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**Citation**: Wang, Z., Aouf, N., Pizarro, J. & Honvault, C. (2024). Robust Adversarial Attacks Detection for Deep Learning based Relative Pose Estimation for Space Rendezvous. Advances in Space Research, doi: 10.1016/j.asr.2024.11.054

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**ADVANCES IN SPACE RESEARCH** (a COSPAR publication)

Advances in Space Research xx (2024) xxx-xxx

www.elsevier.com/locate/asr

# Robust Adversarial Attacks Detection for Deep Learning based Relative Pose Estimation for Space Rendezvous

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#### Abstract

Research on developing deep learning techniques for autonomous spacecraft relative navigation challenges is continuously growing in recent years. Adopting those techniques offers enhanced performance. However, such approaches also introduce heightened apprehensions regarding the trustability and security of such deep learning methods through their susceptibility to adversarial attacks. In this work, we propose a novel approach for adversarial attack detection for deep neural network-based relative pose estimation schemes based on the explainability concept. We develop for an orbital rendezvous scenario an innovative relative pose estimation technique adopting our proposed Convolutional Neural Network (CNN), which takes an image from the chaser's onboard camera and outputs accurately the target's relative position and rotation. We perturb seamlessly the input images using adversarial attacks that are generated by the Fast Gradient Sign Method (FGSM). The adversarial attack detector is then built based on a Long Short Term Memory (LSTM) network which takes the explainability measure namely SHapley Value from the CNN-based pose estimator and flags the detection of adversarial attacks when acting. Simulation results show that the proposed adversarial attack detector achieves a detection accuracy of 99.21%. Both the deep relative pose estimator and adversarial attack detector are then tested on real data captured from our laboratory-designed setup. The experimental results from our laboratory-designed setup demonstrate that the proposed adversarial attack detector achieves an average detection accuracy of 96.29%. © 2024 COSPAR. Published by Elsevier Ltd All rights reserved.

*Keywords:* Pose Estimation; Deep Learning; Adversarial Attack ; Adversarial Attack Detection; Explainable Artificial Intelligence

#### 1. Introduction

 The growth of deep learning-based techniques has drawn increasing attention in various domains of application, such as image processing, speech recognition, and many other [c](#page-17-0)hallenging Artificial Intelligence (AI) based tasks [\(Guo](#page-17-0)

https://dx.doi.org/10.1016/j.jasr.xxxx.xx.xxx

[et al., 2016\)](#page-17-0). Vision-based autonomous orbital space ren-dezvous [\(Wie et al., 2014\)](#page-18-0), is an application for which adopting deep learning approaches to spacecraft position and attitude estimation is continuously gaining interest within the research 9 [c](#page-17-1)ommunity and the space agencies [\(Song et al., 2022;](#page-18-1) [Kisantal](#page-17-1) 10 et al.,  $2020$ ).

The state-of-the-art achievements in deep learning (DL) research demonstrate that the Convolutional Neural Networks 13 (CNNs) have successfully gained outstanding performance in <sup>14</sup> computer vision applications, such as object detection and tar- <sup>15</sup> get localisation [\(Ren et al., 2017;](#page-18-2) [Redmon & Farhadi, 2018;](#page-18-3) <sup>16</sup>

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 [Cebollada et al., 2022\)](#page-17-2). Determining the pose of a spacecraft's relative state by processing input images is typically achieved through the six Degree-of-Freedom (6 DOF) pose estimation of the target object frame relative to the camera (onboard the spacecraft) frame. These vision-based pose estimation meth- ods are traditionally computed by matching relative features on images captured by the camera to relative locations in the tar- get frame. Different from the traditional approaches, the CNNs can be trained to detect features from raw image data and es- timate the relative pose by regressing the position and attitude, without the need for manual feature engineering which is often required in traditional computer vision methods. The advan- tages of CNN-based pose estimation approaches are that they can potentially lead to better performance in complex orbital scenarios and more robustness to variations in lighting, view-32 point, and cluttered background.

 Recent achievements in DL-based pose estimation demon- [s](#page-18-4)trate outstanding accuracy performance [\(Phisannupawong](#page-18-4) [et al., 2020;](#page-18-4) [Oestreich et al., 2020;](#page-18-5) [Rondao et al., 2022;](#page-18-6) [Chekakta et al., 2022\)](#page-17-3). However, the vulnerability of such deep learning scheme can be questionable [\(Chawla et al.,](#page-17-4) [2022;](#page-17-4) [Nemcovsky et al., 2022;](#page-18-7) [Tian et al., 2024\)](#page-18-8).Indeed, mi- nor changes in the spacecraft onboard camera acquired images that is used by the CNN-based pose estimator can cause CNNs to make wrong predictions due to their reliance on low-level af- fected features, such as edges and textures, and their high sen-43 sitivity to slight variations in the input space. Those changes in the input images and thus on features that the CNN-based pose [e](#page-17-5)stimation relies on can be caused by adversarial attacks [\(Lin](#page-17-5) [et al., 2020\)](#page-17-5). Adversarial attacks aim to make small perturba- tions to the input images that are imperceptible to human vi- [s](#page-17-6)ion and can significantly affect the CNN's prediction [\(Grabin-](#page-17-6) [ski et al., 2022\)](#page-17-6). For real-world applications where CNNs are applied to estimate the relative pose of spacecraft, applying an adversarial attack to the input images can potentially make the CNNs output the wrong position or attitude of the target. This could seriously damage the autonomous rendezvous operation system if wrong pose data are involved to generate any further

actions, such as guidance commands for the spacecraft to ren- <sup>55</sup> dezvous and/or dock to the target satellite.

One of the significant challenges associated with deep neural networks is that these models usually lack of transparency,  $\frac{58}{2}$ which means people cannot understand how the deep neural 59 networks achieve their decisions. To address this issue, eX- 60 plainable AI (XAI) aims to provide an understandable explanation for the AI models' decision-making process. By apply- 62 ing XAI methods to CNNs, such as Class Activation Mapping  $\frac{1}{63}$ (CAM) [\(Pope et al., 2019\)](#page-18-9), Layer-wise Relevance Propagation (LRP) [\(Nazari et al., 2022\)](#page-17-7) and SHapley Additive exPlanation  $65$ (SHAP) values [\(Lundberg & Lee, 2017\)](#page-17-8), users can understand  $66$ how CNNs work and why models output their relative pose 67 predictions. This nice characteristic of XAI methods can po- <sup>68</sup> tentially be adopted in detecting adversarial attacks on CNN 69  $models.$   $70$ 

This work aims to present an innovative demonstration of  $\frac{71}{11}$ the vulnerability of CNN-based spacecraft rendezvous relative  $\frac{72}{2}$ pose estimation scheme to digital adversarial attacks on camera  $\frac{73}{2}$ input images and proposes a novel method for detecting those  $\frac{74}{6}$ adversarial attacks when they may occur. In this paper, a vision  $\frac{75}{6}$ based orbital autonomous rendezvous dynamic scenario is sim- <sup>76</sup> ulated. A CNN-based pose estimator is designed and trained to  $77$ estimate the relative position and attitude of the target satellite  $\frac{78}{6}$ involving a modified Darknet-19 [\(Redmon & Farhadi, 2017\)](#page-18-10) as  $\frac{79}{9}$ a feature extractor. The Fast Gradient Sign Method (FGSM) is 80 employed to introduce small perturbation attacks to the input 81 images. Various configurations of the FGSM attack are devel-<br>82 oped to demonstrate the impact of digital adversarial attacks <sup>83</sup> on the CNN-based pose estimator. An LSTM-based detector 84 exploiting the explainable Shap values of the CNN based estimator is then proposed to detect the adversarial attacks acting s6 on the input images and thus the CNN based estimator outputs. 87 To this end, this paper makes the following contributions:

 $\bullet$  Firstly, a CNN-based relative pose estimator for closerange rendezvous is introduced, which is subsequently for- 90 mulated as the target DL-based navigation system against  $\frac{91}{2}$ adversarial attacks.  • Secondly, the Fast Gradient Sign Method (FGSM) [Good](#page-17-9)<sup>94</sup> [fellow et al.](#page-17-9) [\(2014\)](#page-17-9) is utilised to generate invisible pertur- bations in the input images, introducing a range of FGSM attack configurations to illustrate the effects of digital ad-versarial attacks on the CNN relative pose estimator.

 • Then, an LSTM-based adversarial attacks detection mech- anism is proposed, leveraging the explainable (SHAP) value [\(Lundberg & Lee, 2017\)](#page-17-8) from the CNN-based nav- igation system to identify adversarial attacks affecting the input images.

 • Subsequently, the CNN-based relative pose estimator and LSTM-based adversarial attacks detection mechanism have been evaluated in both synthetic data and real-world data obtained from our laboratory to demonstrate the per-formance of proposed frameworks.

 The paper is organised as follows: Section [2](#page-4-0) provides an overview of current DL-based spacecraft pose estimation ap- proaches and discusses existing methods for detecting adver- sarial attacks. Section [3](#page-6-0) outlines the proposed design of the CNN-based pose estimator, how to adopt FGSM attacks to the pose estimator, and the design of the LSTM-based adversarial attack detector. Section [4](#page-9-0) presents the test experiments that are conducted on both simulation data and real-world data obtained from our laboratory. Finally, Section [5](#page-17-10) concludes the paper and discusses future work.

#### <span id="page-4-0"></span><sup>118</sup> 2. Background and Related Works

#### <sup>119</sup> *2.1. DL-based Spacecraft Relative Pose Estimation*

 [Sharma et al.](#page-18-11) [\(2018\)](#page-18-11) proposed a relative pose classification network which is based on AlexNet [\(Krizhevsky et al., 2012\)](#page-17-11) 122 architecture for non-cooperative spacecraft. In their design, the convolutional layers in AlexNet are initially trained on Ima- geNet dataset [\(Deng et al., 2009\)](#page-17-12) as feature extractors. The pre- trained feature extractors are adopted with two fully-connected layers and one classification layer with training on ten sets of synthetic images that were created from Tango spacecraft flown in the Prisma mission [\(Persson et al., 2006\)](#page-18-12). Their work shows

that the CNN-based relative pose classification outperforms the 129 accuracy of an architecture based on classical feature detection algorithms. However, this network is designed to output 131 a coarse pose classification and cannot meet the precision re- <sup>132</sup> quirements for fine position and attitude estimation missions. 133

[Yang et al.](#page-18-13) [\(2021\)](#page-18-13) have proposed a CNN-based pose estima-<br>134 tion method to estimate the relative position and orientation of 135 non-cooperative spacecraft. In their approach, the pre-trained 136 ResNet-50 [\(He et al., 2016\)](#page-17-13) is adopted as the feature extrac-<br>137 tor, and two fully-connected layers are concatenated after the 138 feature extract to output the relative position and orientation of <sup>139</sup> the target spacecraft, respectively. To adapt the network to esti- <sup>140</sup> mate the relative pose of other similar spacecraft, an additional  $_{141}$ output layer is concatenated with the output of position and ori- <sup>142</sup> entation to predict the category of the target spacecraft. Dif- <sup>143</sup> ferent from previous work introduced by [Sharma et al.](#page-18-11) [\(2018\)](#page-18-11), <sup>144</sup> this work can output the relative position and orientation of the 145 target spacecraft, instead of a coarse pose classification. Sim- <sup>146</sup> ilarly, pre-trained ResNet has also been used as the backbone 147 by [Proença & Gao](#page-18-14) [\(2020\)](#page-18-14). In this work, the estimation of po- <sup>148</sup> sition is achieved by two fully-connected layers with a simple 149 regression, and the relative error is minimised based on the loss 150 weight magnitudes. Then, the continuous attitude estimation is 151 [p](#page-17-14)erformed via classification with soft assignment coding [\(Liu](#page-17-14) 152 [et al., 2011\)](#page-17-14).

Rather than estimating the relative pose of spacecraft by using a single input frame, consecutive image inputs have been 155 [c](#page-18-6)onsidered by group previous work, named ChiNet [\(Rondao](#page-18-6) 156 [et al., 2022\)](#page-18-6). The ChiNet featured a Recurrent Convolutional 157 Neural Network (RCNN) architecture, which involves a mod-<br>158 ified Darknet-19 [\(Redmon & Farhadi, 2017\)](#page-18-10) as an image fea- <sup>159</sup> ture extractor and followed by LSTM units to deal with the sequences of input images. The ChiNet takes 4-channels input 161 which not only includes the RGB image but also a thermal image of the spacecraft that has been stacked to the fourth channel 163 of input. The ChiNet also proposed a multistage optimisation <sup>164</sup> approach to train the deep neural network to improve the per- <sup>165</sup> formance in spacecraft relative pose estimation.

#### <sup>167</sup> *2.2. Explainability in CNNs*

 While recent approaches to DL-based spacecraft relative pose estimation demonstrate outstanding performance in terms of prediction accuracy, understanding how these models predict relative pose is essential for providing robust solutions for fu- ture space rendezvous missions. As a new approach solution, eXplainable AI (XAI) techniques offer the possibility to anal- yse gradients in DL models to indicate the significance of input variables in the estimation decision-making process.

 [Lundberg & Lee](#page-17-8) [\(2017\)](#page-17-8) proposed the SHAP values to in-177 terpret complex machine learning models. The SHAP value is based on a concept from game theory called Shapley val- ues. These are used to fairly distribute the payoff among the players of a cooperative game, where each player can have dif- ferent skills and contributions. Similarly, SHAP values assign each feature an importance value for a particular prediction and provide insights into the contribution of each feature. By ex- amining the SHAP values of machine learning models, we will able to understand the predictions of complex machine learning <sup>186</sup> models.

 Contrastive gradient-based (CG) saliency maps [\(Simonyan](#page-18-15) [et al., 2013\)](#page-18-15) are visual explanation methods for deep neural networks. They produce a heat map where the norm of the model's gradients indicates the significance of input variables. The heat map highlights the areas in the input image that would change the output class if they were changed. By accessing the heat map, users can identify the most relevant features for the model's prediction.

 Class Activation Mapping (CAM) [\(Zhou et al., 2016\)](#page-18-16) gen- erates visual explanation maps by finding the spatial locations in the input image that contribute the most to a specific pre- diction. The CAM is particularly helpful in image classifica- tion tasks through CNNs. Similarly, gradient-weighted CAM (Grad-CAM) [\(Selvaraju et al., 2017\)](#page-18-17) extends the work of CAM and provides visual explanations for decisions made by a wide range of CNN-based methods. Grad-CAM utilises the gradients of any target concept, flowing into the final convolutional layer to produce a localisation map that highlights the important regions in the input image for predicting the concept. These XAI methods interpret the CNNs, making people understand how <sup>206</sup> and why CNNs make certain predictions. However, since then, <sup>207</sup> there has been no specific analysis on interpreting the DL-based <sub>208</sub> spacecraft relative pose estimation to improve their explainabil-<br>209  $ity.$  210

**2.3. Adversarial Attacks** 211

Adversarial attacks for CNNs aim to make small perturba-<br>
<sub>212</sub> tions on the original input images where original and perturbed 213 images look similar in human vision but can significantly im- <sup>214</sup> pact the CNNs' predictions. However, very limited research 215 works are investigating how adversarial attacks can impact DL- <sup>216</sup> based pose estimation systems. [Chawla et al.](#page-17-4) [\(2022\)](#page-17-4) demon- <sup>217</sup> strate the effect of different types of adversarial attacks on the 218 predictions of the DL-based pose estimation system. Their 219 work shows that adversarial attacks can significantly impact 220 monocular pose estimation networks, leading to increased tra-jectory drift and altered geometry. Similarly, [Nemcovsky et al.](#page-18-7) 222 [\(2022\)](#page-18-7) illustrate that the physical passive path adversarial at- <sup>223</sup> tacks can seriously increase the error margin of a visual odom- <sup>224</sup> etry model which is used in autonomous navigation systems 225 leading onto potential collisions.

The impacts of adversarial attacks have garnered significant 227 attention in the DL-based autonomous systems. [Ilahi et al.](#page-17-15) <sup>228</sup> [\(2021\)](#page-17-15) provide an extensive overview of recent methodologies <sup>229</sup> for adversarial attacks on Deep Reinforcement Learning mod- <sup>230</sup> els applied to autonomous systems, as well as the existing tech- <sup>231</sup> niques for mitigating these attacks. Wang  $&$  Aouf [\(2024\)](#page-18-18) ex- $\frac{232}{2}$ amine the effects of perceptual perturbations on vision-based 233 autonomous driving systems and propose an optimised pol- <sup>234</sup> icy to counter adversarial attacks on observation perturbations. 235 [Tian et al.](#page-18-8) [\(2024\)](#page-18-8) explore multi-label adversarial example at- <sup>236</sup> tacks targeting multi-label False Data Injection Attacks for lo- <sup>237</sup> cational detectors, highlighting significant security vulnerabili- <sup>238</sup> ties in DL-based smart grid systems.

To protect the DL-based system from adversarial attacks, <sup>240</sup> [Liu et al.](#page-17-16) [\(2020\)](#page-17-16) proposed a detection method based on the ro- <sup>241</sup> bustness of the classification results. Their results show that 242

 the detector performs well against gradient-based adversarial attacks. Our group work, [Hickling et al.](#page-17-17) [\(2023\)](#page-17-17) proposed a CNN-based adversarial attack detector and an LSTM-based adversarial attack detector for Deep Reinforcement Learning (DRL) based Uncrewed Aerial Vehicle guidance. The simula- tion results show that the LSTM-based adversarial attack detec- tor leads to 90% detection accuracy on the DRL model. It also suggests that the LSTM-based detector performs much more accurately and quicker than the CNN-based adversarial attack detector. Indeed, the LSTM-based detector is demonstrated to meet the real-time requirement in DRL based guidance.

 To the best of our knowledge, as of yet, there is no literature looking at the impact of adversarial attacks in spacecraft relative pose estimation and how to detect those adversarial attacks in DL-based spacecraft relative pose estimation systems and this work first time proposes this. Our objective is to ultimately create an adversarial attack detector for the space navigation system, which employs SHAP values explainability mechanism to detect and flag potential adversarial attacks.

#### <span id="page-6-0"></span><sup>262</sup> 3. Methodology

 In this section, a CNN-based spacecraft relative pose esti- mator is newly designed with the aim of providing a reliable estimated position and attitude of the target spacecraft in as rendezvous scenario. Then, the FGSM attacks are adopted on the spacecraft onboard camera resulting in an adversarial im- age to evaluate the impacts on the proposed deep pose estima- tor. Next, SHAP values are introduced to generate XAI sig- natures for both adversarial and normal input images. Finally, an LSTM-based adversarial detector is proposed and trained, which learns normal and adversarial SHAP values to detect the adversarial attacks on the spacecraft relative pose estimator.

# <sup>274</sup> *3.1. CNN-based Spacecraft Relative Pose estimator* <sup>275</sup> *3.1.1. Overall architecture design*

 Similar to most DL-based spacecraft relative pose estimation 277 algorithms, CNN is applied to extract features in the proposed pose estimator. The overall design of the pose estimator follows the design methodology in ChiNet [\(Rondao et al., 2022\)](#page-18-6). The Darknet-19 [\(Redmon & Farhadi, 2017\)](#page-18-10) is originally trained in ImageNet [\(Deng et al., 2009\)](#page-17-12) dataset which has an input size of  $244 \times 244 \times 3$ . In our design, input images of the pose estimator 282 have a larger size than ImageNet images. Therefore, the first 283 convolutional layer in Darknet-19 is configured with a kernel <sup>284</sup> [s](#page-18-3)ize of  $7 \times 7$ . Following the approach of Darknet-53 [\(Redmon](#page-18-3) 285  $&$  Farhadi, 2018), the maxpooling layers in the Darknet-19 are replaced by  $3 \times 3$  convolution operation with a stride of 2. Similarly, as the Darknet-53 approaches, the residual connections 288 are also adopted to the proposed pose estimator. Batch Normalisation (Ioff[e & Szegedy, 2015\)](#page-17-18) layers are applied after each  $290$ convolutional layer.

Our deep spacecraft relative pose estimator aims to output 292 the relative position and attitude of the target directly. There- <sup>293</sup> fore, two separate FC layers are applied. The first FC layer in- <sup>294</sup> volves 3 output nodes to output the relative position in  $(x, y, z)$  295 and the second FC layer adopts a 6-dimensional (6-D) vector to <sup>296</sup> represent the relative attitude of the target spacecraft. Finally, <sup>297</sup> two FC layers are concatenated together to output the relative 298 6-DOF pose. In the second FC layer, 6-D vectors are applied to represent the relative attitude of the target spacecraft, instead of 300 using quaternion representation. The reason is that the relative  $301$ pose estimator is designed as a regression problem where the 302 output has to be continuous. However, the normal attitude representation of quaternion is discontinuous, due to its antipodal <sub>304</sub> ambiguity, i.e.  $-q = q$ . Therefore, the proposed pose estimator 305 applies the 6-D vector formulated by [Zhou et al.](#page-18-19)  $(2019)$ , which  $306$ mapped the 3-dimensional rotations into a 6-D continuous rota-<br><sub>307</sub> tion. The overall design of the spacecraft relative pose estimator 308 is presented in Fig. [1.](#page-7-0)  $\frac{308}{200}$ 

#### *3.1.2. Synthetic data generation* 310

To train and test the spacecraft relative pose estimator, syn- <sup>311</sup> thetic datasets are generated in Blender, which is an open- <sup>312</sup> source 3D modelling software. The spacecraft target model 313 used in the synthetic dataset generation is the Jason-1 satellite, which was downloaded from the NASA 3D Resources 315 website [\(Jason-1 3D Model\)](#page-17-19). Dynamic simulation of the ren- <sup>316</sup> dezvous is developed to generate the synthetic dataset in which 317



Fig. 1. The overall architecture of the proposed spacecraft relative pose estimator. The blue blocks represent the convolutional layers, which are formatted as (*layer size*, *kernel size*, *stride*)). Each convolutional layer is followed by a batch normalisation layer and LeakyReLu activation. The yellow block indicates the Global Average Pooling (GAP) layer that downsamples the exacted features to a fixed 1D vector of 1000 units. The green blocks represent FC layers that will output the estimated relative position and attitude, respectively.

 the camera onboard the chaser spacecraft starts at 60 metres away in *z* − *axis* from the target and end at 10 metres away 320 from the target in  $z - axis$ , i.e. (0,0,10). Random rotation of the camera and target is considered in the synthetic data gen- eration. Many trajectory sequences are generated and each se- quence contains 2,500 RGB images. Each image has a size of 744  $\times$  480. To prevent overfitting in the deep relative pose estimator network, random rotation of the target spacecraft is applied to the model, and the camera is initialised at various positions in the synthetic data generation. Table [1](#page-7-1) illustrates the synthetic dataset generated for training and validating the deep pose estimator.

<span id="page-7-1"></span>



#### <span id="page-7-0"></span>*3.1.3. Loss Function* 330

Training the spacecraft relative pose estimator can be for- <sup>331</sup> mulated as a regression problem, where the total loss function 332 combines the loss in position and loss in attitude. These are 333 computed by Eq.  $(1)$  and Eq.  $(2)$ , respectively, which were orig- $334$ [i](#page-18-6)nally proposed by [Kendall et al.](#page-17-20) [\(2018\)](#page-17-20). Followed by [Rondao](#page-18-6) 335 [et al.](#page-18-6) [\(2022\)](#page-18-6), a trainable weight is attributed to each loss, which  $336$ corresponds to the task-specific variance of the Gaussian distri-<br>337 bution. The total loss is then formulated in Eq.  $(3)$ .  $338$ 

<span id="page-7-2"></span>
$$
L_p = \sum_{i=0}^{B} (||p_{pred}^i - p_{gt}^i||) \tag{1}
$$

<span id="page-7-3"></span>
$$
L_r = \sum_{i=0}^{B} (||r_{pred}^i - r_{gt}^i||)
$$
 (2)

<span id="page-7-4"></span>
$$
L_{total} = exp(-2\sigma_p)L_p + exp(-2\sigma_r)L_r + 2(\sigma_p + \sigma_r)
$$
 (3)

where the *p*<sub>pred</sub> and *r*<sub>pred</sub> indicate the predicted position and 339 attitude, and  $p_{gt}$  and  $r_{gt}$  indicate the ground truths position and  $340$ attitude, respectively. *B* is the batch size and  $\|\cdot\|$  donates the  $L_2$  341 norm.  $\sigma_p$  and  $\sigma_r$  represent the learnable weights for position  $\sigma_{342}$ and attitude, respectively. 343

#### *3.2. Adversarial Attacks* 344

In this work, the adversarial examples are generated by 345 FGSM attacks [\(Goodfellow et al., 2014\)](#page-17-9). The FGSM attacks 346 aim to add small perturbations to the input images which will 347



<span id="page-8-1"></span>Fig. 2. An example of applying FGSM attacks to the input image. (a) the original input image. (b) perturbation patch with  $\epsilon = 0.05$ . (c) resultant adversarial image.

 maximise the network's loss. The efficacy of adversarial at- tacks, including the FGSM attacks used in this work, can be influenced by the backbone neural network employed in per- ception systems. Different neural network architectures may exhibit varying levels of robustness and vulnerability to spe- cific types of adversarial attacks. Therefore, the effectiveness of these adversarial patches is inherently related to the specific CNN architectures employed. The equation in Eq. [\(4\)](#page-8-0) describes how to generate an adversarial example for a given input image *x* by FGSM attack.

<span id="page-8-0"></span>
$$
x' = x + \epsilon \times sign(\nabla_x L(\theta, x, y))
$$
\n(4)

358 where  $\epsilon$  is a value of the perturbation effect which describes <sup>359</sup> how strong the attack is. *L* is the loss of the input *x* with ground 360 truth of *y*. The  $(\nabla_x \text{ calculates the loss gradient, } L \text{ for input im-}$ 361 age *x* with relative ground truth *y*, and  $\theta$  indicates the trained <sup>362</sup> network's parameters. Depending on the quality of input im-<sup>363</sup> ages and the attack strength, the result of the FGSM attack can  $364$  be modified by changing the  $\epsilon$  value.

365 In real implementation, the  $\epsilon$  needs to be small enough to ensure the perturbations on the input image are seamless and cannot be visible by human vision but still significantly change 368 the deep model's predictions. The  $\epsilon$  value should be in the range of  $(0,1)$ , where a value of 0 means the adversarial image will be the same as the input image without any perturbation and a value of 1 means the adversarial image will be perturbed as significant distorted image to human vision. Fig. [2](#page-8-1) illustrates 373 an example of applying FGSM attacks to input images of the spacecraft relative deep pose estimator.

#### *3.3. Explanability and Adversarial Attacks Detector* <sup>375</sup>

#### *3.3.1. Explanability via DeepSHAP* 376

The black-box nature of deep neural networks makes users 377 can only observe the prediction of these models, but do not 378 know the reasons for getting correct or wrong predictions. XAI 379 techniques are developed to interpret the DL models. When the 380 model's prediction is changed, the XAI will generate relative 381 explanations to explain why the model is getting the prediction. In this work, we proposed a novel approach that adopts XAI 383 techniques by applying the change in SHAP values of the input 384 images as a measure to determine whether an adversarial attack 385 happens on input images.

Originally, SHAP is proposed based on the idea of Shap- <sup>387</sup> ley values, which are designed to assign a credit to every input feature for a given prediction. Generating SHAP values for DNNs can be computationally expensive, as the DNNs nor-<br><sub>390</sub> mally contain a huge amount of features. Thanks to the work of 391 DeepLIFT [\(Shrikumar et al., 2017\)](#page-18-20), Shapley values for DNNs 392 can be estimated by linearising the non-linear components of a 393 [n](#page-17-8)eural network, a method referred to as DeepSHAP [\(Lundberg](#page-17-8) [& Lee, 2017\)](#page-17-8). This is achieved by utilising a reference input  $\frac{395}{2}$ distribution, which can be linearly approximated, to estimate 396 the expected value for the entire model. 397

However, directly generating SHAP values for the spacecraft 398 relative pose estimator still requires a large amount of computa- <sup>399</sup> tional resources. The pose estimator is based on CNNs with image inputs that contain thousands of pixels. Using DeepSHAP 401 for image input requires generating Shapley values for each sin- <sup>402</sup> gle pixel for every output neuron. Therefore, in this work, we  $403$   consider computing the SHAP values for the subsampling layer in the pose estimator, instead of computing them for the input image. As demonstrated previously, the spacecraft relative pose estimator contains a GAP layer that downsamples feature maps from the prior convolutional layer to 1000 samples. For exam- $_{409}$  ple, computing SHAP values for a  $744 \times 480$  RGB image needs to compute 1,071,360 pixels, instead, the GAP layer in the pose estimator only employs 1000 neurons. As a result, SHAP val- ues are generated for the outputs of the GAP layer that only need to compute 1000 features. This saving in the computation makes the generation of SHAP values for the deep pose estima-415 tor could potentially meet the implementation time constraints.

#### <sup>416</sup> *3.3.2. Adversarial attack detector*

 To detect any incoming adversarial attacks on the spacecraft deep relative pose estimator trhough the onboard camera, an LSTM-based adversarial attacks detector is proposed. The de- tector aims to monitor the SHAP values generated from the out- put of the GAP layer and detect any slight anomaly changes that could result based an adversarial attack. The LSTM is a type 423 of Recurrent Neural Networks (RNNs) that is widely used in [l](#page-18-21)earning from time-series data, such as speech recognition [\(Yu](#page-18-21) [et al., 2019\)](#page-18-21). The LSTM architecture was originally proposed to 426 address the long-term dependency issue in conventional RNNs. It can enable the propagation and representation of information over a sequence without causing useful information from dis-tant past time steps to be ignored.

 In our approach, the SHAP values are generated based on the prediction of each output neuron in the proposed deep pose estimator. Different from applying adversarial attacks on a clas- sification CNN that only change the output label, when an at- tack occurs on the deep pose estimator, it could affect all output neurons to estimate for wrong position and attitude. Therefore, it can be assumed that there might exist a certain level of depen- dencies among those output neurons. From this point of view, building an LSTM-based adversarial attack detector can poten-tially achieve high detection accuracy.

<sup>440</sup> Fig. [3](#page-9-1) introduces the architecture of the proposed adversar-<sup>441</sup> ial attack detector. The detector takes the SHAP values that are



<span id="page-9-1"></span>Fig. 3. Proposed adversarial attack detector. The yellow block indicates the LSTM layer which has an input shape of (9,1000) and an output space of 100. ReLu is applied as the activation function for the LSTM layer. The blue blocks are FC layers in the format of (*units*, *activation*). The green block indicates the output layer of the adversarial detector, which is also formed from the FC layer and outputs a Boolean to detect adversarial attacks.

computed from the GAP layer of the deep pose estimator. As  $_{442}$ there are nine output neurons in the proposed deep pose estimator, the shape of the SHAP values is  $(9, 1000)$ . To input SHAP  $_{444}$ values to the detector, the SHAP values are formatted as a se- <sup>445</sup> quence data with a length of 9. The detector outputs a Boolean,  $446$ *True*/*False*, to indicate the result of detecting adversarial at-<br>tacks. tacks.

#### <span id="page-9-0"></span>**4. Experimental Results** 449

To validate our adversarial detection approach, two experi- <sup>450</sup> ments are performed. The first experiment is built on the sim- <sup>451</sup> ulation environment with synthetic data as mentioned in Sec-<br>452 tion [3.](#page-6-0) The second experiment is built on our lab environment  $453$ to testing our approach with real data. For both sets of exper- <sup>454</sup> iments, the spacecraft deep relative pose estimator and the adversarial attack detector are tested for their relevant accuracy, <sup>456</sup> and then the two systems are integrated to test the overall suc- <sup>457</sup> cessful detection rate of adversarial attacks. <sup>458</sup>

# *4.1. Results on Synthetic Data*  $\frac{459}{459}$

# *4.1.1. Accuracy of the Spacecraft Deep Relative Pose Estima-* <sup>460</sup> *tor*  $461$

To train the deep relative pose estimator, image data are col- <sup>462</sup> lected from the Blender 3D model. There are 13 sequences of 463 images generated from Blender with the relevant trajectories 464 that are mentioned in Table [1.](#page-7-1) By following the trajectories in  $465$   Table [1,](#page-7-1) 2,500 images are generated for each trajectory, result- ing in a dataset with 32,500 images for training and testing in total. Fig. [4](#page-10-0) shows two examples of synthetic data generation in Blender.



Fig. 4. Examples of synthetic data generated from Blender. (a) image captured at a camera position of (0, 0, 60m). (b) image captured at a camera position of (0, 0, 10m). Random rotation is applied to the target spacecraft.

<span id="page-10-0"></span><sup>470</sup> The synthetic dataset is split by a train-test ratio of 0.8, i.e. 80% of images in the dataset are used for training the deep rel- ative pose estimator, and 20% of images are used to test the model's accuracy. Each image is associated with a ground truth label in the format of  $(x, y, z, w, x_i, y_j, z_k)$ . The first three ele- ments in the ground truth label represent the relative position of the chaser onboard camera to the target and the rest 4 elements represent the target attitude in quaternion representations in the chaser camera frame. The deep relative pose estimator outputs the attitude in a 6-D vector. Therefore, to calculate the loss in attitude, the quaternion representations are converted to the [6](#page-18-19)-D vector representation by following the approach in [\(Zhou](#page-18-19) [et al., 2019\)](#page-18-19). A dropout rate of 0.2 is applied to the GAP layer in the training process. Multiple data augmentation techniques are considered in training the deep relative pose estimator, includ- ing Gaussian blur, Gaussian noise, image compression, random brightness and so on. These techniques help to prevent the model from overfitting the training dataset. The deep relative pose estimator is trained by stochastic gradient descent with an Adam optimiser. The triangular2 [\(Smith, 2017\)](#page-18-22) policy is ap-plied for cycling learning rate with base learning of 2.5e-5.

 After training the deep relative pose estimator for 50 epochs with the training batch size of 32, The model's average predic- tion accuracy for both training and test datasets is reported in Fig. [5.](#page-10-1) In this experiment, the position error is measured by Eq.  $(5)$  and the attitude error is measured by Eq.  $(6)$ .

<span id="page-10-2"></span>
$$
p_{err} := ||p_{pred} - p_{gt}|| \tag{5}
$$

<span id="page-10-3"></span>
$$
r_{err} := 2 \arccos(q_{pred}^{-1} \otimes q_{gt}) \tag{6}
$$

where  $p_{pred}$  and  $p_{gt}$  represent the prediction of position and the  $497$ ground truth of position magnitude. The  $q_{pred}$  and  $q_{gt}$  indicate  $\frac{498}{2}$ the prediction of attitude and the ground truth of attitude in  $499$ quaternion representation. The  $\otimes$  denotes the quaternion multiplication and  $\|\cdot\|$  denotes the  $L_2$  norm.



<span id="page-10-1"></span>Fig. 5. The prediction accuracy of the proposed pose estimator on training and test dataset after 50 epochs. The blue bar presents the average error on training data and the orange bar represents the average error on test data

The proposed spacecraft relative pose estimator achieves an accuracy of around 0.49m in position error and 0.68 deg in at-titude error on the test dataset. Table [2](#page-11-0) reports a comparison 504 between the proposed deep relative pose estimator and state-<br>
<sub>505</sub> of-the-art performance of other DL-based space relative pose <sup>506</sup> estimation approaches based on their datasets. The comparison here aims to show that the proposed spacecraft deep relative 508 pose estimator can achieve relatively good performance on the 509 synthetic data and can be applied as a baseline model to imple-<br>510 ment the adversarial attack algorithm on and test the adversarial 511 attack detector. The comparison is not meant to be a quantita- <sup>512</sup> tive benchmark evaluation of our approach relative to existing 513 performing approaches.  $514$ 

#### *4.1.2. FGSM Adversarial Attacks* 515

As discussed in Section [3,](#page-6-0) the perturbation made by FGSM  $_{516}$ attacks can be adjusted by changing the  $\epsilon$  value. To investi-

Table 2. Comparison with other approaches in DL-based space relative pose estimation

<span id="page-11-0"></span>

Model	Dataset	Position Error (m)	Attitude Error (deg)
Proença & Gao $(2020)$	SPEED (Kelvins - ESA's Advanced Concepts Competition Website)	0.56	8.0
Rondao et al. (2022)	Synthetic	1.73	6.62
Yang et al. (2021)	Synthetic	[0.052, 0.039, 0.077]	[0.213, 0.233, 0.097]
Ours	Synthetic	0.49	0.68

 gate the impact of adversarial attacks on DL-based space rela- tive pose estimation, different  $\epsilon$  values are selected to generate adversarial onboard camera image input to the proposed deep relative pose estimator. Typically, the  $\epsilon$  applied in this experi-522 ment are 1, 0.5, 0.3, 0.1, 0.05 and 0.01. The larger value of  $\epsilon$  is, the more perturbations are made to images. The FSGM attack is applied to all synthetic test data, where all images in the test data. Then, the perturbed images are fed to the deep relative pose estimator for testing the impact of the FGSM attack. The average prediction relative pose errors of applying different  $\epsilon$ values are reported in Fig. [6.](#page-11-1)



<span id="page-11-1"></span>Fig. 6. Comparison of the prediction error of pose estimator under FGSM attack on test data with various  $\epsilon$  values. The blue bar indicates the average position error and the red bar indicates the average attitude error on test data. The error magnitude for the position is metres and the error magnitude in rotation is measured by degrees.

528

529 We can see that as the  $\epsilon$  value increases, the deep model's <sup>530</sup> prediction error becomes larger. The attitude error is quite sta-531 ble on  $\epsilon = 0.1, 0.05$  and 0.01, but has a dramatic increase if the  $532 \quad \epsilon > 0.3.$ 

 To assess well how the adversarial attack can impact the DL-based navigation system in a space rendezvous scenario, a simple guidance scheme is implemented with the proposed deep relative pose estimator. The guidance scheme takes the estimated relative pose from the proposed deep relative pose  $537$ estimator and then provides relative control actions to move 538 the camera (spacecraft) to the target position. In the guidance 539 scheme, the camera has an initial position of  $(0, 0, 60)$  and a  $_{540}$ target position of  $(0, 0, 10)$  with  $\pm 1m$  tolerance. The guidance  $\frac{541}{2}$ scheme updates the camera position with a maximum velocity 542 of  $1m/s$ , as described in Eq. [\(7\)](#page-11-2) and Eq. [\(8\)](#page-11-3)  $\frac{543}{2}$ 

<span id="page-11-2"></span>
$$
p_{new} = \begin{cases} p_{est} - 1 & \text{if } diff \ge 1 \\ p_{est} - diff & \text{otherwise} \end{cases}
$$
 (7)

<span id="page-11-3"></span>
$$
diff = p_{est} - p_{tar} \tag{8}
$$

544

547

where  $p_{new}$ ,  $p_{est}$ ,  $p_{tar}$  present the updated position, estimated 545 position and target position of the camera, respectively. The <sup>546</sup> test system is implemented as shown in Fig. [7.](#page-11-4)



<span id="page-11-4"></span>Fig. 7. Test system for proposed pose estimator on Blender in simulated space rendezvous scenario.

In this experiment, the test system is continuously attacked  $_{548}$ by FGSM on image data with various acquired camera frames. <sub>549</sub> The *success* attack is defined as the camera (spacecraft) missing 550 the target position while the *f ailure* attack means that the camera (spacecraft) can still reach the target position under continu- <sup>552</sup> ous FGSM attack. Experimental results are reported in Table [3](#page-12-0) 553  $-7.$  554

<span id="page-12-0"></span>Table 3. FGSM attacks on the simulated space rendezvous scenario with  $\epsilon$ =0.5

$\epsilon = 0.5$					
Continuously		5	10	15	20
<b>Attacked Frame</b>					
	60	failure	failure	failure	<b>Success</b>
	50	failure	failure	failure	<b>Success</b>
Attack start point	40	failure	failure	<b>Success</b>	<b>Success</b>
	30	failure	failure	<b>Success</b>	<b>Success</b>
(m)	20	failure	<b>Success</b>	<b>Success</b>	<b>Success</b>
	10	<b>Success</b>	<b>Success</b>	<b>Success</b>	<b>Success</b>

Table 4. FGSM attacks on the simulated space rendezvous scenario with  $\epsilon$ =0.3

$\epsilon = 0.3$					
Continuously		5	10	15	20
<b>Attacked Frame</b>					
	60	failure	failure	failure	<b>Success</b>
	50	failure	failure	<b>Success</b>	<b>Success</b>
Attack start point	40	failure	failure	<b>Success</b>	<b>Success</b>
	30	failure	failure	<b>Success</b>	<b>Success</b>
(m)	20	failure	<b>Success</b>	<b>Success</b>	<b>Success</b>
	10	failure	<b>Success</b>	<b>Success</b>	<b>Success</b>

Table 5. FGSM attacks on the simulated space rendezvous scenario with  $\epsilon$ =0.1

$\epsilon = 0.1$					
Continuously <b>Attacked Frame</b>		5	10	15	20
	60	failure	failure	failure	failure
	50	failure	failure	failure	failure
Attack start point	40	failure	failure	failure	<b>Success</b>
	30	failure	failure	failure	failure
(m)	20	failure	failure	failure	failure
	10	failure	failure	failure	<b>Success</b>

Table 6. FGSM attacks on the simulated space rendezvous scenario with  $\epsilon$ =0.05

$\epsilon = 0.05$					
Continuously		5	10	15	20
<b>Attacked Frame</b>					
	60	failure	failure	failure	failure
	50	failure	failure	failure	<b>Success</b>
Attack start point	40	failure	failure	failure	<b>Success</b>
	30	failure	failure	<b>Success</b>	<b>Success</b>
(m)	20	failure	<b>Success</b>	<b>Success</b>	<b>Success</b>
	10	failure	<b>Success</b>	<b>Success</b>	<b>Success</b>

<span id="page-12-1"></span>Table 7. FGSM attacks on the simulated space rendezvous scenario with  $\epsilon$ =0.01



From Table [3](#page-12-0) - [7,](#page-12-1) we can clearly see that the adversarial attack can result in a significant impact on the guidance scheme if DNN-based relative navigator is attacked, typically when the 557 distance between the camera and the target is less than 30m.  $\frac{558}{20}$ In most cases, continuously attacking the deep model for more 559 than 15 frames after the camera approaches less than 30m to the 560 target, the camera (spacecraft) will fail to reach the target posi- <sup>561</sup> tion. In a real space rendezvous mission where a chaser relies  $_{562}$ on a DL-based relative pose estimation system, an adversarial 563 attack has the potential to cause the chaser to fail in approaching the target position, resulting in mission failure. Therefore,  $565$ detecting adversarial attacks on DL-based pose estimators be-<br>
<sub>566</sub> comes critical.

# *4.1.3. LSTM-based Adversarial Attack Detector* <sup>568</sup>

The proposed adversarial attack detector is designed based 569 on the LSTM architecture. It aims to detect the change in SHAP 570 values when an adversarial attack occurs on the input image. As  $571$ mentioned in Section [3,](#page-6-0) the SHAP values are computed at the 572 output of the GAP layer in the proposed deep relative pose es- <sup>573</sup> timator. The GAP layer contains 1000 neurons, therefore, 1000  $574$ values are calculated for each output neuron, resulting  $9\times1000$  575 output SHAP values. 576

In our approach, the SHAP values of the GAP layer are cal-<br> $577$ culated by DeepSHAP (Lundberg  $&$  Lee, 2017) algorithm. The  $\frac{578}{20}$ DeepSHAP algorithm computes SHAP values for inputs by integrating over background samples. It then estimates approx- <sup>580</sup> imate SHAP values in a manner that sums up the difference 581 between the expected deep model's output on the background <sub>582</sub> samples and the current model's output. In this work, 1000 583 images are randomly selected from the training dataset to compute the downsampled features at the GAP layer. These sam-<br>
<sub>585</sub> ples serve as the background samples for the DeepSHAP ex- <sup>586</sup> plainer. To train the adversarial attack detector, we generated 587 15,000 sets of SHAP values for normal samples and an addi- <sup>588</sup> tional 15,000 sets of SHAP values for adversarial samples. The normal samples consist of the entire test dataset, which is used 590 for testing the deep pose estimator, along with a random selec-  $\frac{591}{2}$ tion of images from the training dataset. This random selection 592  was made to reach a total of 15,000 samples, thereby bridging the gap between this number and the number of images in the test dataset by the deep relative pose estimator. The adversar- ial instances are crafted by launching attacks on the DRLs at 597 arbitrary time steps with random  $\epsilon$  values: 0.5, 0.3, 0.1, 0.05, and 0.01. Subsequently, 3,000 perturbed images are randomly 599 selected from each  $\epsilon$  value for calculating the corresponding SHAP values.

<sup>601</sup> The SHAP values for both normal and adversarial samples are split into a training and testing set using a 0.8 train-test ra- tio, resulting in 24,000 samples for training and 6,000 samples for testing. The adversarial attack detector is trained using the Stochastic Gradient Descent (SGD) method with the Adadetal optimiser for 1,000 epochs. After training the adversarial at- tack detector, it achieved a training accuracy of 99.98% and a test accuracy of 99.90% on the test dataset. In this case, the detection accuracy is calculated by Eq. [\(9\)](#page-13-0)

<span id="page-13-0"></span>
$$
accuracy = \frac{successful Detection}{Total No. of Frames} \times 100\%
$$
 (9)

 where the *success f ul Detection* is defined by that the input frames with the adversarial attack are detector as *True* and frames without adversarial attack are detector as *False*. The experimental results show that the proposed detector can suc- cessfully detect adversarial attacks on the DL-based relative pose estimator with high accuracy. The adversarial attack de- tector is integrated with the deep relative pose estimator and the 617 DeepSHAP explainer to enhance accuracy in space rendezvous scenarios. The overall system is presented in Fig. [8.](#page-14-0)

<sup>619</sup> The adversarial attack detector is then tested with three tra-<sub>620</sub> jectories. In each trajectory, the camera (spacecraft) starts 60 meters away from the target, positioned at various points in the *x* and *y* directions within the range of  $[\pm 25, \pm 15]$ . The camera is oriented directly toward the target, with an attitude repre- $\epsilon_{624}$  sented as quaternion  $(1, 0, 0, 0)$ . The end position is  $(0, 0, 10)$ . The camera (spacecraft) moves linearly at a rate of 0.25 meters per frame along the *z*-axis. It follows a projectile trajectory in the *x* and *y* directions, resulting in a total of 2,500 frames for each trajectory. The FGSM attack is applied to test trajectories

<span id="page-13-1"></span>Table 8. The average accuracy of the adversarial attack detector in test trajectories with various  $\epsilon$  values.

$\epsilon$	Trajectory	<b>Detection Accuracy</b>
	1	100%
0.5	2	100%
	3	100%
	1	100%
0.3	2	100%
	3	99.98%
	1	99.96%
0.1	2	99.98%
	3	99.96%
	1	100%
0.05	2	99.98%
	3	99.98%
	1	97.06%
0.01	2	96.94%
	3	99.02%
Average		99.21%

with an attack probability of 0.2. Once FGSM is initiated, attacks continue for the subsequent 5 frames. The results of the 630 proposed adversarial attack detector are presented in Table [8.](#page-13-1) 631 From the test results, the proposed adversarial attack detector 632 successfully detects all incoming FGSM attacks when the  $\epsilon = 633$ 0.5. As the  $\epsilon$  value goes small, i.e. fewer perturbations are made  $\epsilon_{0.54}$ to input images, the detection accuracy has slightly dropped. 635 For these three test trajectories, the proposed adversarial attack  $\epsilon_{0.6}$ detector achieves a detection accuracy of 99.21% on average.  $637$ 

# *A.2. Experimental Results on Real Data* 638

In previous experiments, both the proposed deep relative  $\frac{639}{639}$ pose estimator and the adversarial attack detector exhibited high accuracy on synthetic data. To further evaluate the per- 641 formance of both systems, we tested them with real-world im- <sup>642</sup> ages obtained from the Autonomous Systems and Machine In- <sup>643</sup> telligence Laboratory (ASMI Lab) at City, University of Lon- <sup>644</sup> don. These data include sensor noise, camera calibration noise, <sup>645</sup> ground truth measurement noise, and different lighting condi- <sup>646</sup> tions that are not present in the training synthetic images. 647

# *4.2.1. Accuracy of the Spacecraft Dee Relative Pose Estimator* <sup>648</sup>

At the ASMI Lab, a scaled mock-up model of the Jason- <sup>649</sup> 1 spacecraft is constructed. This mock-up model is 1/9 the 650 size of the actual Jason-1 spacecraft. The vision sensor ap- 651



Fig. 8. The experimental system includes the integration of an adversarial attack detector with the relative pose estimator and SHAP values generator.

 plied for real data acquisition is the ZED 2 camera, which out- puts images with a resolution of 1920 $\times$ 1080. The deep relative pose estimator is retrained on new synthetic data, referred to as the Synthetic-Lab Dataset, with an input RGB image size of 480  $\times$  270 to match the aspect ratio of the camera used in the ASMI Lab. As before, the Synthetic-Lab Dataset is generated using Blender, where the target was replaced with a 3-D model of the ASMI Lab mock-up Jason-1. To simulate the space ren- dezvous scenario over a distance range from 60m to 10m, the 3-D model is scaled up by a factor of 9 in Blender data genera-tion. An example of the re-training images is shown in Fig. [9](#page-14-1)



Fig. 9. Am example of images generated from Blender for training the pose estimator.

<sup>663</sup> Similar to the previous synthetic data experiment, multiple <sup>664</sup> trajectories are generated to collect images from the Blender, <span id="page-14-0"></span>resulting in a total of 32,500 images on Synthetic-Lab Dataset 665 for training and testing. The hyperparameter settings for train- <sup>666</sup> ing are the same as the settings applied in previous synthetic  $\frac{667}{667}$ data experiment, including the learning rate, optimiser, batch size, and data augmentation methods. The pose estimator was 669 trained for 100 epochs with a train/test split of 0.8.  $\frac{670}{ }$ 

<span id="page-14-1"></span>There are three sets of images captured from the ASMI  $671$ Lab, referred to as the ASMI Dataset, with each set contain- 672 ing a total of 750 images. To acquire images for the ASMI  $673$ Dataset, the camera movement is controlled by the Rethink  $674$ Robotics Sawyer (Sawyer| [Rethink Robotics\)](#page-18-23) moving along the 675 *z*-axis, and the ground truths relative poses of the images in 676 ASMI Dataset are recorded by the OptiTrack Motion Capture 677 Systems [\(OptiTrack\)](#page-18-24). The OptiTrack Motion Capture System 678 records the position and attitude of the ASMI Lab mock-up 679 Jason-1 and the ZED camera at a frame rate of 120 frames 680 per second and assigns a timestamp to each frame. Images are 681 acquired by the ZED camera at a resolution of  $1920 \times 1080$  682 and a frame rate of 30 frames per second, with relevant timestamps. The ground truth pose for each frame acquired by the 684 ZED camera are assigned by matching the corresponding times-<br>685 tamps from the OptiTrack Motion Capture System. Then, the 686 <sup>687</sup> relative ground truth position is calculated by the difference between the actual positions of the ZED camera and the ASMI <sup>689</sup> Lab mock-up Jason-1, as shown in Eq. [\(10\)](#page-15-0),

<span id="page-15-0"></span>
$$
Pos_{lab} = Pos_{camera} - Pos_{target}
$$
 (10)

 where *Poslab* donates relative ground truth position in ASMI Dataset. The *Poscamera* and *Postarget* donate the actual position of the ZED camera and ASMI Lab mock-up Jason-1 recorded by OptiTrack Motion Capture System, respectively.

 Due to different camera intrinsic matrices applied between the Synthetic-Lab Dataset and ASMI Dataset, to represent the relative position in the trained model, the position ground truths 697 of the ASMI Dataset are collaborated with the camera view by the following processing:

$$
K_{Blender} = \begin{bmatrix} 640 & 0 & 240 \\ 0 & 360 & 135 \\ 0 & 0 & 1 \end{bmatrix}
$$
 (11)

 $T_{\text{aroot}}$   $I = 0 \times T_{\text{aroot}}$  (13)

699

$$
K_{zed} = \begin{bmatrix} 1400.41 & 0 & 956.29 \\ 0 & 1400.41 & 557.258 \\ 0 & 0 & 1 \end{bmatrix}
$$
 (12)

700 701

<span id="page-15-3"></span>
$$
Pos_{real} = 1400.41 \times \frac{240}{956.29} \times \frac{Target_{lab}}{Target_{real}} \times \frac{1}{640} \times Pos_{lab} \quad (14)
$$

 where *KBlender* and *Kzed* represent the camera intrinsic matrices for the camera used in Synthetic-Lab Dataset collection and the ZED camera that is used to acquire images in the ASMI Lab,  $r_{05}$  respectively. *Target<sub>real</sub>* and *Target<sub>lab</sub>* indicate the target sizes in the Blender 3-D model and the actual size in the ASMI Lab. *Posreal* and *Poslab* denote the relative position of the target in the pose estimator and the ground truth position in the ASMI Lab, respectively. Table [9](#page-15-1) illustrates the range of relative posi- tions in the ASMI Dataset and representative relative positions in trained pose estimator. Furthermore, all images in the ASMI Dataset are segmented with a black background and resized to 480  $\times$  270 to fit the input image size of the trained pose estima- tor. An example of images captured in ASMI Lab is shown in <sup>715</sup> Fig. [10.](#page-15-2)

<sup>716</sup> Once the deep relative pose estimation model is trained, it <sup>717</sup> is initially tested on the test set of Synthetic-Lab Dataset, fol-<sup>718</sup> lowed then by testing its prediction accuracy on real world data

<span id="page-15-1"></span>Table 9. Camera moving range on ASMI Dataset and its representative range on trained pose estimator. The camera is moving along the *z* − *axis*. The representative range is calculated by Eq.  $(14)$ .

trajectory ID	ASMI Lab Range (z-axis)	Representative Range(m)
$ASMI-1$	$3.122 - 2.569$	51.180 - 42.11
$ASMI-2$	$2.296 - 1.748$	37.64 - 28.66
$ASMI-3$	$1.564 - 1.015$	$25.64 - 16.64$



<span id="page-15-2"></span>Fig. 10. Examples of images captured in ASMI Dataset. (a) Original image captured in ASMI Lab (b) Segmented image with bakc background.

captured from the ASMI Lab, i.e. ASMI Dataset. The prediction accuracy of the deep relative pose estimator is reported in  $_{720}$ Fig. [11.](#page-16-0) Similar to the previous synthetic testing, position error  $\frac{721}{121}$ and attitude error are calculated by Eq.  $(5)$  and Eq.  $(6)$ , respectively. Compared with the prediction accuracy on the Synthetic-<br>  $723$ Lab Dataset, the position error of the ASMI Dataset is slightly  $_{724}$ higher. This could be attributed to variations in the illumination conditions compared to the Synthetic-Lab Dataset, as well  $_{726}$ as factors such as ground truth measurement noise and camera 727 calibration noise. On the other hand, the predicted attitude error in the ASMI Dataset is much smaller than the synthetic data.  $\frac{729}{200}$ One possible reason could be that the target remains stable at a  $\frac{730}{2}$ fixed position with rotation effects during the images capture.  $\frac{731}{731}$ 

# *4.2.2. FGSM Attacks on ASMI Dataset* <sup>732</sup>

To evaluate how the pose estimator can be impacted by adversarial attacks on real data, the FGMS attack is then applied  $_{734}$ to the ASMI Dataset. In this case, the FGSM is configured as  $735$ the same  $\epsilon$  as previously applied in synthetic, including 1, 0.5,  $\tau_{36}$ 0.3, 0.1, 0.05 and 0.01. In this experiment, all images are per- $\frac{737}{6}$ turbed by the FGSM attack. The model's average prediction 738 error under FGSM attacks with various  $\epsilon$  values on the ASMI  $\pi$ 39 Dataset are illustrated in Fig. [12.](#page-16-1)

As shown in Fig. [12,](#page-16-1) FGSM has a significant impact on position estimation but only slight impacts on attitude estimation. 742 In comparison to the previous experiment with synthetic data,  $743$ 



<span id="page-16-0"></span>Fig. 11. The prediction accuracy of the proposed pose estimator on Synthetic-Lab Dataset and ASMI Dataset after 100 epochs.The blue bar presents the average error on training data and the orange bar represents the average error on test data on Synthetic-Lab Dataset. The green bar indicates the average error on the ASMI Dataset.



<span id="page-16-1"></span>Fig. 12. Comparison of the prediction error under FGSM attack on ASMI Dataset with various  $\epsilon$  values. The blue bar indicates the average position error in meter and the orange bar represents the average attitude error in degrees.

 the FGSM attack has a more pronounced effect when  $\epsilon$  is less than 0.05 on the predicted position in the ASMI Dataset. How- ever, the attitude error remains quite stable, typically less than 1 degree, for all tested  $\epsilon$  values.

# <sup>748</sup> *4.2.3. LSTM-based Adversarial Attack Detector*

 To evaluate the adversarial attack detector on the ASMI Dataset, SHAP values are obtained by processing the pose es- timator on the Synthetic-Lab Dataset. Similar to the previous synthetic data experiment, the SHAP values are obtained from the output of the GAP layer in the deep relative pose estimator by DeepSHAP algorithm.  $1,000$  images from the training  $754$ data on Synthetic-Lab Dataset are randomly selected to gen-<br>  $755$ erate background data. A total of 30,000 SHAP value samples, consisting of 15,000 normal samples and 15,000 adversarial samples, are used to train the adversarial attack detector. The 15,000 normal samples consist of all images from the 759 test data on the Synthetic-Lab Dataset and randomly selected 760 images from the training data to account for the difference between 15,000 and the total number of images in the test data.  $762$ Adversarial samples are generated by applying FGSM attacks 763 to the normal sample images with randomly selected  $\epsilon$  values  $764$ from [0.5, 0.3, 0.1, 0.05, 0.01].

The SHAP values are shuffled and split by a train-test ratio  $_{766}$ of 0.8, i.e. 24,000 samples for training and  $6,000$  samples for  $\pi$ <sup>67</sup> testing. The adversarial attack detector is trained by SGD with 768 an Adadelta optimiser for 2000 epochs. Early termination is im- <sup>769</sup> plemented to reduce the training time. To do that, the training  $770$ data are further split into 80% for training and 20% for valida- $\frac{771}{771}$ tion. If the validation loss does not decrease over 20 epochs, the  $772$ training process will be terminated. After the early termination  $773$ condition, the proposed adversarial attack detector achieves a  $774$ detection accuracy of 99.18% on training data and 98.8% on  $775$ test data.  $\frac{776}{60}$ 

Subsequently, the pose estimator, FGSM attacks, and adversarial attack detector are integrated to evaluate the overall  $778$ performance on the ASMI Dataset. The integrated system is  $779$ identical to the one shown in Fig. [8,](#page-14-0) with the exception that 780 the 'Blender Image Generation' part is replaced by the ASMI 781 Dataset. In the ASMI Dataset, a random attack probability of 782 0.2 is applied to FGSM attacks. When an attack occurs, input 783 images are continuously perturbed by FGSM for the next 10  $784$ frames. The detection accuracy is calculated by Eq. [\(9\)](#page-13-0). Ta- <sup>785</sup> ble [10](#page-17-22) presents the detection accuracy on the ASMI Dataset for  $\frac{786}{60}$ various  $\epsilon$  values.

As shown in Table [10,](#page-17-22) the proposed adversarial attack detec-<br>  $\frac{788}{2}$ tor achieves an average correct detection rate of 96.29% on the 789 ASMI Dataset. The accuracy slightly drops when the  $\epsilon$  value 790 becomes smaller, which is caused by fewer perturbations ap-

<span id="page-17-22"></span>Table 10. The average accuracy of the adversarial attack detector in ASMI Dataset with various  $\epsilon$  values.

F	Detection Accuracy
0.5	$100\%$
0.3	$100\%$
0.1	100 %
0.05	98.44%
0.01	90.44%
Average	96.29%

 $792$  plied to the input images as  $\epsilon$  decreases.

#### <span id="page-17-10"></span><sup>793</sup> 5. Conclusion

<sup>794</sup> This paper firstly examines the impact of adversarial attacks <sup>795</sup> on DL-based spacecraft relative pose estimation in space ren-<sup>796</sup> dezvous scenarios. To do this, a CNN-based relative pose es-<sup>797</sup> timation algorithm is proposed. FGSM adversarial attacks are <sup>798</sup> implemented, which have a significant impact on the model's <sup>799</sup> predictions. Subsequently, an LSTM-based adversarial attack 800 detector is proposed to identify adversarial attacks on input im-801 ages. XAI techniques are adopted to analyse the model's pre-<sup>802</sup> dictions and generate SHAP values-based explanations for the 803 model's predictions. Multiple experiments are carried out to <sup>804</sup> evaluate the performance of the CNN-based spacecraft relative <sup>805</sup> pose estimator, how the adversarial attacks can impact on DL-806 based pose estimator in space rendezvous missions, and the per-<sup>807</sup> formance of the proposed adversarial attack detector. The pro-808 posed methods have been tested on both synthetic and real im-809 age datasets. The results show that the adversarial attack detec-810 tor performs robustly in detecting adversarial attacks, achiev-811 ing an average of 99.21% detection rate on synthetic data and 812 96.29% on real data collected from the ASMI Lab.

813 Although the impact of digital adversarial attacks on DL-<sup>814</sup> based spacecraft relative pose estimation has been analysed in 815 this work, how to physically implement the adversarial attacks 816 still needs to be explored. Moreover, the proposed method 817 demonstrates high accuracy in detecting adversarial attacks for <sup>818</sup> the DL-based spacecraft relative pose estimation, how to cor-819 rect the estimated pose after detecting adversarial attacks be-820 comes critical to provide a robust DL-based system for future <sup>821</sup> space missions.

# Acknowledgments and the season of the se



Project ID: ESA  $AO/2$ -1856/22/NL/GLC/ov.

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<span id="page-17-21"></span><span id="page-17-20"></span><span id="page-17-19"></span><span id="page-17-16"></span><span id="page-17-14"></span><span id="page-17-11"></span><span id="page-17-8"></span><span id="page-17-7"></span><span id="page-17-5"></span><span id="page-17-1"></span>acceptance of convolutional neural networks for automatic classification of 882 dopamine transporter spect in the diagnosis of clinically uncertain parkin-<br>883 sonian syndromes. *European journal of nuclear medicine and molecular* 884 *imaging*, (pp. 1–11). 885

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