector achieves an average correct detection rate of 96.29% on the ASMI Dataset. The accuracy slightly drops when the ϵ value becomes smaller, which is caused by fewer perturbations ap-791

Table 10. The average accuracy of the adversarial attack detector in ASMI Dataset with various ϵ values.

ϵ	Detection Accuracy
0.5	100 %
0.3	100 %
0.1	100 %
0.05	98.44%
0.01	90.44 %
Average	96.29 %

⁷⁹² plied to the input images as ϵ decreases.

793 5. Conclusion

This paper firstly examines the impact of adversarial attacks on DL-based spacecraft relative pose estimation in space ren-dezvous scenarios. To do this, a CNN-based relative pose es-timation algorithm is proposed. FGSM adversarial attacks are implemented, which have a significant impact on the model's predictions. Subsequently, an LSTM-based adversarial attack detector is proposed to identify adversarial attacks on input im-ages. XAI techniques are adopted to analyse the model's pre-dictions and generate SHAP values-based explanations for the model's predictions. Multiple experiments are carried out to evaluate the performance of the CNN-based spacecraft relative pose estimator, how the adversarial attacks can impact on DL-based pose estimator in space rendezvous missions, and the per-formance of the proposed adversarial attack detector. The pro-posed methods have been tested on both synthetic and real im-age datasets. The results show that the adversarial attack detec-tor performs robustly in detecting adversarial attacks, achiev-ing an average of 99.21% detection rate on synthetic data and 96.29% on real data collected from the ASMI Lab.

Although the impact of digital adversarial attacks on DL-based spacecraft relative pose estimation has been analysed in this work, how to physically implement the adversarial attacks still needs to be explored. Moreover, the proposed method demonstrates high accuracy in detecting adversarial attacks for the DL-based spacecraft relative pose estimation, how to cor-rect the estimated pose after detecting adversarial attacks be-comes critical to provide a robust DL-based system for future space missions.

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